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A Model of Global Positioning System (GPS)

Master Control Station (MCS) Operations

THESIS

David Nicholas Koster Captain, USAF

AFIT/GSO/ENS/92-9

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A Model of Global Positioning System (GPS) Master Control Station (MCS) Operations

THESIS

Presented to the Faculty of the School of Engineering
of the Air Force Institute of Technology
Air University
In Partial Fulfillment of the
Requirements for the Degree of
Master of Science in Space Operations

David Nicholas Koster, B.S.E.E.

Captain, USAF

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December 15, 1992

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Thesis Approval

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A Model of Global Positioning System (GPS)

Master Control Station (MCS) Operations

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David Nicholas Koster

Abstract

The United States Air Force's Navstar Global Positioning System (GPS) provides high-accuracy space-based navigation and time distribution to suitably-equipped military and civilian users. The system consists of earth-orbiting satellites and a world-wide network of ground stations. A single operational control center, the GPS Master Control Station (MCS) monitors, maintains, and commands the GPS satellite constellation. The on-going deployment of the complete satellite constellation and recent changes in the operational crew structure may invalidate previously used planning and management paradigms. There is currently no analytical method for predicting the impact of these and other environmental changes on system parameters and performance. Extensive testing cannot be performed at the MCS itself due to the criticality of the GPS mission and lack of operational redundancy. This research provides and validates a discret event simulation model of the MCS operations center task flow, focusing on the creation and testing of a sliding-widow MCS activity scheduler. The simulation was validated using MCS historical data. Experiments were conducted by varying the number of ground stations and satellite constellation size available to the simulation. The results, while not quantitatively trustworthy, were used to draw general conclusions about the GPS operational environment.

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A Model of Global Positioning System (GPS) Master Control Station (MCS) Operations

I. Introduction.

1.1 Problem.

The United States Air Force's Navstar Global Positioning System (GPS) is rapidly approaching fully operational status. The mission of the GPS is to provide high-accuracy space-based navigation and time distribution to suitably equipped military and civilian users (5:1). The system consists of a growing number of earth-orbiting satellites, a world-wide network of ground stations and communications segments, and a single operational control center, the GPS Master Control Station (MCS). The MCS, its ground stations and support facilities, and the communications network needed to interconnect these facilities, constitute the GPS Operational Control System (OCS).

The MCS monitors, maintains, and commands the GPS satellite constellation. It operates continuously with a crew of eight active-duty Air Force personnel. Their primary responsibilities are maintaining GPS time and navigation accuracy and periodically assessing the functional health of each GPS space vehicle (SV) (5:11). Continuous monitoring of satellite navigation data transmissions, forwarded to the MCS from five remote Monitor Stations (MS), allow the system positioning and timing accuracy to be evaluated. Four GPS Ground Antennas (GAs) provide the primary means of directly contacting SVs to upload improved navigation data, collect SV telemetry, and command configuration changes.

The on-going deployment of the complete satellite constellation and other changes in the operational crew structure may invalidate previously used planning and management paradigms.

There is currently no analytical method for predicting the impact of these and other environmental changes on system parameters and performance. Extensive testing cannot be performed at the MCS itself due to the criticality of the GPS mission and lack of operational redundancy. One solution to this problem is to create a simulation model of the MCS operational activity and perform the needed testing and experimentation using this tool. This will allow both planned and unplanned events affecting the MCS to be assessed without impact to current operations. This thesis provides and validates such a model and uses it to perform experiments on the probable operational impact of changes to the GPS operational environment.

1.2 Proposed Solution.

A model will be created, validated, and used to perform experiments in order to predict the performance of the Global Positioning System Operational Control System under a number of operational environment conditions. The model could allow GPS management to predict system resource loading under circumstances not yet experienced by the system. It could also allow "what-if" analysis to be performed to determine the response of mission effectiveness criteria to predicted system configuration changes.

1.3 Methodology.

In detail, this thesis will:

- Describe and evaluate current literature on the Global Positioning System organizational structure and operations, system modeling, simulation, and scheduling,
- Determine measurable criteria for evaluating the performance of the GPS MCS. MCS output products such as system computer and crew logs, operational performance measures, and interviews with GPS operations staff and crews will be used to develop these criteria.
- Evaluate competing methodologies for producing a model of the operations task flow at the MCS, based on the data gathered and the established performance criteria,
- Develop a model of current GPS MCS operations, including (as a minimum) the effects of:
 - Satellite contact requirements,

- Satellite constellation size,
- Ground station availability,
- MCS crew availability and operational constraints, and
- Unscheduled OCS resource outages.
- Validate the MCS model against current operational data not used in the development of the model,
- Create performance baselines for both the current system configuration and the planned 24 satellite operational constellation,
- Perform the experiments using the model and evaluate the results against the current performance baselines.

1.4 Implementation

It was determined that a computer simulation was the best model type for this task. Factors leading to this decision were the iterative nature of the scheduling technique to be used and the need to emulate the stochastic MCS activity durations. SIMSCRIPT II.5 (a registered trademark of CACI Products Company) was the simulation language selected to implement the model. The organization of the model program was tailored to correspond to the functional organization of the MCS, to aid in development and performance verification. To validate the completed simulation, a program environment was created to duplicate the conditions that existed at the MCS on 0000Z on 10 April 1992. The simulation was then executed and its scheduling performance was compared the actual MCS Master Contact Schedule for that day. The result was that 35.09% (20 of 57) of the satellite activities scheduled by the simulation were within ten minutes of the historical scheduled time, and 31.58% (18 of 57) exceeded one hour. On this basis, the model was judged to be insufficiently accurate to allow direct quantitative comparison to MCS performance. However, experimentation was performed to allow a qualitative analysis of the model's performance under various conditions.

1.5 Results.

The results of experimentation indicate the model may be useful for predicting the general behavior of the Global Positioning System Master Control Station operational performance. The relationships between MCS resource scheduling capability and mission requirements were characterized, as were the effects of satellite constellation growth and ground station outage. While the results are not useful in predicting actual resource utilization rates, they do indicate what can be expected in terms of relative utilization and rate of increased resource use.

II. Background

2.1 Chapter Overview.

The creation of a simulation requires a comprehensive understanding of two systems. The first is the system being modeled. These objects include the people operating the system (if any), the devices used (again, if any), and the resources the system consumes or produces. The system being modeled here is the Global Positioning System, which requires all the above objects. This chapter will describe this system in detail, using information from GPS program development documentation, actual system operating instructions, and first-hand GPS operational experience.

The second system requiring in-depth familiarity is the modeling environment itself. Creation of an effective model requires knowledge of general simulation and operations research theory; the relative advantages of alternative modeling systems; and in-depth knowledge of the environment selected for use. These topics as they pertain to this research are covered in this chapter.

2.2 The Navstar Global Positioning System.

The Navstar Global Positioning System (GPS) is the United States Air Force's space-based, highly-accurate radionavigation system. Since it was first made available for three-dimensional position determination in December 1978, this system has created a revolution in disciplines such as navigation, ordinance delivery, search and rescue, surveying, and precise timing (10:1-7). The GPS allows any number of properly-equipped users to simultaneously determine their time to within 100 nanoseconds of Universal Time; their position within 16 meters spherical error probable (SEP) (5:8), and their velocity to within 0.2 meters per second (10:1-13).

Very simply, GPS consists of a number of earth-orbiting satellites that continuously broadcast the current time and their orbital parameters. Users equipped with GPS receiver sets on land, sea, in the air, and in low-earth orbit, receive this information from all the satellites in their view. Sign and data processing components in the receiver set estimate the current position of each satellite

and measure the propagation delay of radio signals from the satellite to the user. Using these time delays, the set solves simultaneous equations (one for each satellite used) in position and time to produce the users time, position, and velocity.

The accuracy of the above calculations depends on knowing the satellite time and position precisely. However, the satellite on-board time standard and the satellite orbits are subject to both random and cumulative error. GPS satellites are currently not able to make autonomous and real-time corrections to the time and position data they broadcast, so a network of monitoring receivers continuously collect satellite range data. These data are transmitted to a master control station, where accurate estimates of satellite time standard drift and orbital trajectory are calculated. These estimates are mathematically propagated into future, formed into the proper data format, and transmitted back to the satellites via transmitting antennas. The satellite then retransmits this data to the users.

As a system, the GPS consists of three major subsystems (Figure 2.1). The Space segment consists of the satellite constellation. The Control Segment is the Master Control Station, the monitoring and transmitting stations, and the communications links between them. Finally, the User Segment is the community of military and civilian organizations and personnel equipped with GPS receiver sets. The functional inter-relationship of these segments is diagrammed in Figure 2.2. Each segment is described in detail in the following sections.

- 2.2.1 Space Segment. The Space Segment is a growing number of earth-orbiting satellite vehicles (SVs), which constitute the GPS "constellation". An understanding of function of the satellites and the implications of the geometry of their orbits is required for correct model design and analysis.
- 2.2.1.1 GPS Satellite Orbits. The orbital height for operational GPS SVs is nominally 10,930 nautical miles, or about 20,200 kilometers. This orbital altitude gives the satellites an 11 hour

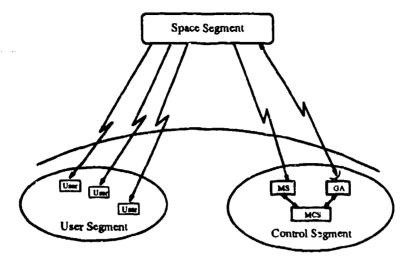


Figure 2.1. Navstar GPS Segments

58 minute (semi-synchronous) orbital period. When combined with the motion of the Earth, this orbit results in each satellite appearing in the same location of the sky once per day for a fixed observer. However, the difference between the satellite orbital period and the length of the solar day causes each satellite to appear approximately 4 minutes earlier each day. The intended orbits are circular; in practice, most have an eccentricity of less than 0.01. There are six orbital planes, labeled "A" through "F", into which the satellites are launched. There will be four satellite in each plane, spaced 90 degrees apart. As noted below, the orbital planes of the research and development SVs have a greater inclination than that of the operational satellites.

The orbital parameters of the GPS constellation were chosen to optimize the performance of the navigation mission. There are a number of tradeoffs involved. High altitude orbits provide long periods of visibility for a given location on the Earth, but these distances require greater transmitted signal power for reliable communication. At lower altitudes, however, the SV/user range changes rapidly as the satellite passes overhead, degrading system accuracy.

2.2.1.2 GPS Satellites. From February 1978 to September 1985, ten GPS research and development satellites (collectively called "Block I") were launched. These satellites, designed and

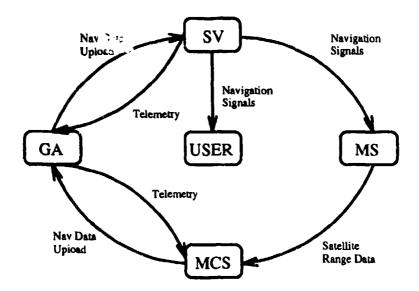


Figure 2.2. Global Positioning System Functional Diagram

built by Rockwell International, were designated BI (for Block I) -1 through -11. (The numbering discrepancy arises from the launch failure of BI-5, whose Atlas booster exploded on the launch pad in December 1981 (10:1-8).) These spacecraft, intended to have a five-year lifespan, allowed the GPS concept to be tested and validated. The Block I satellites have shown remarkable longevity, however, and four of these SVs were still providing service as of July 1992 (3). The eldest of these, BI-8, was launched in July 1983 (10:1-8). Block I GPS satellites have an orbital inclination of 64 degrees, as opposed to 55 degrees for the operational constellation. This was to improve satellite visibility over the test sites (primarily the Yuma Proving Ground in Arizona) during the system test and evaluation phase (10:1-6).

The launching of operational satellites began in February 1989 and continues to the present. These satellites, labeled Block II, in addition to being larger and more expensive, have additional capabilities over the Block I SVs. While the Block I on-board satellite navigation processors had only enough memory to retain 26 hours of navigational data, the Block II SVs can store 14 days of usable data. The Block II vehicle electronics suite was hardened against radiation, the effects of which plagued the Block I vehicles. The design life of Block II GPS SVs is 7.5 years (5:91).

A subsequent upgrade to the Block II vehicles, designated Block IIA, further increased the usable on-board navigation data storage to 180 days. The Block II SVs are numbers BII-1 to BII-9; BII-10 was the first Block IIA satellite. The Block II satellites can be programmed to prevent users without official authorization (in the form of cryptographic keys) from using the system to its fullest capability.

The current target is to have a constellation of 21 operational satellites with three on-orbit spares (5:7). The spares will be active, bringing the minimal number of usable satellites for an operational system up to 24 (10:1-3). This number does not include the remaining operational Block I satellites. The upper limit to the number of operational satellites is 30, due to the design of the Master Control Station (MCS) (5:11). A 24 satellite constellation will result in the 98% three-dimensional positioning availability requirement specified in the System Operational Requirements Document (SORD) (10:6-2).

As a secondary payload, the later Block I and all subsequent satellites carry nuclear burst detection sensors, part of the Nuclear Detonation (NUDET) Detection System (NDS). These sensors detect the detonation of nuclear devices anywhere in the world and relay the exact time and range of the burst to suitably equipped receivers. This data, hopefully from multiple satellites, is then used to calculate the precise time and location of the detonation.

2.2.2 User Segment. Military and civilian users who require precise time or position information and are equipped with suitable GPS receivers make up the User Segment. As described above, each satellite continuously broadcasts information that allows GPS receivers to calculate the satellite's position with respect to a common reference system. With information from three satellites, the GPS receiver can compute ("triangulate", in a sense) the users' position and velocity in three dimensions. However, because the receiver set's clock must be synchronized with the GPS common time, the information from a fourth satellite is needed to derive the users' local clock/GPS time offset. If the user knows (or does not care to know) any part of the time and position equation,

the remaining unknowns can be calculated with data from fewer than four satellites. Two examples of this are ships at sea (their height above sea level is known) and users with highly accurate time standards synchronized to GPS time. These users would require information from only three satellites.

A important byproduct of this process is that the user now has an extremely accurate time reference, as part of the of data received from the satellite describes the offset between GPS time and Universal Coordinated Time, or UTC. These data allow suitably equipped users to calculate the local time to within 100 nanoseconds (100 billionths of a second) of UTC.

There are two levels of GPS accuracy: the Precise Positioning Service (PPS) and the Standard Positioning Service (SPS). PPS users are those with both a suitably-equipped user set and the necessary cryptological keys to decode the PPS transmissions from the GPS satellites. Users with PPS-incompatible receivers, and authorized users without cryptokeys, are restricted to the SPS. PPS provides positional accuracy with errors no greater than 16 meters Spherical Error Probable (three dimensional error not to exceed 16 meters), while SPS provides 76 meters SEP. Both types of users have access to precise time and time interval information (5:7).

Receiver sets vary in size and sophistication dependent on the requirements of the users. For instance, the Small, Lightweight GPS Receiver (SLGR) hand-held set weighs less than five pounds and will fit in the pocket of a standard-issue Battle Dress Uniform (10:1-10). This user set can receive only the SPS signals, which provides better than 100-meter horizontal error for footsoldiers and slow-moving vehicles. Larger and more complex receivers, with the ability to electronically track up to five SVs simultaneously, are designed to mount in high-performance aircraft and have the computing ability to update a fast-moving users' position quickly. As described above, authorized users receive cryptographic keys to decode the highest accuracy signals from the GPS satellites.

2.2.3 Control Segment. Lastly, the Control Segment consists of a central control facility called the Master Control Station (MCS), a number of transmitter and receiver stations located

around the world to gather system data and command and control the satellite constellation, and world-wide communications network that conveys data to and commands from the MCS. The Control Segment has the responsibility for maintaining system availability and accuracy within established standards. Using the dedicated communications network and ground stations, the MCS continuously collects data broadcast by GPS satellites, processes the data into accurate satellite time and location predictions, then transmits the predictions back to the satellites. This data in turn is rebroadcast to the GPS users, who use the data to solve for their own time and location. At the same time the calculated data is transmitted to the satellite, the operational crew at the MCS collects and reviews satellite telemetry data for satellite anomaly detection and performance trending. The Control Segment is the focus of this research and is described in detail in the following sections.

2.3 GPS Operational Control Segment.

The 2nd Satellite Operations Squadron (2 SOPS), located at the Consolidated Space Operations Center (CSOC), Falcon Air Force Base (AFB), Colorado, has the mission of operating the GPS. Falcon AFB is itself located approximately 15 miles east of the city of Colorado Springs, Colorado. The 2 SOPS, a subordinate unit of the 50th Space Wing, is also responsible for maintaining the GPS ground stations, maintaining navigation and timing quality control/quality assurance, and providing user interface for special users. The GPS Control Segment, also called the Operational Control System (OCS), consists of five Monitor Stations (MS), three standard Ground Antennas (GA), a Prelaunch Compatibility Station (PCS), the MCS, and the communications network needed to link the MCS with the remote stations. The OCS also may utilize the ground antenna (part of the Air Force Satellite Control Network (AFSCN)) located at Falcon AFB. This GA has the AFSCN network designation "PIKE". Each of these type of facilities is described below, with Table 2.1 providing the exact locations. Figure 2.3 is a block diagram of the entire OCS.

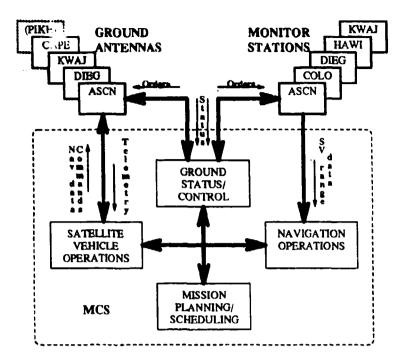


Figure 2.3. GPS Operational Control Segment (OCS) Block Diagram (from (5:119)).

2.3.1 Monitor Stations. Monitor Stations serve as the MCS's listening posts for the navigation signals transmitted by the satellites overhead. There are five GPS Monitor Stations, four scattered about the globe near the equator and one co-located with the MCS at Falcon AFB. (refer to Table 2.1 for specific MS location data). Monitor Stations are sophisticated, semi-autonomous receivers, able to track the signals of up to nine satellites simultaneously under command of the MCS. The MCS provides the MS with information on the satellites in view, allowing data preprocessing at the MS to reduce the amount of data returned to the MCS and to speed the satellite signal acquisition process.

Monitor Stations process the signals received from the SVs and continuously calculate the range between the MS antenna and each SV being tracked. The MS antenna locations have been surveyed to within one meter to accomplish this task with the precision required (5:87). The MS transmits the results of these measurements to the MCS. The MCS then uses these data to

accurately estimate the positions of the SVs and the drift of their on-board time standards. Another critical role of the MS is to provide immediate feedback to the MCS in the case of satellite failure or anomalous behavior. The MCS crew watches the availability and quality of satellite data from the Monitor Stations closely for any sign of navigation signal degradation or satellite subsystem failures.

Table 2.1. GPS Monitor Station and Ground Antenna Locations (5:106)

Designation	Location	Туре	Latitude (deg)	Longitude (deg)
ASCN	Ascension Island, South Atlantic Ocean	MS/GA	7.9S	14.4W
COSPM	Falcon AFB, Colorado	MS	38.8N	104.5W
PIKE	Falcon AFB. Colorado	GA*	38.8N	104.5W
DIEG	Diego Garcia Island, Seychelles, Indian Ocean	MS/GA	7.25	72.4E
KWAJ	Kwajalein Atoll, West Pacific Ocean	MS/GA	8.7N	167.7E
HAWAIM	Oalıu, Hawaii	MS	21.6N	158.2W
CAPE	Cape Canaveral AFB, Florida**	GA	28.5N	80.6W

^{**} Prelaunch Compatibility Station

2.3.2 Ground Antennas. The GPS Ground Antennas are the primary means for command and control of GPS satellites. The three dedicated GPS GAs are co-located with Monitor Stations at Kwajalein, Ascension, and Diego Garcia, as described in Table 2.1. The Prelaunch Compatibility Station (PCS) at Cape Canaveral is an identical facility used for ensuring the interoperability of new satellites with the OCS just prior to launch. When not reserved for its testing role, it serves as a fully functional GA (5:95).

In addition to these dedicated resources, the OCS can use the AFSCN Remote Tracking Station (RTS) located at Falcon AFB for communicating with the GPS constellation. This is achieved through the use of a unique electronic interface device that allows the RTS to emulate

a GPS Ground Antenna. As this facility is not dedicated to GPS and has substantial AFSCN commitments, its use must be prearranged with AFSCN schedulers. The AFSCN operators can perform command and control functions on GPS satellites, but that network is not configured to calculate and upload navigational data.

Each CPS GA is an active S-band transmitter and receiver, equipped with a 10-meter diameter steerable parabolic dish antenna. On command from the MCS, the antenna is pointed using the current MCS estimate of the satellite position. Once two-way contact is established with the SV, the GA transmits real-time or prestored commands and simultaneously receives SV telemetry (5:95). Figure 2.4 shows schematically what the relative GA coverage areas are, given the GA longitude and an approximate 140 degree coverage arc (5:99).

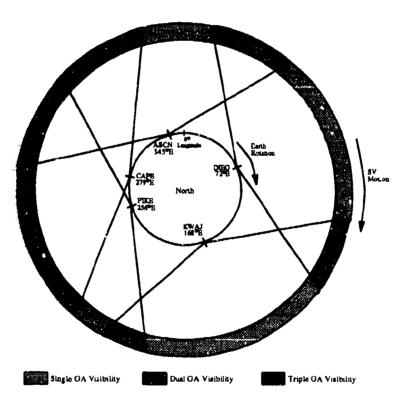


Figure 2.4. GPS Ground Antenna Coverage (from (5:99)).

- 2.3.3 Communications. Control Segment communications link the MS/GAs to the MCS. There are generally two 9600 bit-per-second (bps) links to each of the GAs, and a single 9600 bps link to the MSs. The communications channels are carried by both military and commercial communications networks (5:100).
- 2.3.4 Master Control Station. The MCS acts as the hub for all GPS activities. Its mission is to command and control the GPS satellite constellation and the other GPS resources. The MCS is designed to monitor and control 30 GPS satellites simultaneously in any mix of Block I and Block II types (5:11). The MCS is one of four satellite operations centers in the CSOC. Physically, it consists of the operations center ("ops center") itself, the MCS computer facility, maintenance and support facilities, the Colorado Springs Monitor Station, and an NDS mission support area operated by the Air Force Technical Applications Center (AFTAC), the primary users of the NDS data.

In the ops center, there are 11 computer console positions. Any of the six distinct crew functions (listed in Table 2.2) can be performed at any of the consoles. Detailed descriptions of the responsibilities of each crewmember are provided below.

Table 2.2. GPS MCS Crew Position Summary

Position	Symbol	#	Rank
Flight Commander	FCMDR	1	Lt/Maj
Crew Chief	CCHIEF	1	TSgt-SMSgt
Satellite Analysis Officer	SAO	1	Lt/Capt
Satellite Vehicle Officer	SVO	1	Lt/Capt
Satellite Support Operator	SSO	3	Amn/TSgt
Ground Systems Operator	GSO	1	Amn/TSgt

The MCS computer center, using two IBM 3083 mainframe computers and associated storage and communications hardware, is the heart of the operations center. The computers, together with

the GPS Mission software, provide all data processing support for the operations mission. The major services this system provides includes:

- data collection from the Monitor Stations,
- calculation of satellite trajectory and satellite and MS time standard drift estimates,
- creation and formating of SV navigation uploads,
- maintenance of system parameter databases and logs,
- operations consoles display format and control,
- support for transmission of GPS data to external agencies, and
- support for internal functions such as planning, scheduling and report generation.

The tasks assigned to each MCS crew position provide a comprehensive overview of how the MCS performs its command and control mission. These tasks are described below. The general crew task flow and interaction is shown graphically in Figure 2.5.

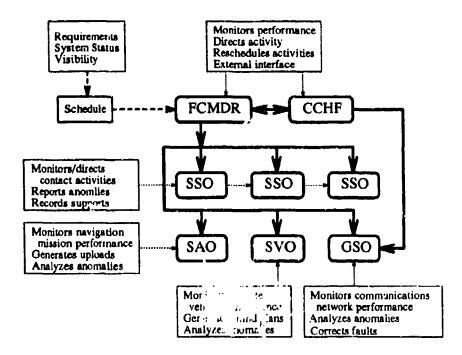


Figure 2.5. MCS Crew Organization and Tasks (5:37)

- 2.3.4.1 Flight Commander (FCMDR). The FCMDR monitors and directs the crew in the performance of mission operations. The FCMDR approves the schedule for his shift, makes schedule changes as required, and gives the final authorization for the preparation and execution of every satellite contact. The FCMDR is also the focal point for communications between the crew and external agencies.
- 2.3.4.2 Crew Chief (CCHIEF). The CCHIEF, in addition to acting as the deputy Flight Commander, is responsible for coordinating the activities of the MCS with maintenance functions and other support activities. The CCHIEF also performs the up-channel reporting tasks assigned to the operational crew.
- 2.3.4.3 Satellite Support Operator (SSO). The SSO personnel perform the satellite contacts. This entails coordinating and monitoring GA operations, configuring the system for the contact, executing the commands directed by the schedule, and recording the details of the contact. The SSO also reviews satellite telemetry during the contact and reports anomalies to the SVO and SAO.
- 2.3.4.4 Satellite Analysis Officer (SAO). The SAO is responsible for the performance of the navigation and timing subsystems. This position monitors mission performance using data from the Monitor Stations and through satellite telemetry received via the GAs. The SAO also manages the navigation message generation process, and assists the SSO in uploading the message to the SVs. The SAO provides the primary technical support for navigation mission anomaly resolution for the crew.
- 2.3.4.5 Satellite Vehicle Officer (SVO). The SVO functions are similar to the SAO, except that this position is responsible for all satellite systems other than the navigation subsystems. These include subsystems such as command/telemetry processors, transmitters/receivers, attitude

control, thermal control, and power generation and conditioning. The SVO plans the satellite commanding performed by the SSO. This position also assists in satellite "bus" anomaly resolution.

- 2.3.4.6 Ground Systems Operator (GSO). The GSO configures and monitors the performance of the GPS communications network and ground stations. The GSO will also perform fault isolation procedures and assist maintenance personnel at the remote sites to correct deficiencies (5:37).
- 2.3.5 MCS Operations. The MCS is not a general use system; it is specifically designed to perform the task of operating the GPS. As such, there are a finite number of well-defined tasks that the MCS as a system can and does perform. This section describes the MCS task generation and scheduling process.
- 2.3.5.1 MCS Activities. The GPS program requirement documents are the ultimate source of GPS activities. They specify the eventual end product (a highly accurate radionavigation system) in detail, defining system accuracy, operational rates, availability standards, etc. They describe the goal.

More detailed, "tactical" documents describe how the requirements are to be met, in terms of when specific activities are to be performed, who is to do them, and the criteria for success. These documents include manufacturers' technical handbooks, middle echelon requirements documents such as the On-Orbit Maintenance Requirements (OOMR), and Headquarters/Wing/Squadron Operational Instructions (OI). These in turn are broken down into detailed "how-to" documentation for the personnel performing the activities. The Navstar Satellite Support Requirements (NSSR) is one of these, produced by the 2 SOPS to define the timing requirements for each activity performed on the GPS satellites.

For the purposes of this document, and when referring to MCS operations, a requirement is the rationale for one or more MCS activities to be performed. The requirement originates from the GPS system requirements documents or from operational squadron operating instructions. An activity is a single, discrete function, performed by the MCS crew as a whole, that satisfies a requirement.

Table 2.3 shows three of the 78 different MCS activities described in the NSSR. ACTIVITY is the acronym used on the MCS Master Contact Schedule. SCHEDULED describes the scheduling interval; these may be periodic (as shown for ADDKEYS), periodic with caveats (like BAT3TEMP), or aperiodic (like BCHGROFF).

Table 2.3. 2 SOPS NSSR Excerpt (1:7)

ACTIVITY: ADDREYS (ADDREYS transmission)

SCHEDULED: Twice per week (Hormally on Sunday and Wednesday if L1 &

L2 transmitters are enabled).

ACTIVITY TH: 10 minutes

PURPOSE: To uplead new navigation processor address keys.

REQUIREMENT: Execute ADEST Procedure.

WINDOW: -1 to +1 hour from scheduled support time.

RESCREDULED: Accomplished prior to reusing address keys to avoid

unauthorized access to the navigation processor.

ACTIVITY: BATSTEMP (Battery 3 temperature monitor)

SCHEDULED: Once per week when the SVE is 105 degrees and

decreasing. (For designated satellites only)

ACTIVITY TH: 10 minutes

PURPOSE: To determine maximum battery 3 temperature. REQUIREMEST: Perform Battery 3 Temp Monitor Ops Procedure.

WIEDOW: +/- 5 minutes from scheduled initiation time.

RESCREDULED: Must not violate WINDOW to ensure that battery 3 does

not exceed 50 degrees Centigrade while connected to

the main bus.

ACTIVITY: BCEGROFF (Battery Charges OFF)

SCHEDULED: As required by a schedule request

ACTIVITY TH: 5 minutes

PURPOSE: To turn a battery charger off

REQUIREMENT: Perform Battery Charger Off Ops Fr. .edure with BAT.ONE,

TWO or TERRE.

WINDOW: -30 to +30 minutes from scheduled initiation time

RESCHEDULED: Must not violate WINDOW to avoid possible damage to the

satellite unless stated otherwise by the schedule request.

The ACTIVITY TM is the expected duration of the activity. PURPOSE and REQUIRE-MENT assists the scheduler and crew in verifying the need for the activity. The allowed slack about the scheduled time is shown as WINDOW; the requirement may be considered failed if this is not met. Finally, RESCHEDULED contains instructions to the FCMDR on how to reschedule or otherwise react in the case where the scheduled time is missed.

2.3.5.2 MCS Scheduling. Taking advantage of the periodic and well-defined nature of satellite support requirements, routine satellite support activities can be scheduled in advance. Scheduling satellite activities is simply a process of matching the time the activity is due to be performed with the resources needed to perform it. This process is complicated by the shifting SV/GA visibility periods; by scheduled and unscheduled equipment outages; by conflicting requirements; and by additional aperiodic contact requirements that "shift" the time of routine contacts.

The MCS personnel create the GPS Master Contact Schedule every day for the next 24 hour period starting at 0600 GMT. The scheduling process is:

- 1. Find the visibility of each SV with each GA. The MCS computer system provides this data as a printed report, using the current estimates of SV trajectory to calculate satellite visibility periods and elevations at each of the ground stations.
- 2. Collect the Schedule Requests (SR) for aperiodic SV contact activities and ground station maintenance. These requests are submitted in writing by staff and support functions to reserve system resources for maintenance and testing at some specific time. The requests are channeled to the scheduling function after approval by the FCMDR.
- 3. Identify the periodic activities that must be performed on the day being scheduled. This is done by examining the system logs to find the last time these activities were performed, then adding the maximum time interval (from the NSSR). The activity must be performed again prior to the resulting time.
- 4. Draft the schedule, starting with the SRs with fixed activity start times, then periodic activities. As each activity is added, it must be examined to ensure it meets visibility requirements, meets system requirements, does not conflict with previously scheduled activities, and otherwise has the necessary system resources available to accomplish its task.
- 2.3.5.3 MCS Operations. During the performance of a satellite support, the MCS crewmembers are cooperating to perform the activities on the Master Contact Schedule. There may be zero, one, or more activities being performed simultaneously in the MCS at any given time, but generally a single crewmember can be involved in only one at a time.

As an example, a common set of activities performed during a satellite contact is a SV state-of-health (SOH) review, a download of SV Global Burst Detector (GBD) processor data (a GBDDUMP), and finally a navigation data upload (NAV). The SSO and the GSO initiate the contact by ensuring the GA is ready. Then with FCMDR authorization, the SSO commands the GA to begin transmitting. After contact is established and telemetry from the SV received, the SSO and the SVO review the status of the satellite's subsystems (the SOH). The SSO then commands the SV to download GBD data (the GBDDUMP). While this was occurring, the SAO was commanding the MCS computer system to calculate "fresh" estimates of the SV's clock and trajectory parameters, form them into a navigation data upload, and transmit the upload to the GA. The SSO then instructs the GA to transmit the data to the SV. The contact ends when the SSO and GSO break the communications link and return the GA to ready status

The schedule has been constructed under the assumption that each of the scheduled activities will take little or no time longer to perform than the activity duration shown in the NSSR. As the result of experience in both scheduling and performing these activities, the durations in the NSSR are conservative and seldom does the performance time exceed the expected. Experienced MCS crew personnel have stated these durations are exceeded as result of satellice, communications, ground station, or MCS computer malfunction (3), (14).

2.4 Modeling.

A model of a system is a representation of the objects, concepts, or ideas of the system in some form other than the system itself (15:2). System models of all types are simply tools for describing system functionality in a language understood by the model builder. Models are also a substitute for real or conceivable systems in cases where the system itself cannot be used for analysis or is otherwise unavailable. That may be due to any number of reasons. The system may not exist (yet), or it may be too expensive or dangerous or inconvenient to test directly. In any

case, the situation requiring the model is this: the analyst requires information about the operation of a system that cannot be readily obtained from the system itself.

The environment in which the model is defined is best chosen to optimize the desired result of the analysis. For instance, if the intent of aircraft design research is to evaluate the ability of a particular design to fly, a physical model or prototype can provide this information directly. However, other factors also must be considered. The type of system, the cost of the modeling process, and even the predilection of the researcher also drive the choice of modeling discipline.

However, the explosion of computer technology in the last thirty years has allowed simulation modeling to become the principal method of describing large and complex systems (16:7). Dedicated high-level simulation languages have promoted this growth, as has the inherent flexibility and speed of digital computers.

2.4.1 Model Types. When a model is required, a fundamental decision to be made is what type of model will be created. There is a wide spectrum of model types, each with characteristic strengths and weaknesses. A common classification technique for model types is to place them on a continuous scale according to how they relate to the real system being modeled. Closest to reality are physical and scaled models. Physical models include full-scale mockups and prototype systems. Scaled models are common for physically large systems, such as a world globe modeling the Earth (15:8).

Less concrete are analog models, which have the same functionality as the real system being modeled, but are less like it in form. A graph is a simple analog model; the length of a line or bar on the graph is an analog for some physical property, such as time or mass. Further into the abstract is the computer simulation, where the representation of the real object or system exists only in computer source code and binary numbers. The least concrete of the modeling environments is the mathematical model, where the representation of the system is described only in mathematical or logical symbols. While this level of abstraction is certainly inexpensive and flexible, often the

assumptions needed to fit reality into this format reduces the accuracy and usefulness of the model (15:10).

Another consideration when choosing the type of model is the need for stochastic properties.

Most models of real-world systems involve some uncertainty in either their input factors or their internal processes. A decision must be made to either include these stochastic properties in the model or to assume reasonable deterministic values and avoid the issue. The model environment must be selected to accommodate either decision (4:13).

2.4.2 Simulation. According to Shannon (15:2), simulation is

...the process of designing a model of a real system and conducting experiments with the model for the purpose either of understanding the behavior of the system or of evaluating various strategies (within the limits imposed by a criterion or set of criteria) for the operation of the system.

Most researchers recommend a systematic approach to developing a simulation model, and there are frequent commonalities between the approaches. Emshoff and Sisson (4:50) describe an approach shown in flow cuart form in Figure 2.6. This sequence differs from that of other authors such as Pritsker (11:10) and Shannon (15:23) in that it emphasizes the *iterative* nature of the model-building process and stresses the creation of subprocesses. This incremental approach is useful for large or logically complex models.

As noted above, the computer simulation is a common method of modeling. It is especially useful when the system being modeled involves stochastic processes or inputs; with the proper programming, the computer can quickly create random variables with many different distributions. The major drawback of computer simulation is its high level of abstraction. In addition, the translation of the desired model into the language of the computer requires specialized knowledge that generally has nothing to do with the environment of the system being modeled.

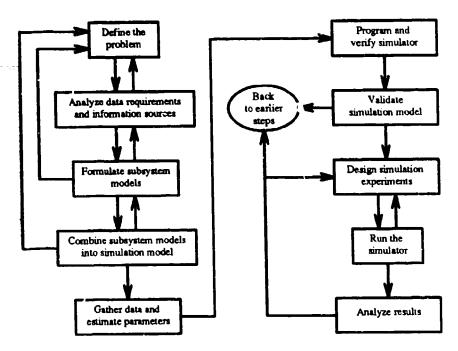


Figure 2.6. Simulation Model Development Sequence (from (4:50)

2.4.3 Operations Simulation Models. The majority of the research in this area concentrates on the modeling of manufacturing or service systems. Manufacturing systems models have little in common with the operational control systems, as the human element and its unique qualities (dynamic decision-making, reaction times, the capacity to make outrageous errors) so integral in operations is not a major factor in mechanized manufacturing processes.

However, recent manufacturing and service system models give an indication of the "state-of-the-art" of modeling in general. This may lead to the discovery of techniques applicable to operations modeling. Medical facilities, for instance, have a complex operation with unique constraints, and often closely parallels the satellite operations environment. A hospital pharmacy in Knoxville, TN was modeled using SIMAN, a modern personal computer-based simulation language (8:91). The level of detail was down to simulating telephone calls interrupting the pharmacy employees in their duties (8:97). Another study modeled the patient flow in a hospital outpatient center then under construction, and its conclusions resulted in the redesign of the facility (17:26).

Research in modeling the operational environment is generally directed toward the creation and testing of so-called "expert systems". These are computerized decision-making assistants, that have been "trained" by analyzing the optimum decisions made by the human operators and creating a computer program to model this decision-making process. The operators then enter the current situation into the computer expert and this program provides the necessary decision (or decision-making aid) more quickly and accurately than a typical human operator.

The most significant current progress in the design of operational control systems is the creation of the Georgia Tech-Multisatellite Operations Control Center (GT-MSOCC). This facility, a joint project of Georgia Tech University and the National Aeronautics and Space Administration (NASA), is a "...real-time, discrete event, interactive simulator of the operator interface to a NASA ground control system" (13:627). This complex model is used to test new control procedures and mission subsystems for NASA prior to real-world implementation. It is the key facility for the research into human/computer interaction and decision-making process for complex systems (13:636).

2.5 Scheduling.

A major task of satellite operations centers such as the MCS is to optimize the use of the resources needed to accomplish their mission. As described above, the availability of these resources constrain the operations center's ability to perform required tasks in the required time frame. The problem then is to find the mutually optimum time to perform a finite number of tasks of fixed but nonuniform duration, which require some limited set of resources. This problem resembles those categorized in the operations research discipline as "scheduling" or "sequencing" problems. The basic theory of this class of problem is described below, along with common solution techniques and problem examples similar to those found in satellite operations.

2.6 Scheduling Theory.

One of the basic problems in domain of operations research is the sequencing, or job-shop, problem. This class of problem arises when the optimum order of tasks is to be determined, with each task requiring some fixed amount of each of a number of resources. The archetypical example (and the source of the problem class's name) is a metalworking shop. Each item of finished product must spend a fixed amount of time (dependent of the type of item) on a variety of machine tools, such as lathes, drills, punches, etc. The order in which the items are sent to the various machines is not fixed. This case is shown graphically in Figure 2.7(a). The problem then is to find the order or sequence of item/machine pairings that optimizes some specific performance measure. Examples of typical measures are total production time (makespan) for one or all items; machine tool idle time; and total waiting time for some number of items to be produced (12:48).

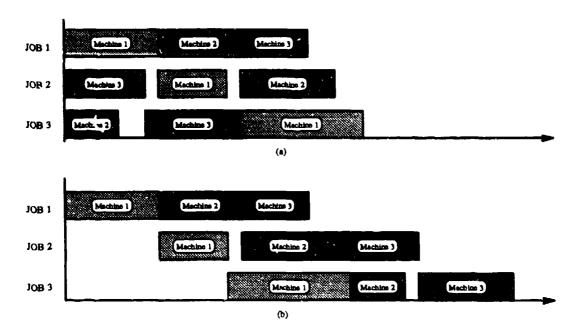


Figure 2.7. Job Shop/Flow Shop Schedules (from (2:180))

If the process requires each item to be smoothed, drilled, and punched an identical order it becomes a special case of the job shop problem and is called the *flow-shop* type problem (Figure 2.7,(b)). A further subtlety is whether items may skip ahead of others while each item maintains

its proper machine sequence. This is called *passing* and naturally complicates analysis of flow-shop problems (12:49).

There are a number of criteria used for judging the effectiveness of schedules. The most common is the amount of time needed to complete all jobs in the system. It is generally desirable to create a schedule to minimize this time, so the criteria is called min-makespan. Another type of measure is the mean job flowtime, the average amount of time jobs spend in the system. This is a good indication of the congestion in the system. When the jobs in a system must meet specific due dates, a measure of scheduling effectiveness is tardiness. This is the number of jobs whose actual completion time is after the required completion time. The mean or total time in excess of the required completion times for the system can also be used (2:223).

An extension of the simple job-shop problem is the parallel identical processor/independent job problem. The resource uniformity allows simple heuristics to solve this class of problem for many scheduling measurement criteria. Iterative techniques using simulation with some form of priority rules are commonly used to solve these types of scheduling problems. Analytical solutions for problems with two and three machines have been discovered. The common solution method for larger problems is to use a multistage decision process, examples of which are dynamic programming, branch-and-bound methods, and backtrack programming. Each of these is a form of intelligent enumeration, that avoids complete enumeration of all existing schedule possibilities (12:49).

III. Methodology

3.1 Overview.

The creation of the MCS Operations Center model follows the strategy described in Section 2.4.2 and shown in Figure 2.6 (Page 2-20). Information describing the processes to be modeled and their quantitative output must be collected. Using these data, measures of performance must be determined, in order to validate the model's performance and measure the experimental result of the model's operation. Once the function, purpose, and measure of performance are known, the model subsystems can be created, verified, validated, and tested. When the subsystems are complete, they are integrated into the whole. This chapter describes in detail how each of these steps were completed for this thesis.

3.2 Data Collection.

Simulating the MCS scheduling process and the operations center task flow required information describing these processes. System requirement information was needed to identify the mission critical activities and how these constrain schedule creation and operations. Operational data were required to define the activity flow internal to the operations center and to derive the actual activity interval and duration time periods. Lastly, historical records were needed to perform model validation. In addition to the above "hard" data, informal interviews were conducted with current and former operational crewmembers to provide initial face validation of the overall simulation concept. The data described as follows was provided by the 2 SOPS to this researcher during an official visit from 29 June to 3 July 1992.

3.2.1 System Requirements. System/mission requirements data were needed to define the constraints under which the MCS created and performed the daily activity schedule. The overall GPS mission requirements are described at the highest level in the Navstar GPS System Operational Requirements Document (SORD). These general requirements, supplemented by technical

documentation provided by the GPS satellite manufacturer (Rockwell International), were used by personnel in the 2 SOPS to create the document called the Navstar Satellite Support Requirements (NSSR) (see Figure 2.3, Page 2-15). The NSSR is the primary reference for operational satellite contact requirements. It is utilized by 2 SOPS personnel for scheduling future satellite contacts and for prioritization of new satellite contacts necessitated by real-time schedule changes. For each routine MCS activity, the NSSR describes:

- whether an activity is periodic or scheduled "as needed",
- how often scheduled activities must be performed,
- the expected activity duration,
- the activity's purpose,
- the time window about the scheduled time in which the activity must be performed, and
- the rules to be followed when rescheduling the activity if it cannot be performed at the scheduled time.

The NSSR provides most of the information needed to create a valid GPS satellite contact schedule, while the SORD puts the MCS requirements in the larger context of the entire Global Positioning System. Both documents were obtained directly from the 2 SOPS.

3.2.2 Operational Data. Information describing day-by-day and minute-to-minute MCS operations was required for two reasons. The first was to provide a detailed description of the action and interaction of the six crew positions during each type of MCS activity. This data allows the simulation to include a functional characterization of each position, so that the effects of chauges to that position on total MCS performance can be observed. Second, the stochastic tendencies of the operations center must be determined using operational data. While the scheduling process can use the standard activity performance times described in the requirements documents, the operations simulation requires the observed intervals and durations.

The permanent record of MCS operations in terms of discrete activities is maintained at the MCS in the form of Pass Plans. These are detailed descriptions of the activities performed during each satellite contact, including printouts of system displays at the time the activities were performed. Actual event times can be obtained directly from these displays, and the function of each crew position in the performance of an activity can be determined by matching the time and type of commands executed by that position.

Due to the large number of Pass Plans generated by the MCS, only the past 30 days are maintained at the MCS. Pass Plans for 65 satellite contacts performed on 2-3 June 1992 (J152-153) were collected from the 2 SOPS. The data spans all the 17 satellites then operational and all the satellite contact activities performed on a daily basis.

3.2.3 Historical Data. The operational information described above provides the information needed to structure the MCS model and provide statistical data on the operations process. In addition to this, a case history of MCS performance over some set period of time is required to provide the basis for model validation. If the initial conditions at the beginning of that time period can be duplicated, a valid model should reproduce the actual MCS performance within a predetermined tolerance of error.

Completed MCS activity schedules, together with the crew log maintained by the crew Flight Commander, completely describes the activities performed by the crew. The schedule kept for record is the one the FCMDR annotates with contact results, actual (as opposed to scheduled) contact start time and duration, and contacts moved or added on a contingency basis in response to schedule changes. In addition, scheduled and unscheduled MCS computer and GA outages are recorded on these documents.

The complete set of Navstar GPS Master Contact Schedules and Crew Logs from 31 March to 13 May 1992 (J90 to J134) were obtain from the 2 SOPS.

3.2.4 Crewmember Interviews. In addition to obtaining hard data describing MCS operations, personal and unstructured interviews were conducted with the members of an MCS crew.

The discussions occured with "Bravo" crew; Captain Keith Dale was the Flight Commander. The information obtained, while not directly applicable to the model building process, contributed to the overall direction of this project. The results of the interview is described more fully below.

Discussions with Captain James Serpa, a experienced GPS Flight Commander and crew evaluator for that position, were also used to verify model details and to provide additional "face" validation for the model.

3.2.5 MCS Performance Criteria. A number of operational criteria are used by the 2 SOPS to evaluate the performance of the Global Positioning System. These criteria are:

- satellite vehicle availability,
- Master Control Station availability,
- · ground station availability,
- navigational/timing accuracy (as perceived by the system),
- operations personnel error rate, and
- satellite support success/failure rate.

Each of these criteria has significance to the system managers and operators. However, their significance cannot be easily inter-related, because they impact (or may impact) different segments of the system. Also, each could have either serious or negligible impact depending on the relative need of the user community for the system at the time of the occurrence, and how that community is affected. Finally, rating the relative impact of these criteria on GPS mission performance is difficult because of the interaction of the events. For instance, a personnel error may remove a satellite or ground station from service, then navigational accuracy may be affected when a satellite support is missed. Although the SORD and other local directives set the acceptable limits for system availability and navigation accuracy, the operational squadron by its nature has a "zero tolerance" for deviations from perfect performance for all the above measures.

Despite the difficulties described above, an accurate indicator of GPS operations performance can be obtained. The key is using measurement criteria that are directly affected by crew performance. One of these is the satellite support failure rate. In the terminology of the 2 SOPS, a scheduled support is called "MISSED" if it could not be performed within 60 minutes after the scheduled time, but is completed prior to the maximum allowable interval as specified in the NSSR. A "FAILED" support could not be performed before the time limit, or could not be performed at all (3). Although not officially monitored by the system managers, this discrete GOOD/MISS/FAIL criteria can be extended to a continuous indicator by tracking how much time passes until a MISSED or FAILED support is completed.

An ex officia criteria for evaluating the performance of the MCS is ground antenna utilization. As described above (and with the exception of PIKE), the GPS ground antennas are GPS dedicated resources. Idle time cannot be used by any other system, so there is no managerial pressure to maximize utilization. However, when the number of scheduled contacts per unit time is high and the system has less scheduling slack, real-time schedule changes are more difficult to resolve without MISSED or FAILED supports. Currently there are more than sufficient "GA-minutes" to allow flexible real-time rescheduling (as is demonstrated by the low MCS satellite support failure ate), but as the satellite constellation grows or GAs break, this utilization statistic will become more important to system decision makers. Even now, GA utilization is an adequate measure of the amount of slack GA resource available to the system.

The MCS model described below uses both satellite support success statistics and GA utilization as performance measures.

3.3 Model Development.

After MCS data was collected and reasonably sound measurement criteria and indicators of success determined, the preliminary decisions regarding the type and basic structure of the

MCS model could be made. For each prospective type of model, there are a number of tradeoffs and assumptions. These involve both internal (having to do specifically with the structure and content of the model) and external (concerning the development process) factors. Although the external factors would appear to be irrelevant to the research, they had considerable impact on the methodology and sequence of model development. Once these decisions were made, and after a review of the problem statement and the goals of the project, development began on the model itself.

- 3.3.1 Goals. In keeping with the tentative solution as described in Chapter 1, the overall goals of the model development process were that the model would:
 - 1. Utilize actual MCS-derived data for initialization and operation.
 - Reproduce the scheduling and operations performance of the MCS with maximum fidelity, given the fact that perfect fidelity is not possible due to the non-deterministic nature of the scheduling process.
 - 3. Provide performance measures that are similar to the mission effectiveness criteria used by the MCS management.
 - 4. Accommodate any set of initialization parameters that were consistent with some past or future state of the Global Positioning System.
 - 5. Be sufficiently clear in construction and operation so that maintenance, follow-on research, or operational use could be performed with minimum effort.
- 3.3.2 Model Selection. Among the types of simulation models briefly described in Chapter 2, the computer simulation was chosen to be the most apt. The first reason for this selection was the desire to model stochastic activity durations. The two simulation languages available for this research (Simulation Language for Alternative Modeling (SLAM) and SIMSCRIPT II.5) allowed accurate modeling of this facet of the operational environment.

Another reason for the choice of computer simulation was the lack of clear analytical solution methodology for the scheduling problem presented by the MCS. The scheduling technique used, an iterative approach with sliding time windows and changing priorities, was selected because

it corresponded to the actual MCS scheduling process. This algorithm was not easily modeled mathematically, although an integer programming or branch-and-bound technique might have been theoretically possible. Even if the scheduling process could have been performed analytically, there would be interface problems with the operations segment of the simulation. The scheduler would have to be called real-time when the operations simulation required a new schedule. The external scheduler would then have to read the current state of the simulation and then provide the new schedule in a form usable to the simulation. These difficulties precluded the use of a "mathematically"-based scheduler model and led to the development of the scheduling "simulator" described below.

3.4 Model Description.

3.4.1 Overview. The following is a detailed description of the MCS simulation code. The source code itself is provided in Appendix A). As the MCS.SIM source code is well documented, this description will concentrate on fitting the individual routines into the larger structure of the program, and will describe the rationale and logical basis for the techniques used. This description assumes some knowledge of the SIMSCRIPT II.5 programming language; readers are referred to the language manual (Bibliography reference (7)) for answers to syntax questions.

Coding conventions were standardized and applied consistently throughout the program to allow easier understanding and modification. These conventions are.

- A block of asterisks separate SIMSCRIPT routines and processes,
- Statements affected by conditional (IF) and flow control (DO...LOOP) constructs are indented from the controlling statement for every level of nesting,
- Remarks precede the sections of code they describe and are inset to the appropriate level,
- Local variable names start with a period(.),
- Variables and entities are UPPERCASE.

3.4.2 Initialization Process. This section describes the sequence of events necessary to initialize the simulation. The initial conditions for the simulation run are maintained in data files external to the program. This allows different simulation runs to be made without modifying the simulation source code. These initial conditions include data that describes the MCS state at the desired simulation start time, the desired resource changes that occur during the simulation, and operational parameters that define the limits of the simulation run. The routines described will be the PREAMBLE and the READ.DATA, READ.VIS, INIT.GA.USE, and INIT.ACTS routines.

Reader familiarity with the following frequently used program variables and entity attributes is helpful during the discussion of the program's operation. In some cases, the same attribute names are used with more than one entity. "SV" and "GA" are text attributes that store the names of satellite vehicles and ground antennas. The SV names are each seven characters, starting with either "BII-" or "BI-" (Block I or II), followed by a number indicating the launch order. Satellite numbers greater than 20 indicate the satellites not yet operational at the time data was collected on the system. "GA Index" is an integer assigned to each GA to simplify the manipulation and selection of ground antenna data.

- section is used primarily to define variables (7:45). System resources, entities, queues, and processes are also defined and their attributes described. Of note in the PREAMBLE are DEFINE statements that set the rank ordering of the queues used by the scheduling and operations routines. These commands play a critical role in these processes and will be explained in detail in later sections. The TALLY statements create monitoring subroutines that continuously collect statistics on system variables relating to simulation performance (7:240).
- 3.4.2.2 READ. DATA Routine. This routine opens the default input data file, of the name specified by the user at program execution time. This file contains the filenames of the remainder of the input and output files. The first file in the list (with filename assigned to the text

variable TIME.FILE) is then opened. The desired simulation start and end times (in 1992 Julian day/hours/minutes) are then read, converted to the number of minutes since 0000 hrs on 1 January 1992 (henceforth called "jminutes"), and then stored in the variables SIM START TIME and SIM.END.TIME. The variable CURRENT.TIME is set to SIM.START.TIME; this will not change until after the initial scheduling is performed and the simulation clock starts. The next values read from the file are three simulation parameters. NUM.SIMUL.CONTACTS is the maximum number of simultaneous satellite contacts that can be performed by the MCS during the simulation. Technically, the most the system can handle is four (5:7). However, because there are currently only three SSO's per operational crew, the practical limit is three. The variables MAX.PRIORITY and INTERVAL.OFFSET.STEP are used by the scheduler in managing the rescheduling process. Finally, this routine creates the system resources used in the management of the operations simulation. Refer to Point A on Figure 3.1, which is the first of a number of figures designed to show the operation of thesimulation schematically. Each labeled block represents a data storage structure, either an array or a queue. In the case of queues, the data are stored as SIMSCRIPT entities. The simulation functions primarily by moving these entities among the various queues at the appropriate times. The order in which the entities are maintained and accessed is also important. As each entity has a start time attribute, the entities are generally in forward or reverse chronological order.

3.4.2.3 READ.VIS Routine The READ.VIS routine opens the file indicated by the VIS.FILE text variable and reads data that describe the time intervals when each GPS satellite used in the simulation is visible at each of the selected ground antennas. Each period of visibility is then stored as a temporary entity and placed in a queue called VIS.TABLE. Refer to Point B on Figure 3.1.

Any source of visibility data can be used to provide data for this program. Initially this routine contained the code necessary to create the satellite visibility tables using ground station

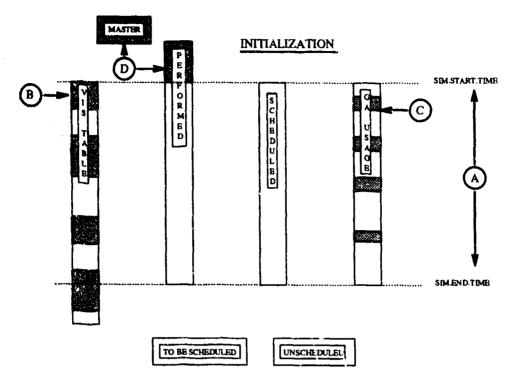


Figure 3.1. Simulation Initialization

locations and satellite orbit element sets. However, the complex and repetitive calculations required by this method proved to be time-consuming to run and difficult to code in SIMSCRIPT. The practical alternative was to create the data externally. The optimum source of visibility data is the MCS-produced visibility charts, especially when attempting to duplicate past MCS performance. However, this data was not available for the time period used by the simulation, so this research used the data produced by Pass Scheduler, written in Pascal for MS-DOS-compatible computers by Kelso (6). This program, using standard two-line orbits element sets and the latitude, longitude, and elevation of the viewing location, calculates satellite rise and set times for a specified time period. The results can be saved as a text file and easily manipulated to conform to the input specifications of the READ.VIS routine (described below). The program Pass Scheduler and its use with this research is described in detail in Appendix D.

The first line of the READ.VIS data file is the number of ground antennas used by the simulation. The second line contains the number of days from the start of the year to the start of the month being simulated, including 1 January and excluding the first day of the month. The visibility data is then read one line at a time, with each line containing the data for one SV/GA visibility period. Each GA list, which can contain any number of satellite visibility periods for each satellite, is structured as follows (values are separated by one or more space):

• Line 1:

- Name of the GA,
- GA Number.

• Line 2 through n:

- SV name,
- Start visibility day-of-the-month,
- Start visibility hour,
- Start visibility minute,
- End visibility hour,
- End visibility minute,

After the final entry for each set of Ground Antenna visibility entries, the word "END" followed by five zeroes separated by spaces end the list for that GA; the next GA set (if any) starts on the next line.

For each line of data containing visibility data read, a VIS.EVNT entity is created. This entity has five attributes: Rise time, Set time, SV name, GA name, and GA index. The rise and set times are stored in jminutes.

3.4.2.4 INIT.GA. USE Routine. The purpose of this routine is to initialize the preplanned times when some or all of the GPS Ground Antennas will be unavailable for use during the simulation. During the validation stage, this capability allows the simulation to model the historical MCS and GA outages. For experiments, the outage data provides a significant test variable for evaluating system response. MCS total-system outages can be simulated by making all Ground Antennas unavailable for the duration of the outage. Refer to Point C on Figure 3.1.

The desired outage times are stored in the datafile pointed to by the OUTAGE.FILE text variable. The first two lines of the file are text strings. These strings allow the user to describe the contents or use of the file; the strings are reprinted at the end of the simulation along with the results of the simulation. The next line is an integer value indicating the number of outage description lines to follow. Each line contains (separated by spaces):

- the GA index number,
- the GA outage start time in jminutes,
- · the outage duration in minutes,
- the reason for the outage, in one word.

To store the above data, a two-dimensional matrix called GA.USAGE is created. The first index is the GA index number; the second is the simulation time, in minutes since simulation start. Each entry in the matrix is a text value (a "string" of characters) describing how each GA is being used at each specific minute. These strings replace the null strings (text variables containing no characters) which are the default matrix entries. For each minute the GA is planned to undergo maintenance or be otherwise unavailable, the value of the matrix cell for that minute and that GA will contain the one-word description of the outage. Unreserved minutes will contain a string of ten space characters, which are better for output formatting than null strings.

A number of techniques were considered to maintain the GA use record, including an entity/attribute list similar to the VIS.TABLE and the use of system resources to manage GA utilization. The large number of table look-ups required made the entity method too slow, while the large number of resource entities required (one for each GA-minute for the duration of the simulation) stretched the storag? capacity of the VAX-implemented SIMSCRIPT software. The matrix

scheme described above is the result of a compromise between processing speed, storage capacity, and programming convenience.

3.4.2.5 INIT.ACTS Routine. In order for the simulation scheduling routine to schedule new MCS activities conforming to system requirements, information on previously performed activities must be known. In this simulation, each distinct task performed during a contact for a specific satellite is called an "ACTIVITY", or "ACT". Each ACT is a SIMSCRIPT entity. The details of the ACT required for the scheduling and operations functions are maintained as attributes of the ACT entity. This routine reads contact history data contained in the file pointed to by the VIS.FILE variable. The data describes the name, start time and other details of the most recent (prior to the simulation start time) occurrence of each type of contact activity. For instance, only the description of the most recent navigation upload activity for satellite BII-010 would be included in the input data file. Refer to Point D on Figure 3.1.

After opening the specified file for input, the routine reads first the number of satellites being simulated. then the number of individual activities to be read. A permanent entity called a MASTER.ACT is created for each of the activities to be read; the input data is then assigned to attributes of these entities. The activity attributes fall into two categories: those that remain unchanged during the life of the entity, and those that are modified during the simulation. The fixed attributes are:

- NAME: Text label describing the function of the activity. For the basic simulation, the labels are:
 - NAV Navigation data upload,
 - SOH Satellite state-of-health determination, using the satellite telemetry data,
 - NAVBUFF Navigation processor data buffer dump,
 - GBDDUMP Global Burst Detector processor data dump,
 - NDUTLM Navigation processor real-time telemetry examination.
- SV: Text label indicating the satellite vehicle pertaining to this activity. The label consists of the the Block number, a dash, then the satellite launch number. Future satellites have launch numbers greater than 20.

- BLOCK: The satellite Block number, either the integer 1, 2, or 3. Research and development vehicles are Block I (1), the first series of operational satellites are Block II (2), and the later version of the Block II SV is labeled (3).
- DURATION: The expected duration of the activity in integer minutes.
- VARIANCE: The variance of the activity duration. This real value is used in the operations sin. Sion process, not in the scheduling process. Defaults to 1.0
- INTERVAL: The maximum interval between occurrences of this activity, in minutes.
- CRITICALITY: This integer value describes the impact of this activity on MCS mission completion. Planned to be used as another tiebreak criteria for scheduling conflict resolution, a system not currently implemented in the simulation code. Defaults to 1.

The dynamic activity attributes are:

- GA, GA.INDEX: After the activity is scheduled or performed, the text attribute GA stores the abbreviated name of the location of the Ground Antenna that did or will perform the contact. The GA.INDEX attribute desig. tes each Ground Antenna with an unique integer number. The values used in the simulation are:
 - ASCN (1)- Ascension Island
 - CAPE (2)- Cape Canaveral
 - DIEG (3)- Diego Garcia Island
 - KWAJ (4)- Kwajalein Island
 - -- PIKE (5) Falcon AFB, Colorado (PIKE is the AFSCN designation)
- START.TIME: This is the time of either the scheduled activity performance (if after the current time) or the actual performed time (if prior to current time). The integer value is in jminutes.
- NEXT.START.TIME: This attribute is used to hold the tentatively scheduled start time of the next occurrence of this activity during the scheduling process. The format is the same as START.TIME.
- PRIORITY: This integer number is the primary means for deconflicting activities during the scheduling process. The default is 0, which is incremented if this activity conflicts with another while being scheduled.
- INTERVAL.OFFSET: The scheduling process will allow activities that are difficult to schedule to exceed the maximum time between activity performance, if changing the activity priority alone does not resolve the scheduling conflict. This attribute stores the current value (in minutes) of the amount of excess interval allowed. Default=0.
- LEAD.TIME: When an activity is selected to be scheduled, the time remaining from CUR-RENT.TIME to the next expected START.TIME is calculated (the units are minutes) and stored in LEAD.TIME. This value represents the "urgency" of scheduling the next activity of this type, and can be used to prioritize the activities waiting to be scheduled.
- STATUS: This text attribute describes the current status of this activity. The permissible values are:

- UNSCHD Unscheduled
- SCHD Scheduled
- TENT Tentatively Scheduled
- RESCHD Reschedule
- MISSED Activity has exceeded maximum PRIORITY
- FAILED Activity has exceeded maximum INTERVAL.OFFSET

The newly-created MASTER.ACT entities are then filed in a queue labeled MASTER; these serve as a permanent record of the original activities and as a tool for indexing when temporary entities representing activities must be accessed. A set of temporary entities identical to the MASTER.ACT entities is also created at this time and filed in the queue PERFORMED. These form the starting reference for the scheduling routine, representing the activities performed just prior to the start of the simulation.

3.4.3 Scheduling Process. This section of the MCS simulation code creates a valid MCS schedule for the period of time from the current simulation time to the end of the simulation. The MCS schedule is a chronological list of activities that are to be performed. While this simulation only pertains to satellite contact activities, the real MCS schedule also lists equipment outages for maintenance, system housekeeping activities, scheduled communication requirements, and so on.

For this application, a "valid" schedule is a satellite contact schedule that satisfies MCS operational constraints and either allows all the NSSR satellite contact requirements to be met or flags those requirements not attained. This models operational reality, as the MCS constraints are fixed by the MCS hardware (i.e., number of Ground Antennas) or operational software (i.e. maximum number of satellite the system can operate). On the other hand, the NSSR-specified requirements can often be stretched"; generally there is no concrete and immediately deleterious effect of exceeding the maximum interval between activities called out by the NSSR. Tradeoffs can (and often must) be made, such as delaying a STATE OF HEALTH activity on a reliable satellite (until after its maximum interval) to allow a contingency NAV UPLOAD activity to correct

navigation signal errors. The result is that late or missed activities do not invalidate a schedule, as long as the deviations from the system requirements is known.

In addition to the above requirements, there are other aspects to the schedule creation process that must be considered. The system requirement documents rarely specify the minimum interval between activities, so a valid schedule could be created that would be inefficient due to activities being too frequently performed. So activity intervals should be maximized inside the bounds allowed by system requirements. Another consideration is the question of priority. Given limited satellite contact resources (time, antenna availability, satellite visibility, etc.), there will be instances where satellite activity requirements for one or more satellites cannot be met. Which activity has priority for the available resources?

The scheduling process addresses these issues. Three routines support the scheduling function:

MAKE.NEW.ACT, PRESCHEDULE, and SCHEDULE.

3.4.3.1 MAKE.NEW.ACT Routine. The smallest unit of schedulable time in this simulation is represented by the ACT entity. As will be seen below, the presence of this entity in a particular queue is the primary mechanism of the scheduling process. As the reproduction of an ACT entity is a common requirement in the process, it was efficient to create a generic routine to perform this function whenever required.

This routine is passed the pointer of the entity to be duplicated by the calling routine. A new ACT entity is then created, its attributes identical to the original. The pointer to the new entity is then passed back to the calling routine and this routine ends.

3.4.3.2 PRESCHEDULE Routine. Fundamental to the scheduling process is knowing which activities need to be scheduled. This routine selects the activities (if any) that must be performed at least once more prior to the end of the simulation, based on the last time it was performed or scheduled to be performed. If there are activities still to be scheduled, this routine

then sets up the system queues to the configuration expected by the SCHEDULE routine and calls that routine. If there are no activities requiring scheduling, it prepares the system for the operations simulation process and exits. The PRESCHEDULE routine uses the activities in the MASTER.ACT queue as means of indexing through every SV/activity pair exactly once. As described above, there is one entry in this queue for each activity required by each SV (for instance, SV BII-010 will have NAV, NAVBUFF, GBDDUMP, and SOH activity entity in MASTER.ACT). The INIT.ACTS routine also placed a copy of each of these entities in the PERFORMED queue, which is rank ordered by latest (i.e., numerically largest) START.TIME. For each activity in turn, the queue PERFORMED is searched for the *latest* occurance of that activity (Point A in Figure 3.2).

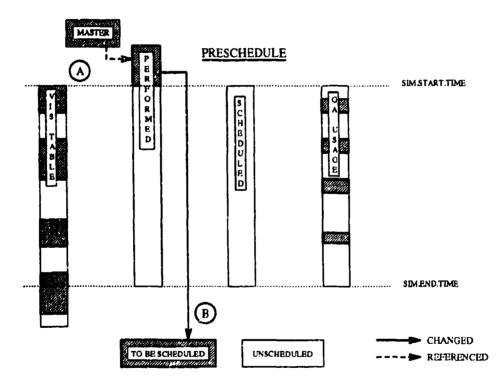


Figure 3.2. Prescheduling Routine

The status of the found entity will depend on when in the simulation PRESCHEDULE has called. At simulation start, the latest ACT entity is one of those loaded at initialization, with a START.TIME prior to the SIM.START.TIME. If the simulation has been running and a new

schedule is required, the latest ACT would be one that the operations simulation has just performed, with a START.TIME indicating the simulation time the activity was started. Once the desired ACT entity with latest START.TIME is found, it is evaluated to determine if another activity of this type for this SV is required in the time frame of the simulation. This will be true if the activity interval plus the interval offset plus the activity duration is less than the time remaining from the last occurrence of this activity to the end of the simulation. This is shown graphically in Figure 3.3. In the figure, new activity "A" will be scheduled, while "B", the same activity but with a larger INTERVAL.OFFSET, would not be scheduled. Adding the activity DURATION to the previous START.TIME prevents the system from scheduling contacts that cannot be completed in the simulation time interval.

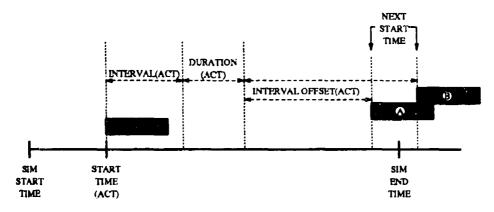


Figure 3.3. Activity Selection Timeline

If the activity must be performed again, a copy of the previous ACT entity is made. This new ACT has its STATUS attribute changed to "UNSCHD" and its LEAD.TIME calculated. The values of PRIORITY and INTERVAL.OFFSET are collected for statistical analysis, then reset to zero. Finally, the new ACT entity is filed in the TO.BE.SCHEDULED queue (Point B in Figure 3.2).

If the above steps have been completed without finding a candidate activity to schedule, the TO.BE.SCHEDULED queue will now be empty (it is always empty when PRESCHEDULE is called). If so, the scheduling process is complete; all the tentatively scheduled activities have their

STATUS changed to "SCHD" and are moved to the SCHEDULED queue. The entities representing "FAILED" supports remain in the SCHEDULED queue, indicating a failed attempt to schedule an activity for the time near the value of START.TIME. If there are entities in TO.BE.SCHEDULE at the end of PRESCHEDULE, the routine SCHEDULE is called to attempt to fit these activities into the tentative schedule.

determined, they must be scheduled in accordance with the requirements described in the NSSR and the constraints driven by system resources. In addition, the activities should be scheduled smartly, with consideration given to the efficient use of system resources. The scheduling process is basically sequential trial-and-error. The activities to be scheduled are examined one-by-one and the optimum time to next perform that activity is determined. Then, starting at that time and moving toward current time, the system satellite contact resources are scanned for a suitable starting time. At the first instance when all the necessary resources are available to perform the activity, the search process stops. The activity is then tentatively scheduled for that time and resources are reserved.

If any one of the current activities waiting to be scheduled cannot be scheduled in the allowed time window, all activities, both tentatively scheduled and still waiting, are returned to be scheduled again. However, the activity creating the conflict is increased in priority so that it gets the first opportunity to reserve the needed resources. The entire process is then repeated. The next escalation in conflict resolution occurs when, due to repeated conflicts, the priority of any activity has been increased to a predetermined threshold. At that point, the process "gives up" trying to find a spot in the schedule that meets the maximum interval requirement. It now tries to schedule the activity to minimize lateness. The search interval for that activity is expanded slightly and the schedule process repeated. Finally, if the lateness of the activity reaches a set maximum value without the activity being successfully scheduled, the scheduling routine simply

stops trying. It inserts a specially-tagged ACT entity to hold the place of the missing activity and continues to schedule without the troublesome activity. The implementation of the above scheme in SIMSCRIPT is described in detail below.

The first step is to remove the first ACT entity with a STATUS of "UNSCHD" from TO.BE. SCHEDULED. Which entity is first depends on the rank ordering for this queue as defined in the PREAMBLE. During simulation validation, the ordering was optimized to minimize the difference between the activity start times as scheduled by this routine and the times the activities were actually scheduled by personnel at the MCS. The closest match occurred when the entities in TO.BE.SCHEDULED were ranked by their high PRIORITY, then low INTERVAL, then low START.TIME attributes. After the entity is removed from TO.BE.SCHEDULED, its attributes are referenced using the entity pointer called ACT for the remainder of the routine. This process is shown as point A on Figure 3.4.

Now the time period over which the "resource space" is searched is determined. This interval begins at START.TIME(ACT) (which is the last time this activity was started for this SV), plus the INTERVAL(ACT), plus the INTERVAL.OFFSET(ACT). This is the *latest* simulation time at which this activity will be tentatively scheduled. Then stepping back to either START.TIME or CURRENT.TIME, whichever is later, each minute is tested as a candidate starting time for this activity. The current candidate minute is stored in the variable .CNCT.START, shown as Point B on Figure 3.4.

Once a possible contact start time is set, each VIS.EVNT entity (which define the GA/SV visibility periods) in VIS.TABLE is tested for: 1) applicability to the SV of this activity, 2) visibility that begins prior to the tentative contact start, and 3) visibility that ends after the duration of the activity. The attributes of the first VIS.EVNT entity to meet these requirements (see point C on Figure 3.4) are referenced in the remainder of the routine by the pointer VIS.EVNT. Most

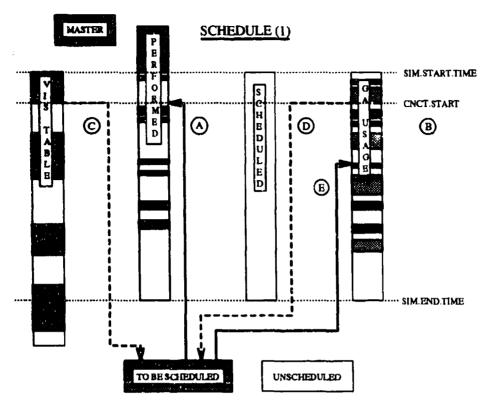


Figure 3.4. Scheduling Process (1)

importantly, the VIS.EVNT selected determines the GA (name) and GA.INDEX at which the tentative schedule entry will be performed.

The last check to be made is to ensure the Ground Antenna referred to by the VIS.EVNT is not reserved by some other activity or scheduled event (Figure 3.4, point D). The status of each GA is maintained in the GA.USAGE two-dimensional text array. The first index the GA number and the second a count of the simulation time in minutes. The minute index of the array runs from 1 to SIM.END.TIME - SIM.START.TIME + 1, so a correction is needed to convert simulation time to the array index. The array index, .OFFSET, corresponding to the simulation minute being examined is equal to .CNCT.START+.t-(SIM.START.TIME-1). One must be subtracted from SIM.START.TIME to correct for the fact that the simulation starts at time zero but the GA.USAGE array starts with index "1". To test each minute from the prospective contact start time through

the duration of the contact, the counter ".t" is incremented from zero to the DURATION of the ACT, as long as the .CONFLICT flag, earlier set to zero, stays unchanged. For each minute, the following occurs:

- 1. The target GA is tested for previous use,
- 2. Every GA is tested for a previous reservation for the SV being scheduled,
- 3. The total number of GAs reserved for satellite contacts is tested for exceeding the number of simultaneous contacts (NUM.SIMUL.CONTACTS).

If any of the tests are true, the .CONFLICT flag is set and the .t loop ends immediately. This condition results in the program flow bypassing the "no conflict" structure (described below). The next VIS.EVNT in the VIS.TABLE that describes favorable SV/GA visibility (if any) is referenced. If there is dual (or triple) visibility, the program again examines GA.USAGE for those other As at this simulation time. If all GAs visible to this SV turn out to have no free time for this contact, the VIS.EVNT loop ends, the .CNCT.START counter is decremented, and the entire process repeats for the simulation minute prior to the previous. This process will continue until all possible contact start times have been examined, or until a place is found in the schedule for this activity.

If at any time in this process a suitable place in the schedule is found, the attributes of the ACT entity representing this activity are modified. The lucky .CNCT.START time is stored in NEXT.START.TIME, the GA name and GA.INDEX (from the VIS.EVNT entity) are stored in like-named ACT attributes, and STATUS(ACT) is changed to "TENT". When the STATUS is changed, the do-loops stepping through the .CNCT.START times and the VIS.EVNT list stop indexing and fall though their "loop" statements to the remainder of the routine. Lastly, the SV and NAME attributes from the successfully (but tentatively) scheduled activity are written into the GA.USAGE array for the scheduled times, reserving this resource for this activity (Figure 3.4, point E).

If the process described above completes without finding a place in the schedule for this activity (shown as Figure 3.5, point A), further steps are called for. First, if the current PRI-ORITY of this entity is less than the maximum, the PRIORITY is incremented by the current value of .PRIORITY.STEP. This variable increases as the number of activities not being scheduled increases, which adds variability to the prioritization process. The entity's STATUS is then changed to "RESCHD" (to allow tracking) and the .RESCHD.FLAG is set, which later directs the process to bypass the PRESCHEDULE routine and repeat SCHEDULE with the current but subtlety different list of activities. If this ACT entity's PRIORITY is at or greater than maximum, its STATUS is changed to missed and its INTERVAL.OFFSET is incremented by the value of INCREMENT.OFFSET.STEP. This expands the search range of the schedule process, with the understanding that the maximum interval between activity performances has been exceeded in this case, and the critical action is to schedule the activity as soon as possible.

If after the above operation the INTERVAL.OFFSET value has not exceeded one-half the original INTERVAL, this activity is placed back into the TO.BE. SCHEDULED queue (Figure 3.5, point B) to try to find a place in the schedule the next time SCHEDULE is called. If the maximum interval offset has been exceeded, the entity STATUS is changed to "FAILED" and the FAILED variable is incremented. The activity's START.TIME attribute is set to the optimum simulation time for this activity to be performed, had it been able to be scheduled. Then a copy of this entity is made (using the MAKE.NEW.ACT routine) and stored in the PERFORMED queue. This holds the place for this activity and allows futher activities of this type to be scheduled. With a STATUS of "FAILED", this entity will not be moved to the SCHEDULED queue to be performed by the operations simulation process. The original ACT entity is stored in the UNSCHEDULED queue (Figure 3.5, point C), so that it's attributes can be examined during the output phase of the program.

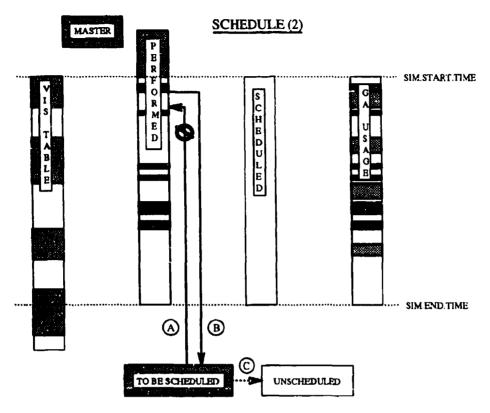


Figure 3.5. Scheduling Process (2)

If the above process completes without scheduling conflicts for any reason, the tentatively scheduled activities in the TO.BE SCHEDULED queue have their new start time (stored in NEXT.START.TIME) copied to the START.TIME attribute. The STATUS is set to "TENT" and the entity moved to the PERFORMED queue, where it becomes the basis for planning the next batch of new activities. Now the TO.BE SCHEDULED queue is empty, the new set of activities are in PERFORMED (but marked as tentative), and the PRESCHEDULE routine is called to determine whether further scheduling in necessary. As noted in Section 3.4.3.2, if no further activities require new contacts, the tentatively scheduled ACT entities in PERFORMED are all promoted to "SCHD" STATUS and moved to the SCHEDULED queue for input to the operations simulation segment (Figure 3.6).

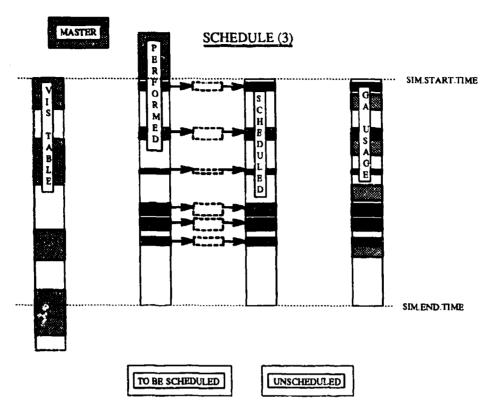


Figure 3.6. Scheduling Process (3)

On the other hand, if scheduling conflict did occur, the attributes of the tentatively scheduled a tivities in TO.BE.SCHEDULED are used to erase their GA reservations in the GA.USAGE array. They then are marked as unscheduled and the system is back to the state it was in when SCHEDULE was first called, except that the conflicting activities either: 1) have new PRIORITY and therefore will be scheduled in a different order, 2) have a larger INTERVAL.OFFSET, allowing a longer simulation time period to be searched, or 3) have been marked as "FAILED" and removed from scheduling consideration. At this point SCHEDULE is called again and the "old" entities in TO.BE.SCHEDULED try again for space on the schedule.

3.4.4 Operations Simulator The task of the operations simulation segment of the MCS Simulator is simply to perform the MCS schedule as devised by the scheduling process, Given

the complexity of the activities performed and system performing them, and the fact that human beings are "in the loop", some variability in the performance of the scheduled activities is to be expected. The operations simulation, in attempting to emulate the function of the MCS, adds that uncertainty. The function is performed by two processes: START.OPS and PERFORM.ACT.

3.4.4.1 START.OPS Process. This segment of the MCS Simulation consists of two processes. The first, called START.OPS, takes no simulation time to perform its function and occurs once every simulation minute. START.OPS first updates the variable CURRENT.TIME by adding the time from simulation start to SIM.START.TIME. It then checks for available CONTACT resources; there will be NUM.SIMUL.CONTACTS (provided by the user at initialization) of these resources. If one or more CONTACTs are available, the next scheduled activity is selected (Point A on Figure 3.7) and handed off to the PERFORM.ACT process (described below). This next scheduled activity is the ACT entity in the queue SCHEDULED with the lowest START.TIME. If the simulation end time has not been reached, this process then schedules itself to occur again in one simulation minute.

3.4.4.2 START.OPS Process. Once an activity has been selected to be performed, the PERFORM.ACT process simulates the execution of that activity. The first step is to reserve one of the CONTACT resources for the duration of the process. The actual duration is then determined. If statistical deviation from the standard duration for this activity is desired, the DURATION and VARIANCE attributes of this ACT entity are used in conjunction with internal SIMSCRIPT statistical distribution functions to calculate a new activity duration. This value is maintained locally in the .DURATION variable, and is also written back into the DURATION attribute of this entity for "permanent" record.

Next, the scheduled start time of this activity is compared to the CURRENT. TIME. If the START.TIME(ACT) is greater the the current time, the process waits until the activity start time to continue. If the activity scheduled start time has passed, the process checks the VIS.TABLE

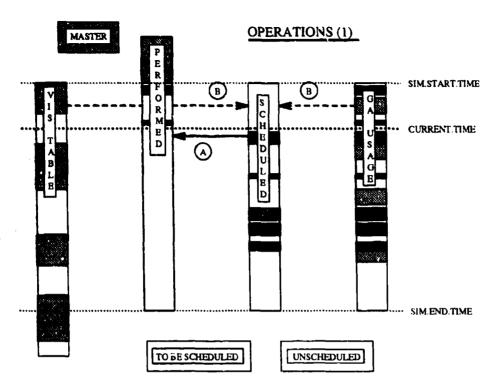


Figure 3.7. Operations Process (1)

queue and GA.USAGE array for sufficient visibility and free time at the Ground Antenna scheduled for this activity (labeled B on Figure 3.7). If enough visibility and GA time is available to perform the activity beginning at the current time, then the process continues and the scheduled activity is performed starting at the current time. If the activity cannot be performed now due to lack of GA visibility or free time, .OFF.SCHEDULE is set to 1.

If this activity can be performed at this CURRENT.TIME, the ACT entity start time is changed to the current time, the STATUS is changed to "PERF", and the process waits the number of minutes specified by the DURATION attribute. After the wait is completed (DURATION minutes later), the process returns the CONTACT resource to the common pool and files this ACT entity in the PERFORMED queue. If the OFF SCHEDULE flag is set, the current activity cannot be performed at this time, and the current schedule is no longer valid. A new schedule starting at the current time and including the activity that caused the problem is required. So

for every entity in SCHEDULED with start times at or after the current time, the GA.USAGE reservation for that activity is erased and the entity destroyed (Figure 3.8). After all are removed, the PERFORM.ACT process relinquishes its CONTACT resource and the PRESCHEDULE routine is called. PRESCHEDULE will start the scheduling process at the CURRENT.TIME, using the activities already performed by the operations simulation process as "history".

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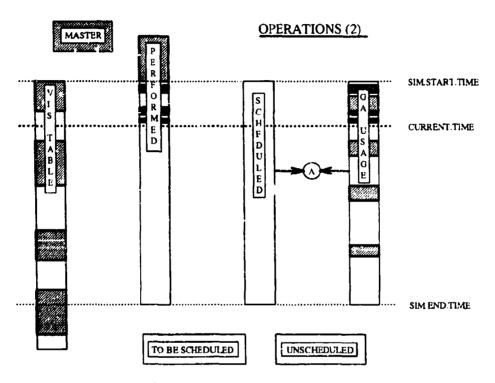


Figure 3.8. Operations Process (2)

3.4.5 Output. Once the scheduling and/or operations processes have completed, the statistics reflecting the performance of the simulation are calculated and saved for analysis. There are five distinct output routines, to allow the simulation operator to tailor the type and amount of output to the current need.

3.4.5.1 REPORT.VIS/REPORT.QUEUES Routines. The first routine dumps the attributes of the VIS.EVNT entities in the VIS.TABLE queue. The rise and set times of the satellite

visibility is converted back to days, hours, and seconds from the jminute format of the RISE.TIME and SET.TIME attributes. REPORT.QUEUES performs the same function for the SCHEDULED, TO.BE. SCHEDULED, PERFORMED, and UNSCHEDULED queues. However, only selected attributes of the ACT entities in these queues are reported. In both cases, the data is written to the SIMSCRIPT default output file.

3.4.5.2 REPORT. USE Routine. The REPORT. USE routine outputs the contents of the GA.USAGE array. The result is a table that shows the minute-by-minute use of each of the Ground Antennas for the entire simulation period. This report also goes to the SIMSCRIPT default output file.

3.4.5.9 ANALYSIS Routine. This routine provides the primary data on the results of the simulation run. It makes extensive use of SIMSCRIPT's automatic statistical collection facilities. The output of this routine goes to the file whose name is the contents of the text variable OUT.FILE, which is set during initialization.

First, each minute of the GA.USAGE array for each GA is scanned for the value of the array variable. Separate sets of variables are maintained for individual GAs and for the overall usage. The system automatically calculates statistics for these variables through the use of TALLY statements. A list of simulation parameters are then printed to permanently label the results to follow. The remarks in the GA outage data file are read and printed to further describe the starting conditions for the simulation. A table is then printed listing the number, maximum, minimum, and standard deviation of four system variables being monitored by TALLY statements. "OFFSET" and "PRIORITY" are collected as the activity entities in the PERFORMED are evaluated for the need for further scheduling in the PRESCHEDULE routine. "GA UTIL" and "GA RESV" are the statistics for the variables UTIL and RESV and were collected earlier in this routine. Histograms for the OFFSET and PRIORITY variables, and then the individual and total GA statistics are reported.

scheduling results at a specific time in the past to the actual MCS schedule (produced by scheduling personnel at the 2 SOPS) for that time period. This validation data is the entered from MCS logs into a file called "valact.dat". The format of the file is the same as that for the initial activity data read during initialization, except that the GA at which the contact was historically performed is included with the other activity information. The activities it contains are the first of each type after the time set as the simulation start. The VALIDATE routine first opens the validation data file and the first activity record is read. The activity name and SV from the validation activity are used to search for the first (earliest) similar activity in the SCHEDULED queue. As there is a one-to-one correspondence between the validation activities and the scheduled activities, a matching activity will be found. The absolute value of the difference between the historical and simulation activity start times is recorded. In addition, a counter is incremented when the historical and simulation-scheduled GAs match.

3.5 Validation

The first validation method for the MCS model relies on the expertise and experience of the model-builder. Each of the subfunctions of the model were created to duplicate the corresponding function in the MCS: Initialization, Scheduling, Operations, and Reporting. At each stage in model creation, the builder tested the subfunctions and compared the result to first-hand knowledge of how the like MCS function operates.

The second methodology for the MCS Simulation validation will use detailed MCS data for a period of time prior to and after some fixed date in the past. This date becomes the validation datapoint. The data prior to validation time is required to "hot start" the simulation in any case, but it also allows the simulation to start with the exact state of the MCS at that instant. The

MCS data after the validation time is used to provide the simulation the identical environmental conditions experience! by the MCS, throughout the run of the simulation.

Once the simulation has been run, under conditions matching those experienced by the MCS for the same time period, the simulation schedule is compared to the actual performance of the MCS. Quantitative measures are taken of the differences between the MCS and simulation results. The magnitude of the differences are an indication of the simulation's scheduling "fidelity", that is, its ability to reproduce the MCS generated schedule. An exact reproduction of the MCS schedule would result in zero difference. The measures do not indicate the absolute scheduling efficiency or quality of either the MCS personnel or the simulation. The parameters of the simulation can be adjusted to maximize the simulation fidelity (by minimizing the measures), but considering the vast number of possible schedules that meet all the scheduling requirements, a perfect match is not likely.

3.6 Experimentation

Once its validity has been tested, the MCS Simulation will be used to examine the probable response of the MCS to changes in the operating environment. The applicability of the simulation results to the actual system largely depends on the accuracy of the scheduling process. If after the validation step the differences between the simulation and human scheduler are zeroed, then the experimental results could be assumed to be representative of the actual system response under the test conditions. This will not be the case, so the best that can be assumed from the outset is that the test results are consistent among themselves. If this can be demonstrated, then trends observed between experiment results can be assumed to be true also for the MCS, even if the absolute results cannot be compared.

The input parameters used for the experiments are the number of operational satellite vehicles and the number of operational GPS Ground Antennas. The SVs will vary between 16 (the number

used for validation) and 24, the maximum number of operational vehicles currently planned. There is no plan to add to the current suite of five GAs, so the number of operational GAs used for testing will vary from three to five. Each combination of three and four operational GAs will be tested. As time allows, alternative scheduling priorities will be tested to determine their affect on simulation results.

As described in Section 3.4.5.3, the simulation results will be a measure of the scheduling effort and the GA utilization. These results will be tested for consistency, correlation, and trends.

IV. Analysis and Results

4.1 Overview.

During the process of coding the simulation model described in Chapter 3, care was taken that the flow of the program emulate the MCS operational process as closely as possible. There were two primary reasons. The first was that this process was well understood, and by mapping the flow of the code to the function of the MCS, it could be clearly seen that the program was executing as expected. Secondly, having the program execution process model the MCS operations process provided reassurance that the simulation would duplicate the results of the MCS with a certain minimal accuracy level. The first reason is related to verification, the second to validation (11:11). The results of these two critical modeling steps are described in this chapter.

Once the simulation was completed, verified, and validated, a set of experiments were designed to exercise the model in circumstances similar to those encountered by the MCS and provide data for analysis.

4.2 Verification.

In the verification process, the concern is the internal consistency of the model (15:210). This process ensures the model has been constructed properly, according the the rules of the chosen modeling environment. For physical models, this may entail re-measurement of critical dimensions. In mathematical models, perhaps supporting proofs are re-examined for validity. With computer simulation models, the verification process includes a code review and "debug" to ensure the program parameters and logic are as intended.

4.2.1 Verification Analysis. For this computer simulation, verification includes an examination of the SIMSCRIPT II.5 source code to ensure there are no numerical or logical errors. This was performed at the completion of each SIMSCRIPT module, then again just prior to the start

of experimentation. As described earlier, this task was simplified by designing the initialization routines to maximize the amount of data stored external to program. For the working data created by the program during it's operation but not routinely output, additional print statements were included in the code during development and validation to make these values external. This allowed not only verification of the numerical correctness of internal variables, but also allowed the program logic to be examined in greater detail and verified. Most of these statements were removed prior to the experiment stage but their vestiges remain in the various REPORT routines retained in the operational code.

As an example of the type of data collected during verification and validation, the results of one early prototype run is shown in Tables 4.1 and 4.2. The first table shows the SIMSCRIPT modules announcing they had been called. When the routine SCHEDULE was manipulating activity entities, the routine printed each action, the current simulation time, and entity attribute data.

Table 4.1. Scheduling Verification/Validation Sample Data.

INIT.ACTS PRESCREDULE SCHEDULE PREVIOUSLY: SYEO8/Supp.1 (15/ 360) scheduled for 323990 new act: SVEOS/Supp.1 (15/ 360) scheduled for 324350 HOY: (15/ 360) scheduled for 324350 at EWAJ \$V#08/Sapp.1 PREVIOUSLY: SVEOS/Supp.4 (12/ 360) scheduled for 323990 new act: SVE08/Supp.4 (12/ 360) scheduled for 324337 HOV: SVEOS/Supp.4 (12/ 360) scheduled for 324337 at KWAJ PRESCRIEDULK SCHEDULE PREVIOUSLY: SVH09/Supp.1 (15/ 360) scheduled for 323980 new act: SVE09/Supp.1 (15/ 360) scheduled for 324340 HOY: 8VB09/Supp. 1 (15/ 360) scheduled for 324340 at DIEG HOV: SV#10/8upp.1 (15/ 360) scheduled for 324330 at ASCE PREVIOUSLY: SVE10/Supp.4 (12/ 360) scheduled for 323970 new act: SVE10/Supp.4 (12/ 360) scheduled for 324289

Table 4.2 shows one way the operations simulation process was verified. As each activity was assigned to a different segment of the operations task flow, output statements in the code printed the time of the activity status change and details about the function and the entity. As seen in

Table 4.2. Operations Verification Sample Data.

324000 ASSIGE: Supp.4 , SVE26 at PIEE (6), 324012/12 minutes WAIT: Supp.4 , SVE26 at PIRE (5), 324012/12 minutes 324000 324000 , SVE25 at EVAJ (4), 324013/12 minutes ASSIGN: Supp. 4 324013 PERFORM: Supp.4 , SVE25 at EWAJ (4), 324013/12 minutes PERFORM: Supp.3 , SVE20 at CAPE (2), 324014/10 minutes 324024 DONE: Supp.4 , SVE26 at PIKE (5), 324012/12 minutes 25 Rescheduling REPORT USE: Current time = 324024

the last two lines of Table 4.2, the operations routine also notified the operator when it required a schedule change, then printed the status of the system queues.

4.2.2 Verification Results. The model is technically verified, in light of its error-free compilation and execution. Since all the expected scheduling and operational steps occurred as planned, this verifies that the program logic is as designed.

Validation.

The goal of the validation process is to determine the fidelity of the model, how well it reproduces the function of the modeled system in terms of the chosen measurement criteria. The MCS Master Contact Schedule is the master plan for all scheduled MCS activities, so the central theme of the MCS model validation process is how well the model reproduces this key document. In addition to this validation method, the overall performance of the model was evaluated by an MCS-experienced operator. This "face" validation was top-down, as it occurred at each step of the model building process.

4.3.1 Validation Analysis. Two validation principles were applied to test whether the MCS model faithfully reproduced the performance of the actual MCS. The first was applied both during the model-building process and after the model was completed. This was basic face validation. This test asks if behavior of the model agrees with that of the real system, in the opinion of knowledgeable experts (4:204). In this case, the system expert is also the model-builder. The

expertise was gained through three years of association with the GPS MCS, as both an operational crewmember and subsystem manager. This experience provides the a priori knowledge-base from which the model was designed and constructed. This knowledge and experience is also applied to the logical analysis of the model's output. During testing and experimentation of the MCS model, the data provided by the model was continually examined for aptness and reasonability. The model passed these validation tests.

As a second, more quantitative means of testing the validity of the model, a schedule produced by the model for the test date of 10 April 1992 was mathematically compared to the schedule produced and used by the MCS on that date. The hypothesis was that a model that duplicates the MCS scheduling process perfectly would produce the same schedule as the MCS, given identical starting conditions. From the outset it was understood that perfect fidelity is theoretically possible but practically very unlikely. This is due to the very large number of alternative schedules that meet all the criteria of a valid schedule. Often in practice, when faced with equally valid scheduling choices, the human scheduler will apparently decide the result arbitrarily. There may be some personal rule the human uses to choose between two similar choices, but the simulated scheduler has no chance to duplicate either some subtle (and unique) rule or a "mental coin-flip".

Compounding this problem, once a deviation is made in the schedule, the differences tend to cascade with time. For example, the human scheduler at 1000 hours has the equally valid option of using either CAPE or ASCN ground antenna for a 30 minute activity on an SV. She chooses ASCN by some arbitrary means. As the scheduling process continues, an activity on another SV with ASCN visibility is needed at 1020. The earlier choice of ground station now has a bearing on the current decision, as ASCN is no longer available for use. Had the earlier choice been CAPE, the alternative schedule would have ASCN available. Over time, similar small changes eventually result in two quite different schedules, all due to the decision made earlier in the process Slight deviations in the scheduled time of activities also has the effect of causing subsequent changes. For

these reasons, it is not expected that the MCS model can duplicate the scheduling performance of the MCS schedule makers exactly.

(It is important to note, however, that the model scheduling algorithm has the capability of producing better schedules than the human scheduler. A better schedule may make more efficient use of system hardware resources or personnel; may work around system resource outages with fewer "tardy" activities; or may even produce the same quality schedule in less time. Further testing would be required to ascertain the relative effectiveness of automated scheduling methods.)

Given an inherent and unavoidable lack of fidelity, the validation test was constructed to detect the schedule differences over a short time span on the date selected. First, the activities of the prior day (9 April) were collected from the Master Contact Schedule and entered into the initial activity file (described in Section 3.4.2.5). Only the 14 SVs operational on 9/10 April were used. Next, the scheduled GA and MCS outages that occurred on 10 April (also taken from the Master Contact Schedule) were entered into the OUTAGE.DAT file. These resources were "worked around" by the MCS scheduler and would also have to be handled by the model. Finally, the data on first activity of each type for each SV as really scheduled on 10 April was entered into a validation activity file. No further activities would be useful due to the compounding error effect described above.

Under these conditions, the validation run was performed 14 times, each with slight changes to the primary scheduling rule. The modifications changed the order in which the activities to be scheduled were selected for insertion into the tentative schedule. This was done to optimize the performance of the model scheduler in terms of reproducing the MCS schedule. This "tuning" of the model does not optimize the scheduler against some absolute criteria; it minimizes the difference between the real system and the model. The validation test inputs and raw results are in Appendix C; these results are summarized in Table 4.3. Overall, the differences between the various ranking methods were negligible, and resulted in the minor reordering of scheduled activities.

Table 4.3. Validation Results

Test	Priority	Max	Min	Mean	Offset	No. of
No.	Scheme	Offset	Offset	Offset	Std Dev	GAs
		(minutes)	(minutes)	(minutes)	(minutes)	Matched
1	hP/hL/hI	148	0	41.1404	40.9769	23
2	hP/hL/lSV	148	0	41.1404	40.9769	23
3	hP/IL/hI	168	0	41.9825	43.0053	25
4	hP/IL/ISV	168	0	41.9825	43.0053	25
5	hP/ISV/IL	148	0	40.8947	41.0565	25
6	hP/ISV/hL	148	0	40.8947	41.0565	25
7	hP/hI	148	0	40.8246	41.0648	25
8	hP/II	148	0	40.5965	40.8355	26
9	hP/hD	148	0	40.5965	40.8355	26
10	hP/lD	148	0	40.8246	41.0648	25
11	hP/II/hSV	148	0	40.6667	40.8276	26
12	hP/II/ISV	148	0	40.5965	40.8355	26
13	hP/II/hST	148	0	41.1404	40.9769	25
14	hP/lI/lST	168	0	41.4211	42.6972	25

Priority Scheme Key

h - High i - lo

P - PRIORITY L - LEAD.TIME

1 - INTERVAL SV - SV (name)

ST - START TIME D - DURATION

The Priority Scheme column describes the rank ordering of the activity entities in the TO.BE SCHEDULED queue. The scheduling process will try to fit the activities into the schedule in this order. The initial sorting is always by high PRIORITY, to allow the rescheduling priority system to operate. It was anticipated that high LEAD.TIME would be the best starting choice for the second sorting level, as this would allow the activities closest to being tardy to be scheduled first. A third sorting level was included to allow further differentiation of the results. Fourth and greater levels had no effect on the outcome and were removed. The test proceeded by successively changing the second and third ordering criteria in an effort to converge on the simulation result schedule that best matched the MCS schedule. Tests 8 and 9 resulted in the lowest mean differential between the activities in the simulated versus MCS schedule. Additional tests were then performed to test alternative ordering arrangements.

Table 4.4. Activity Types.

SV	Block	Activity	Start Time	Duration	Interval
		Name	(Jminute)	(minutes)	(minutes)
BI-011	I	SOH	144375	10	480
BII-01	II	GBDDUMP	144220	10	720
BII-01	11	SOH	144210	10	720
BII-12	IIA	NDUTLM	144175	10	720
BII-01	H	NAV	144950	15	1440
BI-011	1	NAV	144385	5	1440
BII-01	II	NAVBUFF	144230	5	1440

The ordering factors that resulted in the best match, low INTERVAL and high DURATION, are not independent. The high dura ion activities generally have the highest inter-activity interval. The reason the results are identical between Test 8 and 9 are that both orderings minimize the interference between activities being scheduled. Table 4.4 shows the duration/time characteristics for the seven different activity types. Note that the two activities with the lowest duration (Block I NAV and Block II NAVBUFF) have 1440 minute intervals. This means that whether the low INTERVAL or high DURATION ordering mode is selected, the simulation will generally schedule the low INTERVAL activities first. The chronological separation of the successively-scheduled activities reduces contention.

A further indication of this effect is shown by the results of Test 14. When the START.TIME of the previously scheduled activity is used to further sort the activities to be scheduled, the ability of the simulation to match the MCS schedule is reduced. Table 4.5 on Page 4-8 shows this effect in detail, by showing the offsets for each activity in a histogram. The two tests shown, Test 8 and Test 14, are the best and worst, respectively, in terms of mean activity start time differential. As stated above, the differences between the test have only a minor effect on the cumulative statistics. The high PRIORITY/low INTERVAL ordering scheme was used for the experimental runs.

Table 4.5. Validation Test Comparison (Test 8 vs. Test 14).

	Tea	st 8	Tes	t 14
Start Time	No. of	Percent of	No. of	Percent of
Offset (min)	Activities	Activities	Activities	Activities
0-10	20	35.09	18	31.58
10-20	7	12.28	9	15.79
20-30	3	5.26	5	8.77
30-40	3	5.26	_ 3	5.26
40-50	4	7.02	3	5.26
50-60	2	3.51	1	1.75
60-70	5	8.77	3	5.26
70-80	0	0	2	3.51
80-90	3	5.26	2	3.51
90-100	4	7.02	4	7.02
100-110	2	3.51	2	3.51
110-120	2	3.51	3	5.26
120-130	0	0	0	0
130-140	0	0	0	_ 0
140-150	2	3.51	1	1.75
150-160	0	0	0	_ 0
160-170	0	0	1	1.75
170+	0	0	0	0

4.3.2 Validation results. The overall assessment of the validation results is that the model does not reproduce the MCS Master Contact Schedule with sufficient fidelity to use the model for quantitative comparisons.

The model does bass the face validation test, as it does operate in a manner comparable to the MCS operations center. The schedule it produced during validation testing does meet the requirements described in the Navstar Satellite Support Requirements document. However, the quantitative results seen when this schedule is compared numerically to the Master Contact Schedule cannot support the use of the model results for quantitative predictions of MCS behavior. This judgement is based on the fact that

- the average difference between 57 activities was over 40 minutes,
- 18 of 57 activities deviated by more than 60 minutes,

- the validation test looked at only the first scheduled activities, which would yield the best results the system could produce,
- the large number of deviations this early in the scheduling process would compound as the process continued.

The opposite side of this coin is that the model does perform as the MCS well enough to make general comparisons between the model's and the MCS's response. The successful face validation indicates the pieces of the MCS are all in the model. Even though it cannot be assumed the resulting numbers have meaning, the operation of the model should result in data thet will match and predict trends in MCS resource utilization. On this basis, it was decided to proceed with experimentation. The experimental results would be examined for indications that they conform to the expected performance of the MCS (as well as is known). If sufficient parallels can be found between known or suspected MCS behaviors and model performance, it may be possible to make conclusions about general MCS trends based on the model's output.

4.4 Experimentation.

The MCS model created for this research is not subtle. It is the embodiment of an MCS high-level block diagram. As such, there are not a huge number of experimental parameters that can be modified to perturb the model's operation. The model is also currently dedicated to research involving the two major environmental factors that can and will create problems for the MCS - the growing number of satellites in the GPS constellation and the reliance on a limited network of ground stations to keep those satellites operational. These factors dictate the scope and direction of experimentation.

As described in Section 4.3.2, the output from the model is presumed to be not of sufficient quality to make one-to-one quantitative predictions. The focus of experimentation is to determine if the model can at least be used to make general conclusions about MCS behavior.

- 4.4.1 Experiment Plan. The major factors that can be varied to exercise the MCS model are:
 - 1. the number of active GPS satellites in the constellation,
 - 2. the number of GPS Ground Antennas available for use, and
 - 3. the number of simultaneous MCS-to-satellite contacts possible.

At the validation date of 10 April 1992, the were 15 fully operational GPS satellites and one operational but undergoing testing (BII-12). The 15 satellites broke down into four aging Block I (R&D) SVs, nine Block II SVs, and two Block IIA SVs. The operational constellation will include 21 perational satellites and three operational spares, 24 total. The MCS is programmed to handle as many 30 satellites simultaneously, in a mixture of Block I, II, and IIA vehicles. For the purposes of this experiment, the number of satellites was varied between the current number (16) and the operational number (24), keeping the four Block I's and adding eight Block IIA's. If the four currently operational Block I satellites remain healthy and the Air Force Space Command's goal of six launches per year is met, this 24 satellite configuration will occur sometime in 1993.

There are currently no plans to augment the number of GPS Ground Antennas now operational. However, it often happens that one or more GAs are unavailable for satellite contacts for extended periods. In the event of a natural disaster or catastrophic failure, any GA could be out of service for days or even weeks. Another circumstance is that the CAPE GA is reserved for prelaunch compatibility test for days prior to a GPS satellite launch. So the number of operational GAs is an experiment parameter. Each GA was removed from the system in turn, then removed in pairs. The exception to this rule is the PIKE GA. It is a unique case in both the operation of the model and in the experiment matrix. As a non-dedicated GPS resource, it is normally used far less than the remaining GAs. This is accounted for in the model by making PIKE last on the list for selection when the scheduler looks for open GAs. PIKE also covers nearly the same coverage area as CAPE, as they are only 10 degrees apart in latitude and 25 degrees separated in longitude.

Therefore, assuming PIKE outages have minimal effect on MCS operations, it was excluded from paired outage table. This reduced the number of this type of experiment from ten to six.

While the MCS is designed to handle up—four satellite contacts at one time, the current operational practice is to have a maximum of only three contacts. This is due to crew manning (each crew has three Satellite Support Operator personnel trained and available to handle routine contacts) and MCS computer system loading. Due to the large number of experiments planned with just the number of SVs and GAs being varied, it was decided to fix the maximum number of contacts to three and not vary this parameter experimentally.

More significantly, the stochastic nature of the operations segment of the simulation was not included in this set of experiments. This is due in part to insufficient time to perform experiments with this facet of the simulation. However, the reason these tests were of lesser importance was that the data collected at the MCS and discussion with the MCS crew indicated that variable activity durations were not a significant MCS operational factor. The standard activity durations listed in the NSSR were so conservative that seldom if ever did the activity durations as performed exceed the scheduled times allowed. In the rare instances where this occurred, the variation was due to system anomalies that disrupted more than just the activity being performed. The wide variety of anomalies that create this situation and their diverse effect on operations are not practical to include in this simple model.

The experimental suite required results from 12 different GA outage configurations and 9 SV quantities, for total of 108 distinct runs of the MCS model. To keep the simulation run time reasonable, the simulation length was limited to 48 hours. The initialization configuration used during validation was the experimental baseline. This configuration corresponds to the state of the OCS as of 0000Z 10 April 1992. As each SV was added to the configuration, its activities on 9 April were included in the activity initialization file, as was it's GA visibility data. In the case of SVs not yet launched, an fictional activity history consistent with the "real" activities was

created. The input conditions during testing are described more fully and the data itself provided in Appendix B. The results of the experiments are summarized in the next section, in both tabular and graphic formats.

4.4.2 Experiment Analysis. The data provided by the experiment runs fall into two general categories, based on the measurement criteria selected to characterize the MCS model performance. The first category is data that describes the activity scheduling process and the number of missed/failed satellite support requirements. The second category of data relates to the utilization of OCS resources, primarily the GPS Ground Antennas.

4.4.2.1 Activity Scheduling Analysis. The most significant indicator of the effect of constellation size and GA outages on scheduling ability is the number of activities that could not be scheduled. This number, along with the total number of activities scheduled for that experiment, is shown after the slash in Table 4.6. No slash indicates there were no activities that could not be scheduled.

Table 4.6. Number of Activities Scheduled/Failed

					Ground Ar	ten a(s)	Unavail	able				
SVa	NONE	ASCN	CAPE	DIEG	KWAJ	FIKE	A/K	A/C	A/D	K/D	C/D	C/K
16	184	184	164	184	184	184	184	185	187/1	181/3	184	184
17	196	196	196	196	196	196	196	198	199/1	193/3	196	196
18	208	20A	208	208	208	208	208	210	210/1	205/3	208	208
19	220	220	220	221	226	220	220	222	226/1	216/3	222	220
20	232	232	232	233	232	232	232	234	238/1	231/3	234	232
21	244	245	244	245	246	244	246	247	250/3	245/3	246	246
22	256	257	256	257	258	256	258	259	262/3	257/3	258	259
23	268	269	268	269	270	268	270	271	274/3	269/3	270	271
24	280	281	280	281	283	240	284	283	286/4	284/3	282	284

As seen in the NONE column, under optimal conditions there were 12 additional satellite activities added to the schedule with each additional satellite added to the constellation. As GA resources were removed from availability, the number of activities paradoxically increased, especially when the system had to cope with a large constellation. Follow along the bottom row of Table 4.6 to see this effect. The reason this occurs is the scheduling algorithm. The system schedules by

looking first for the latest time an activity can be performed. If that time is unavailable, the system "slides" the activity along the schedule looking for the first earlier time resources are available for the activity.

For example, an activity is normally scheduled four hours apart. The system determines the last time this activity was scheduled and starts looking four hours hence to schedule the next occurrence. If that time is unavailable, it may have to settle for a time only three hours after the last occurrence. The repetition rate of this activity has thus increased, because the effective time interval between activities of this type as scheduled has decreased. If this happens four times over the 48 hour simulation period, the last scheduled activity of this type will occur four hours earlier than in a schedule without conflict. Thus there will be "room" for the scheduler to add another activity of this type to the schedule, increasing the total activity count. This effect can be put to use as an indicator of scheduling "stress", or how much conflict occurs in the scheduling process.

Due to the periodic nature of the scheduled activities, the scheduler will not experience continuous conflict. The various activities will occasionally synchronize and for a short period the system will have trouble finding resources to satisfy the scheduling requirements. This situation is shown clearly in Figure 4.1 on Page 4-14, which displays the number of GAs utilized over simulation time. The periodic nature of scheduling "crunches" is clear, with the most serious problems coming at about 1400 and 2800 minutes into the simulation. Note that increasing the constellation size from 16 to 24 satellites makes the problem worse; most of the additional activities (dotted lines) are clustered around previous clusters.

Using the number of activities scheduled as an indicator, note that both CAPE and PIKE outages cause no increase in stress. This is not unexpected, as these GAs cover approximately the same segment of the satellite orbits and can "cover" for each other (as seen in Figure 2.4). The system first experiences noticeable stress when the Diego Garcia GA is out and there are 21 satellites in the constellation. This is expected, also, as DIEG has the least overlap of all the GAs.

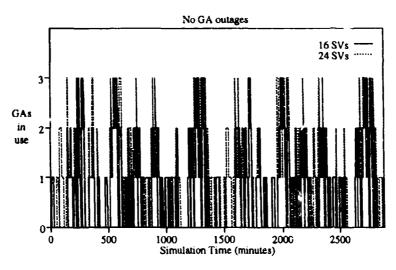


Figure 4.1. Number of GAs In Use vs. Simulation Time.

The system has fewer options with DIEG down. The fact is reinforced by examining the dual outage scenarios in Table 4.6. The only scenarios with activities failing to be scheduled involve DIEG and either of the adjacent GAs. (The number of activities scheduled in these columns naturally does not include the failed supports.) In these dual-outage scenarios, scheduling conflicts are greater and begin earlier than with the single outage scenarios.

The next three tables describe other indicators of scheduling conflicts. Table 4.7 shows the summed priority of all the scheduled activities. As described in Section 3.4.3.3 (Page 3-19), priority is an activity entity attribute used to improve the activity's probability of being scheduled in a conflict situation. The higher the total priority, the more conflicts are occurring. These values give a better indication of the level of scheduling difficulty for each scenario than just the activities scheduled. Of interest in the table is the fact that every run had to use the priority system at least once. A review of the raw simulation results showed that this occurred only 20 minutes into the scheduling period in the first scheduling iteration. Two BII-12 activities were trying to be scheduled for this time while this SV had visibility only at DIEG. The activity with the low interval (SOH) had default priority, but the other activity (NDUTLM) had to start between 0000 and 0020 hours

or exceed its interval requirement. The first iteration bumped the NDUTLM activity ahead of the SOH and both were scheduled during the next scheduling attempt.

Table 4.7. Total Activity Priority.

		Ground Antenna(s) Unavailable										
SVS	NONE	ASCN	CAPE	DIEG	KWAJ	PIKE	A/K	A/C	A/D	$\overline{K/D}$	C/D	C/K
16	1	1	1	9	1	1	1	14	19	13	9	1
17	1	1	1	9	1	1	1	14	19	13	9	1
18	1	1	1	9	1	1	1	14	19	19	9	1
19	1	1	1	9	1	1	1	14	16	19	9	1
20	1	1	1	9	1	1	1	14	16	19	9	1
21	1	11	1	9	1	1	11	32	25	28	9	1
22	1	11	1	9	1	1	11	32	25	28	9	1
23	1	11	1	9	1	1	11	32	25	28	9	1
24	1	11	1	9	3	1	16	32	25	33	9	3

Another interesting window into the inner workings of the scheduling algorithm is the decrease in total priority in the ASCN/DIEG outage scenario as constellation size increases from 18 to 19. Note on Table 4.6 that at this point there is an increase of 16 activities scheduled, the largest change of any for this statistic. What has happened is that a time shift occurred for one type of activity early in the schedule-building process that has allowed subsequent activities to be scheduled with less difficulty than before. The small increase in total interval offset (see below) may indicate a scheduling tradeoff has occurred; one type of activity being pushed very tardy while others are allowed to return closer to their latest "legal" start time.

As seen first in Table 4.6, Table 4.7 shows that the scheduler has the hardest time when DIEG is out, alone or with another GA. When two adjacent GAs are out, the difficulty peaks. While PIKE can pick up the contact load when CAPE is out along with DIEG or KWAJ, PIKE is not much help when ASCN and CAPE are out together. This is because ASCN and CAPE are adjacent and their outage leaves too big a hole in the coverage. Note that the priority level of a CAPE/DIEG or CAPE/KWAJ outage stays fairly level. PIKE utilization should be increasing rapidly during this scenario as the constellation size grows. The dual coverage effect between PIKE

and CAPE is also seen in that the CAPE/DIEG and CAPE/KWAJ total priorities are identical to DIEG and KWAJ alone.

Table 4.8 shows the response of the next level of activity conflict, the interval offset. This attribute is used to extend the scheduling window into the "tardy" time period, in an effort to schedule the activity after the priority system has failed. The values indicate the total number of minutes the system had to add to scheduled activities to complete the schedule. Failed activities are not counted. Table 4.9 shows the maximum interval offset the system had to apply. The synergistic effect of the ASCN/DIEG and KWAJ/DIEG outages on scheduling difficulty are also seen in these tables.

Table 4.8. Total Activity Interval Offset

		Ground Antenna(s) Unavailable										
SV8	NONE	ASCN	CAPE	DIEG	KWAJ	PIKE	A/K	A/C	A/D	K/D	C/D	C/K
16	0	0	0	420	0	0	0	130	498	980	420	0
17	0	0	0	4.20	0	0	0	130	530	980	420	0
18	0	0	0	4.20	υ	0	0	130	523	1120	420	0
19	0	0	0	-20	ŋ	0	0	130	540	1180	420	0
20	0	0	0	420	0	0	0	1.30	540	1180	420	0
21	0	140	0	420	0	0	140	270	930	1390	420	0
22	0	140	0	420	0	0	140	270	930	1390	420	0
23	0	140	0	420	0	0	140	270	930	1390	420	0
24	U	140	0	420	0	0	140	270	930	1410	420	0

Table 4.9. Maximum Activity Interval Offset

	Ground Antenna(s) Unavailable											
SV ₈	System	ASCN	CAPE	DIEG	KWAJ	PIKE	A/K	A/C	A/D	K/D	C/D	C/K
16	U		0	300	0	0	0	40	210	510	200	0
17	0	0	0	200	0	0	0	40	210	510	200	r
18	0	0	0	200	0	0	0	40	210	510	200	0
19	0	0	0	200	0	0	0	40	210	510	200	0
20	O	0	0	200	0	0	0	40	210	510	200	0
21	0	70	0	200	0	0	70	70	390	510	200	0
22) ວ	70	0	200	U	U	70	7u	390	610	200	0
23	0	70	0	200	0	0	70	70	390	510	200	0
24	0	70	0	200	0	0	70	70	390	510	200	0

4.4.2.2 Activity Scheduling Results. The above data confirms the basic assertions made earlier regarding the criticality of the Diego Garcia GA. Using these results, system managers

could rank the GPS GAs by least desirable to be unavailable. The order would be (in order of most preferable to be unavailable)

- 1. CAPE or PIKE (tie)
- 2. KWAJ or CAPE/KWAJ (tie)
- 3. ASCN
- 4. ASCN/KWAJ
- 5. DIEG or CAPE/DIEG
- 6. ASCN/CAPE
- 7. ASCN/DIEG
- 8. KWAJ/DIEG

From a practical standpoint it is never preferable to have two GAs down at the same time, but the simulation indicates it will be easier to build a schedule (and of course perform one) if both ASCN/KWAJ, ASCN/CAPE, or CAPE/KWAJ were down than if DIEG alone were unavailable. This is with PIKE fully available but used only if no GPS-dedicated resources are available.

These data build a case for the increased confidence in the validity of the model's operation. The satellite contact scheduling process appears to react in tune with the constraints placed on the system by the geographic location of the ground stations. In addition, the scheduling algorithm reacts logically to the increased stress placed on the system by the GA outages and constellation build-up.

4.4.2.3 Ground Antenna Utilization Analysis. The overall system usage of the GAs for each experimental scenario is summarized in Table 4.10; the standard deviations of total utilization are contained in Table 4.11. The means represent the mean number of GAs in use for any given minute during the simulation, and vary from a low of 0.7000 to a high of 1.1302. These numbers do not count the GAs unavailable during that experiment. The means relate directly to the number of activities scheduled during that run, because there are a fixed number of GA-minutes available during the simulation run and adding activities to the schedule use these up. This is the reason

for the regular increase in mean utilization as the simulated constellation grows. The standard deviations are simply a confidence builder; their uniformity is reassurance the model is not off the rails in some fashion that is hidden by the other indicators.

Table 4.10. Mean Number of GAs Utilized for SV Activities (per simulation minute).

		Ground Antenna(s) Unavailable											
SVS	NONE	ASCN	CAPE	DIEG	KWAJ	PIKE	A/K	A/C	A/D	K/D	C/D	C/K	
16	0.7132	0.7132	0.7132	0.7132	0.7132	0.7132	0.7132	0.7170	0.7247	0.7000	0 7132	0.7132	
] 17	0.7625	0.7625	0.7625	C.7625	0.7625	0.7625	0 7625	0.7701	0.7740	0.7493	0.7625	0.7625	
18	0.8118	0.8118	0.8118	0.8118	0.8118	0 8118	0.8118	0 8194	0.8309	0.7986	0 8118	0 8118	
19	0.8611	0.8611	0.8611	0.8649	0.8611	0.8611	0.8611	8896.0	0.8840	0.8556	0.8686	0.8611	
30	0.9104	0.9104	0.9104	0.9142	0.9104	0.9104	0.9104	0.9181	0.9333	0.9049	0.9181	0.9104	
21	0.9597	0.9635	0.9597	0.9635	0.9674	0.9597	0.9674	0 9712	0.9826	0.9618	0 9674	0 9674	
22	1.0090	1.0128	1.0090	1.0128	1.0167	1.0090	1.0167	1.0205	1 0319	1.0111	1 0167	1 0205	
23	1.0583	1.0622	1.0583	1.0622	1.0660	1.0583	1.0660	1 0698	1 0812	1 0604	1.0660	1 0698	
24	1.1076	1.1115	1.1076	1 1115	1 1191	1 1076	1 1229	1.1191	1.1302	1.1212	1.1153	1.1229	

Table 4.11. Standard Deviation of Number of GAs Utilized for SV Activities.

		Ground Antenna(a) Unavailable											
SVS	NONE	ASCN	CAPE	DIEG	KWAJ	PIKE	A/K	A/C	A/D	K/D	C/D	C/K	
16	0.8431	0.8039	0.7904	0.7640	0.8039	0.7877	0.7829	0.7150	0.7380	0.7685	0.7442	0.7484	
17	0.4473	0.8083	0.7962	0.7691	0.8130	0.7905	0.7922	0.7303	0.7665	0.7797	0.7494	0.7378	
18	0.8413	0.8041	0.7897	0 7660	0.7997	0.7840	0 7835	0.7355	0.7854	0 7913	0.7565	0.7265	
19	0.8401	0.8376	0.7858	0.7743	0.7985	0.7827	0.8003	0.7556	0.8146	0 8270	0.7946	0.7223	
20	0.8590	0.8730	0.8171	C 7812	0.8377	0.8141	0.8285	0.7942	0.8386	0 8660	0.8102	0.7442	
]] 21	0.8444	0.8790	0.8018	0.7899	0.8167	0.7987	0.7978	0.8055	0.8484	0.8539	0.8184	0.7357	
22	0.8831	0.8840	0.8469	0.8276	0.8277	0.8469	0.8056	0.8241	0.8868	0.8591	0 8332	0.7425	
23	0.9231	0.8990	0.8886	0.8428	0.8611	0.8812	0.8361	0 8319	0.8762	0 8830	0 8386	0.7305	
24	0.9285	0.9074	0.8899	0.8743	0.8917	0.9054	0.8512	0.8572	0 9102	0.9038	0.8502	0.7520	

To associate reality to experimental mean utilizations, a similar calculation was made from the actual Master Contact Schedules for 15, 16, and 17 April 1992. These days were chosen for the lack of apparent OCS anomalies during this period. The GA contact durations for the satellites and satellite activities were totaled, with some approximations made to account for the presence of non-simulated activities mixed-up in the total contact duration. With 141 total contacts counted (each with one, two, or three activities), the total GA use came to 3470 GA-minutes. Since for the three-day period analyzed there were 1440 minutes/day * 3 days * 5 GAs, there were 21600 GA-minutes available for use. Thus the calculated utilization rate for this three day period is 0.1606. For the 16-satellite simulation with no outages, there was a mean of 0.7132 GAs used per simulation minute. But with five GAs operational, each simulation minute = five GA-minutes,

so the simulation GA utilization rate is 0.7132 per 5 GA-minutes, or 0.1426. These numbers are meant for general comparison only, as neither calculation method is particularly rigorous, but the "ballpark" figure indicates the simulation is operating as it was designed.

Page 4-20 through Page 4-31 show the response of individual GAs to the changes in both GA outages and constellation size. A graph is shown with each table to assist the reader in discerning trends and unusual behaviors. Notes associated with each table and graph highlight the salient points of that set and inter-relationships between the data shown and other experimental results.

Table 4.12. Individual GA Utilization with No GA Outages.

1		Percentag	ge of total	simulati	on time	
	No GAs	ASCN	CAPE	DIEG	KWAJ	PIKE
SVs	in use	in use	in use	in use	in use	in use
16	50.59	16.22	27.74	13.51	11.01	2.85
17	47.19	18.16	28.47	13.51	12.71	3.40
18	42.18	20.10	28.47	13.51	15.69	3.40
19	39.24	21.63	28.68	16.70	15.69	3.40
20	36.53	23.16	28.68	17.12	18.68	3.40
21	32.50	26.15	28.68	17.12	20.63	3.40
22	31.56	27.99	29.06	18.26	22 .19	3.40
23	31.35	27.99	30.52	20.21	22.95	4.17
24	29.62	28.99	31.46	20.21	25.94	4.17

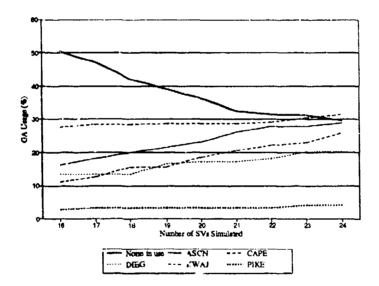


Figure 4.2. GA Utilization - No GA out.

This is the individual GA utilization rates with no outages. The heavy black line shows the percentage of total simulation time there were no GAs in use. This gives a rough idea of the total GA slack in the system. CAPE GAs consistently high utilization is due to the low use of PIKE; activities in the coverage area of both GAs will go to CAPE first. PIKE is last in the GA selection priority to simulate its AFSCN schedule requirements. The utilization of the remaining GAs increases steadily as the constellation grows. It is possible that the satellite activities naturally "chuster" (see Figure 4.1) while in CAPE's coverage area. This would explain CAPE's consistently high utilization.

Table 4.13. Individual GA Utilization with ASCN Out.

	1	Percenta	ge of tota	simulation	on time			Change fo	om "No	Outage"	
	No GAs	ASCN	CAPE	DIEG	KWAJ	PIKE	No GAs	CAPE	DIEG	KWAJ	PIKE
SVs	in use	in use	in use	in us	in use	in use	Δ%	Δ%	Δ%	۵%	Δ%
16	48 47	0.00	31.60	22.22	12.74	4 76	-212	3.86	8.71	1.73	1.91
17	45.07	0.00	32.33	24.17	14.44	5.31	-2.12	3.86	10.66	1.73	1.91
18	40.73	0.00	32.33	26.11	17.43	5.31	-1.45	3.86	12.60	1.74	1.91
19	39.27	0.00	34.27	29.10	17.43	5 31	0.03	5.59	12.40	1.74	1.91
20	37.50	0.00	34.65	29.10	20.45	6.84	0.97	5.97	1198	1.77	3.44
21	35.03	0.00	35.42	31.70	22.40	6.84	2.53	6.74	14.58	1.77	3.44
22	32.81	0.00	37.26	32.85	23.96	7.22	1.25	8.20	14.59	1.77	3.82
23	31.28	0.00	38.72	34.79	25.10	7.60	-0.07	8.20	14.58	2.15	3.43
24	28.99	0.00	39.10	36.74	28.09	7.22	-0.63	7.64	16.53	2.15	3.05

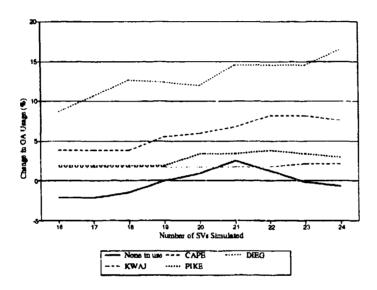


Figure 4.3. GA Utilization Change - ASCN out.

With ASCN GA unavailable, the adjacent GAs (DIEG to the east and CAPE to the west) pick up the slack. CAPE had less slack to start, so most of the additional load falls to DIEG. The extensive overlap of CAPE and DIEG into ASCN coverage means the slack is picked up "clean", with little additional load slipping to KWAJ and PIKE. The "None in Use" line indicates that with 20, 21, and 22 SVs simulated, the system is actually more efficient than the baseline in terms of more time with no GAs in use.

Table 4.14. Individual GA Utilization with CAPE Out.

	1	Percenta	ge of tota	simulation	on time			Change fr	om "No	Outage"	
1 1	No GAs	ASCN	CAPE	DIEG	KWAJ	PIKE	No GAs	ASCN	DIEG	KWAJ	PIKE
SVs	in use	in use	in use	in us	in use	in use	Δ%	Δ%	Δ%	Δ%	Δ%
16	46.98	16.98	0.00	14.27	1.790	20 17	-361	0.76	0 76	6 69	1732
17	43.55	18 92	0.00	14.27	21.94	21.11	-3.54	0.76	0.76	9.23	1771
18	39.62	30.87	0.00	14.27	24.93	21.11	-2.56	0.77	0.76	9.24	17.71
19	35.82	22.€0	0.00	17.47	24 93	21 11	-3 42	0.97	0.77	9.24	1771
20	34.72	24 13	0.00	17.88	27.92	21.11	-181	0 97	0.76	9.24	17.71
21	30.69	27.12	0.00	17.88	29.86	21 11	-1.81	0.97	0 76	9.23	17.71
22	30.07	29.34	0.00	18.65	32.19	20.73	-1.49	1 35	0.39	10 00	17.33
23	29.86	29.34	0.00	20.59	34.20	21.70	-1.49	1.35	0.38	11.25	17.53
24	27.78	30.90	0.00	20.42	37.19	22.26	-184	191	0.21	11.25	18.09

Figure 4.4. GA Utilization Change - CAPE out.

In this scenario, PIKE picks up the majority of the load created by a CAPE outage. CAPE and PIKE coverage overlaps to such an extent and CAPE leaves such a large hole that the system overcomes its inherent reluctance to schedule activities at PIKE. As they are to the east, ASCN and DIEG are on the "back" of PIKE/CAPE coverage and the loading problem is solved before their visibility periods start for the west-to-east travelling satellites.

Table 4.15. Individual GA Utilization with DIEG Out.

		Percenta	ge of total		on time		Change from "No Outage"						
[No GAs	ASCN	CAPE	DIEG	KWAJ	PIKE	No GAs	ASCN	CAPE	KWAJ	PIKE		
SV ₈	in use	in use	in use	irı us	in use	in use	Δ%	Δ%	Δ%	Δ%	Δ%		
16	46.11	25.24	29.27	0.00	12 43	4 38	-4.48	9 02	1.53	1.42	1.53		
17	42.74	27.19	30.00	0.00	14.13	4.93	-4.45	9 03	1.53	1.42	1.53		
18	38.72	29.13	30.21	0.00	17.12	4.72	-3.46	9.03	1.74	1.43	1 32		
19	^5.56	31.25	31.18	0.00	19.34	4.72	-3.68	9 62	2 50	3.65	1.3		
20	32.50	32.78	31.18	0.00	22 74	4.72	-4.03	9 62	2.50	4.06	1.32		
21	30.35	35 76	31.18	0.00	24 69	4.72	-4.03	9.62	2.50	4.06	1.32		
22	28.75	38 37	31.56	0.00	26 63	4.72	-2.81	10.38	2 50	4.44	1.32		
23	26.74	3∌.51	33.78	0 00	27.01	5 · 0	-4.61	11.52	3.26	4.06	1.73		
24	26.70	40 52	34.34	0.00	30.38	5.90	-2.92	11.53	2 88	4 44	1 73		

Nome in use — ASCN —— CAPE

Nome in use — ASCN —— CAPE

NWAJ —— PIKE

Figure 4.5. GA Utilization - DIEG out.

With DIEG out, the GA that overlaps its coverage to the west, ASCN, picks up the slack. The response is similar to the ASCN-out scenario, with ASCN and DIEG taking each others' role.

Table 4.16. Individual GA Utilization with KWAJ Out.

		Percenta	ge of tota	simulation	on time			Change fr	om "No C	utage"	
1 1	No GA	ASCN	CAPE	DIEG	KWAJ	PIKE	No GAs	ASCN	CAPE	DIEG	PIKE
SVs	in use	in use	in use	in us	in use	in use	Δ%	Δ%	Δ%	Δ%	Δ%
16	48.33	15.28	29.62	17.33	0.00	9.10	-2.26	-0.94	1.88	3.82	6.25
17	45.31	17.22	30.35	17.33	0.00	11.35	-1.88	-0.94	1.88	3.82	7.95
18	40.73	19.17	30.35	17.33	0.00	14.34	-1.45	-0.93	1.88	3.82	10.94
19	37.15	20.69	30.56	20.52	0.00	14.34	-2.09	-0.94	1.88	3.62	10.94
20	35.97	22.01	31.88	20.38	0.00	16.77	-0.56	-1.15	3.20	3.26	13.37
21	31.04	24.24	33.40	22.33	0.00	16.77	-1.46	-1.91	4.72	5.21	13.37
22	28.85	27.22	33.40	22.33	0.00	18.72	-2.71	-0.77	4.34	4.07	15.32
23	27.88	26.88	33.75	24.83	0.00	21.15	-3.47	-1.11	3.23	4.62	16.98
24	27.29	27 85	33.96	24.83	0.00	25.28	-2.33	-1 14	2.50	4.62	21.11

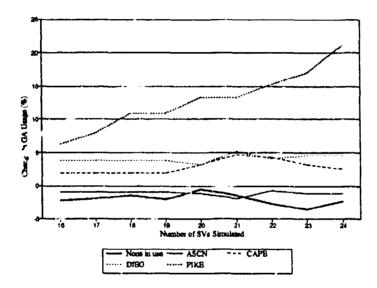


Figure 4.6. GA Utilization - KWAJ out.

Although DIEG picks up some additional use, PIKE's low utilization acts like a magnet to gather the activities normally scheduled for KWA. ASCN use decreases slightly as some of the previous KWAJ (now PIKE) activities are satisfied earlier and skip over ASCN.

Table 4.17. Individual GA Utilization with PIKE Out.

			ge of tota	aimulati	on time		Change from "No Outage"							
	No GAs	ASCN	CAPE	DIEG	KWAJ	PIKE	No GAs	ASCN	CAPE	DIEG	KWAJ			
SVs	in use	in use	in use	in us	in use	in use	Δ%	۵%	Δ%	Δ%	Δ%			
16	47.60	16.98	30.59	14.27	9.48	0.00	-2.99	0.76	2.85	0.76	-1.535			
17	44.06	18.92	32.12	14.27	10.94	0.00	-3.13	0.76	3.65	0 7G	-1.77			
18	40.03	20.87	32.12	14.27	13.92	0.00	-2.15	0.77	3.65	0.76	-1.77			
19	36.46	22.40	32.33	17.47	13.92	0.00	-2.78	0.77	3.65	0.77	-1.77			
20	35.35	23.92	32.33	17.88	16.91	0.00	-1.18	0.76	3.65	0.76	-1.77			
21	31.32	26.91	32.33	17.88	18.85	0.00	-1.18	0.76	3.65	0.76	-1.78			
22	30.94	28.75	32.71	18.65	20.80	0.00	-0.62	0.76	3.65	0.39	-1.39			
23	30.03	29.31	34.17	20.59	21.77	0.00	-1.32	1.32	3.65	0.38	-1.18			
24	29.17	30.31	35.10	20.59	24.76	0.00	-0.45	1.32	3.64	0.38	-1.18			

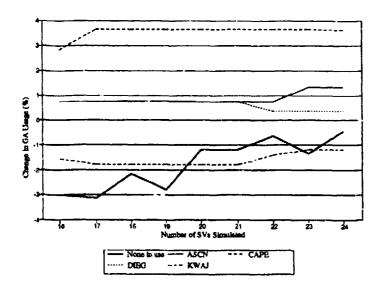


Figure 4.7. GA Utilization - PIKE out.

No GA has to work much harder to compensate for a PIKE outage. The system doesn't seriously require this GA with CAPE available (a crucial "if").

Table 4.18. Individual GA Utilization with ASCN and CAPE Out.

			ge of tota	simulation		Change from "No Outage"						
i	No GAs	ASCN	CAPE	DIEG	KWAJ	PIKE	No GAs	DIEG	KWAJ	PIKE		
SV ₃	in use	in use	in use	in us	in use	in use	4%	Δ%	Δ%	Δ%		
16	42.60	0.00	0.00	24 17	20.87	26.67	-7.99	10.66	9.86	23.82		
17	39.65	0.00	0.00	26.49	23.30	27.22	-7.54	12.98	10.59	23.82		
18	36.56	0.00	0.00	28.44	26.28	27.22	-5.62	14.93	10.59	23.82		
19	34.55	0.00	0.00	31 42	26.28	29.17	-4.69	14.72	10.59	25.77		
20	33.82	0.00	0.00	31.25	28.92	31.63	-2 71	14.13	10.24	28.23		
21	30.63	0.00	0.00	33.47	31.42	32.22	-1.87	16.35	10.79	28.82		
22	29.31	0.00	0.00	34.24	33.37	34.44	-2.25	15.98	11.18	31.04		
23	27.71	0.00	0.00	36.18	36.35	34.44	-3.64	15.97	13.40	30 27		
24	26.49	0.00	0.00	37.74	39.31	34.86	-3.13	17.53	13.37	30.69		

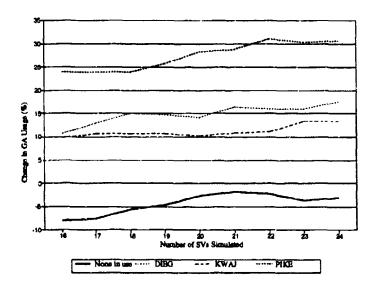


Figure 4.8. GA Utilization - ASCN and CAPE out.

With both ASCN and CAPE out, PIKE is brought to high utilization, together with major increases in the remaining GAs.

Table 4.19. Individual GA Utilization with ASCN and DIEG Out.

		Percenta	ge of tota	simulati	on time		Cha	nge from	"No Outag	e"
SVs	No GAs in use	ASCN in use	CAPE in use	DIEG in us	KWAJ in use	PIKE in use	No GAs	CAPE	KWAJ A%	PIKE
- 16	43.51	0.00	40.59	0.00	20.28	11.60	-7.08	12.65	9.27	8.75
17	41.81	0.00	43.26	0.00	21.98	12.15	-5.38	14.79	9.27	8.75
18	38.51	0.00	44.83	0.00	25.35	12.92	-3.67	16.36	9.66	9.52
19	36.77	0.00	45.38	0.00	27.19	15.83	-2.47	16.7C	11.50	12.43
20	35.03	0.00	46.91	0.00	30.59	15.83	1.50	18.23	11.91	12.43
21	33.13	0.00	47.67	0.00	34.38	16.22	0.63	18.99	13.75	12.82
22	32.01	0.00	49.90	0.00	35.94	17.36	0.45	20.84	13.75	13.96
23	28.85	0.00	52.71	0.00	35.94	19.48	-2.50	22.19	12.99	15.31
24	28.09	0.00	52.01	0.00	39.83	21.18	-1.53	20.55	13.89	17.01

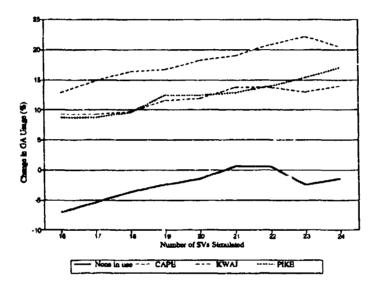


Figure 4.9. GA Utilization Change - ASCN and DIEG out.

As Table 4.6 shows, with this pair unavailable the system experiences major scheduling problems. CAPE exceeds 50% utilization trying to compensate for the outages. This is the first scenario where any GA exceeds even 40%. PIKE coverage overlaps ASCN slightly and DIEG not at all, so its low utilization cannot save the day.

Table 4.20. Individual GA Utilization with ASCN and KWAJ Out.

		Percenta	ge of total		Char	ge from '	No Outa	<u>ξe"</u>		
SV.	No GAs	ASCN in use	CAPE in use	DIEG in us	KWAJ in use	PIKE in use	No GAs	CAPE	DIEG	PIKE
16	46.35	0.00	34.24	25.49	0.00	11.60	-4.24	6.50	11 98	8.75
17	· 3.33	0.00	34.97	27.43	0.00	13.85	-3.86	6.50	13.92	10.45
18	38.92	0.00	34.97	29.37	0.00	16.84	-3.26	6.50	15.86	13.44
19	36.81	0.00	36.53	32.36	0.00	17.22	-2.43	7.65	15 66	13.82
20	35.49	0.00	36.32	32.78	0.00	21.94	-1.04	7.64	15.66	18.54
21	29.79	0.00	37.85	36.94	0.00	21.94	-2.71	9.17	19.82	18.54
22	27.43	0.00	40.07	36.94	0.00	24.65	-4.13	11.01	18.68	21.25
23	26.74	0.00	41.53	38.69	0.00	26.18	-4.61	11.01	18.68	22.01
24	24.90	0.00	42.29	39.69	0.00	30.31	-4.72	10.83	19.48	26.14

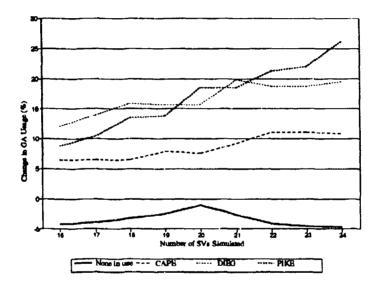


Figure 4.10. GA Utilization Change - ASCN and KWAJ out.

PIKE does better at covering a KWAJ outage than a LIEG outage, so the system can this scenario with more grace. CAPE shoulders ASCN's load and DIEG fills in the gaps on both sides.

Table 4.21. Individual GA Utilization with CAPE and DIEG Out.

		Percenta	ge of tota	simulati	on time		Change from "No Outage"					
	No GAs	ASCN	CAPE	DIEG	KWAJ	PIKE	No GAs	ASCN	KWAJ	PIKE		
SVe	in use	in use	in use	in us	in usc	in use	Δ%	Δ%	Δ%	Δ%		
16	44.79	26.98	0.00	0.00	27.40	16.94	-5.80	10.76	16.39	14.09		
17	41.42	28.92	0.00	0.00	29.06	18.26	-5.77	10.76	16.35	14.86		
18	37.78	30.87	0.00	0.00	31.49	18.82	-4.40	10.77	15.80	15.42		
19	35.59	35.03	0.00	0.00	33.02	18.82	-3.65	13.40	17.33	15.42		
20	33.62	36.56	0.00	0.00	36.42	18.82	-2.71	13.40	17.74	15.42		
21	31.67	41.56	0.00	0.00	36.37	18.82	-0.83	15.41	17.74	15.42		
22	29.51	42.53	0.00	0.00	40.31	18.82	-2.05	14.54	18.12	15.42		
23	26.81	44.27	0.00	0.00	41.39	20.94	-4.54	16.28	18.44	16.77		
24	25.42	45.66	0.00	0.00	44.55	21.32	-4.20	16.67	18.61	17.15		

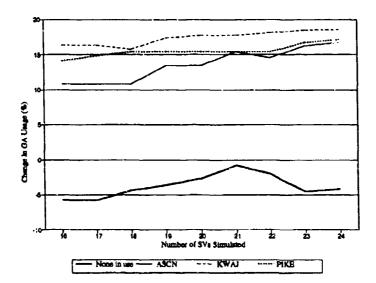


Figure 4.11. GA Utilization - CAPE and DIEG out.

The CAPE outage "forces" the reluctant PIKE to take a fair share of the load, which reduces the additional load on ASCN and KWAJ to reasonable levels.

Table 4.22. Individual GA Utilization with CAPE and KWAJ Out.

		Percenta	re of tota	simulation	on time		Change from "No Outage"						
SVa	No GAs in use	ASCN in use	CAPE in use	DIEG in us	KWAJ in use	PIKE in use	No GAs	ASCN	DIEG	PIKE			
16	44.97	18.09	0.00	15.45	0.00	37.78	-5.62	1.87	1.94	34.93			
17	40.42	20.03	0.00	15.45	0.00	40.76	-6.77	1.87	1.94	37.36			
18	35.94	21.98	0.00	15.45	0.00	43.75	-6.24	1.88	1.94	40.35			
19	32.15	23.72	0.00	18.65	0.00	43.75	7.09	2.09	1.95	40.35			
20	30.76	25.24	0.00	19.06	0.00	46.74	-5.77	2.08	1.94	43 34			
21	26.74	27.47	0.00	21.01	0.00	48.26	-5.76	1.32	3 89	44.86			
22	24.27	30.45	0.00	22.1	0.00	49.44	-7.29	2.46	3.89	46.04			
23	21.22	30.45	0.00	24.86	0.00	51.67	-10.13	2.46	4.655	47.50			
24	20.00	32.01	0.00	24.86	0.00	55.42	-9.62	3.02	4.65	51.25			

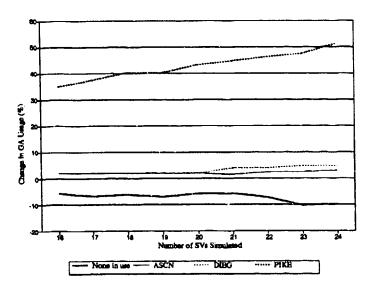


Figure 4.12. GA Utilization - CAPE and KWAJ out.

PIKE is the only alternative for scheduling activities for nearly half of the satellite's orbit. The system would have scheduling problems with this scenario if PIKE use was not close to zero in the baseline. As such, the system handles this setup better than any other dual outage, and better even than a DIEG-alone outage.

Table 4.23. Individual GA Utilization with DIEG and KWAJ Out.

		Percenta	ge of total	simulation		Cha	nge from	No Outag		
sv∎	No GAs in use	ASCN in use	CAPE in use	DIEG in us	KWAJ in use	PIKE in use	No GAs	ASCN	CAPE $\Delta\%$	PIKE
16	47.36	30.07	30.83	0.00	0.00	9.10	-3.23	13.85	3.09	6.25
17	46.50	31.63	31.94	0.00	0.00	11.35	-0.69	13.47	3.47	7.95
18	41.28	32.64	32.88	0.00	0.00	14.34	-0.90	12.54	4.41	10 94
19	38.78	35.83	34.62	0.00	0.00	15.10	-0.46	14.20	5.94	11.70
20	38.02	37.40	34.83	0.00	0.00	18.26	1.49	14.24	6.15	14.86
21	33.96	41.56	35.59	0.00	0.00	19.03	1.46	15.41	6.91	15.63
22	31.42	44.55	35.59	0.00	0.00	20.97	-0.14	16.56	6.53	17.57
23	30.21	45.90	37.61	0.00	0.00	22.33	-1.14	17.91	7.29	18.16
24	27.85	46.32	37.81	0.00	0.00	27.99	-1.77	17.33	6.35	23.82

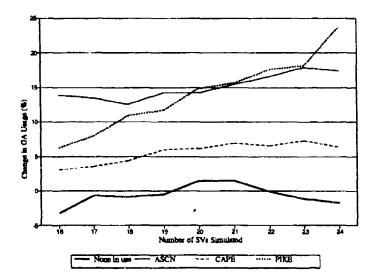


Figure 4.13. GA Utilization - DIEG and KWAJ out.

With PIKE, CAPE, and ASCN clustered into approximately 900 of longitude, this scenario opens the largest gap in the coverage. As indicated earlier, the system has the most failed supports and the highest tardiness while trying to compensate. ASCN is pushed to nearly the same utilization than with CAPE and DIEG out, while CAPE and PIKE stays under-utilized due to their westerly position relative to the outage.

Pages 4-34 through 4-37 show the percentage of total simulation time the system operates with zero, one, two, and three GAs in simultaneous use. Note that even with the simulated 24 satellite constellation, the are three GAs (and thus three Satellite Support Operators) in use only 8.68% of the time at most. Nearly 70% of the time there is less than two GAs in use. If the validity of the model could be improved to the extent these values could be assumed to reflect actual usage, they could be used by management to aid workload planning and personnel scheduling.

One interesting deviation in the otherwise consistent data is that with ASCN unavailable and the constellation size between 19 and 22, three GAs are in use more and one GA in use significantly less than in the other single-outage scenarios. It appears that with ASCN out the system the scheduler must cluster the activities slightly more than the other scenarios. This is also indicated by the higher "none-in-use" percentage during these same experiments. Figures 4.14 and 4.15 show this effect graphically. These graphs show the change in the number of GAs utilized for the ASCN and CAPE outage scenarios over the baseline no-outage condition (the CAPE outage scenario was chosen as a representative example). Both graphs compare the utilization for 21 simulated satellites. Note the higher number and wider range of GA utilization changes for the ASCN-out over the CAPE-out scenario, and the "clumping" of the ASCN-out changes at regular intervals (Figure 4.14).

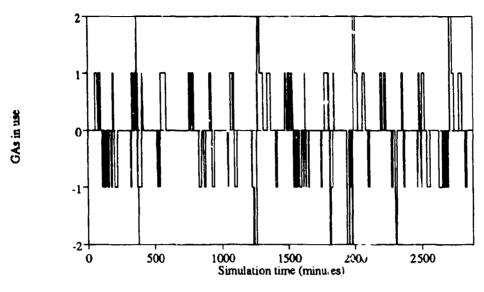


Figure 4.14. Change in GA Use due to ASCN Outage (21 SVs).

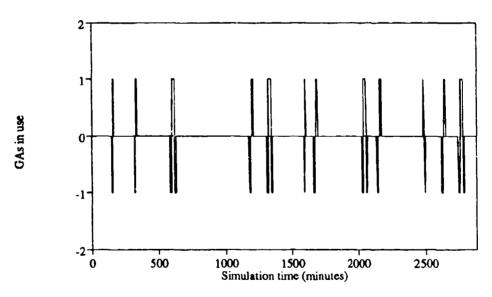


Figure 4.15. Change in GA Use due to CAPE Outage (21 SVs).

Table 4.24. Percent of Total Simulation Time with Zero GAs In Use.

					Ground A	ntenna(s)	Unavail	able				
SVS	NONE	ASCN	CAPE	DIEG	KWAJ	PIKE	A/K	A/C	A/D_	K/D_	C/D	- C/K]
16	- 54 - 75	48.47	46.98	46.11	48.33	47.60	46.35	42.60	43.51	47.36	44.79	44.97
17	19	45.07	43.65	42.74	45.31	44.06	43.33	39.65	41.81	46.50	41.42	40.42
18	42.18	40.73	39.62	38.72	40.73	40 03	38.92	36.56	38.51	41.28	37.78	35.94
19	39.24	39.27	35.82	35.56	37.15	36.46	36.81	34.55	36.77	38.78	35.59	32.15
20	36.53	37.50	34.72	32.50	35.97	35.35	35.49	33.82	35.03	38.02	33.82	30.76
21	32.50	35.03	30.69	30.35	31.04	31.32	29.79	30.63	33.13	33.96	31.67	26.74
22	31.56	32.81	30.07	28.75	28.85	30.94	27.43	29.31	32.01	31.42	29.51	24.27
23	31.35	31.28	29.86	26.74	27 88	30.03	26.74	27.71	28.85	30.21	26.81	21.22
24	29.62	28.99	27.78	26.70	27.29	29.17	24.90	26.49	28.09	27.85	25.42	20.00

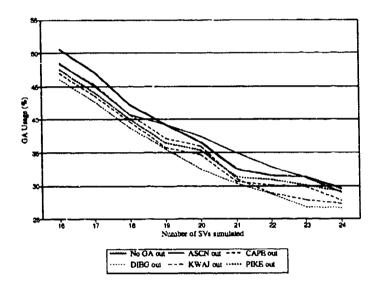


Figure 4.16. Percentage of Time No GAs in Simultaneous Use.

Table 4.25. Percent of Total Simulation Time with One GA In Use.

					Ground A	ntenna (s	Unavail	able	-			1
SVS	NONE	ASCN	CAPE	DIEG	KWAJ	PIKE	A/K	A/C	A/D	K/D	C/D	C/K
16	30.90	34.03	37.43	37.99	34.44	35.35	38.72	44.20	41.81	36 94	40.45	40.24
17	32.78	35.90	39.20	39 49	35.56	37.50	39.83	44.83	40.42	39.06	42.26	44.41
18	36.94	40.14	42.33	43.19	39 79	40.63	43.92	46.08	42.12	39.66	45.28	48.58
19	39.38	39.06	45.17	44.48	42.22	43.06	43.40	45.45	40 94	40.56	45.35	51.42
20	41.15	39.44	43.06	46.25	41 01	40.94	41.70	42 71	40.28	38.16	43.99	49.24
21	44.27	39.06	46.18	45.69	45.17	44.06	47.40	44.93	39.24	40 38	43 44	51.81
] 22	42.95	38.72	44.31	46.08	45.21	41.70	47.67	42.99	38.54	40.97	43.68	51.70
23	39.97	37.43	41.32	46.18	43.75	40.14	45.07	41.25	39.97	39.48	45.03	52.81
24	38.68	38.09	40.69	42.01	40.73	37.92	43.85	40.10	38.47	39.86	43.26	50.59

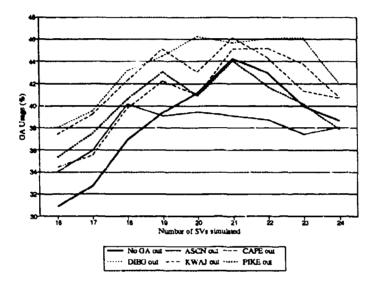


Figure 4.17. Percentage of Time One GAs in Simultaneous Use.

Table 4.26. Percent of Total Simulation Time with Two GAs In Use.

	Ground Antenna(s) Unavailable											
SVS	NONE	ASCN	CAPE	DIEG	KWAJ	PIKE	A/K	A/C	A/D	K/D	C/D_	c/K
16	15.10	15.21	12.68	14.38	14.79	15.17	12.19	12.08	13.40	14.03	13.40	13.30
17	16.63	16.74	14.41	15.94	16.70	16.56	14.10	14.38	16.35	14.86	14.97	13.68
18	16.49	16.35	15.31	16.28	17.05	17.47	14.20	16.22	17.15	16.91	14.93	13 85
19	17.43	17.95	16.04	17.88	17.99	18.40	16.67	18.58	19.41	16.98	15.66	14.58
20	17.08	17.57	18.68	18.58	19.03	21.04	19.10	21.32	21.98	19.13	18.75	18.19
21	17.99	20.42	19.58	21.22	19.79	21.94	19.10	21.15	23.89	21.18	21.39	19.44
22	18.51	22.85	20.28	20.31	21.35	22.88	20.69	24.06	23.68	22.67	22.43	21.74
23	20.17	25.07	21.94	21.22	22 26	23.78	23.06	27.40	25.38	24.38	22.85	23.75
24	23.02	25.69	24.51	24.72	24.76	25.90	25.31	28.40	25.76	24.62	25.69	26.53

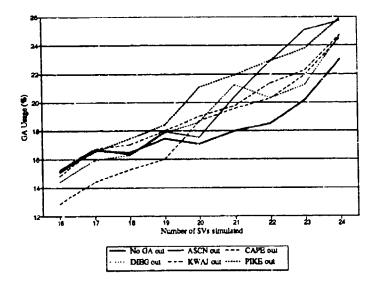


Figure 4.18. Percentage of Time Two GAs in Simultaneous Use.

Table 4.27. Percent of Total Simulation Time with Three GAs In Use.

	Ground Antenna(s) Unavailable											
SVS	NONE	ASCN	CAPE	DIEG	KWAJ	PIKE	74/K	A/C	A/D	K/D	C/D	C/K
16	3.40	2.29	2.71	1.53	2.43	1.88	2.74	1.11	1.28	1.67	1.35	1.49
17	3.40	2.29	2.74	1.53	2.43	1.88	2.74	1.15	1.42	2.05	1.35	1.49
18	3.75	2.76	2.74	1.81	2.43	1.88	2.95	1.15	2.22	2.12	2.01	1.63
19	3.96	3.72	2.95	2.08	2.64	2.08	3.13	1.42	2.88	3.68	3.40	1.84
20	5.24	5.49	3.54	2.67	3.99	2.67	3.72	2.15	3.51	4.69	3.44	1.81
21	5.24	5.49	3.54	2.74	3.99	2.67	3.72	3.30	3.75	4.48	3.51	2.01
22	6.98	5.63	5.35	4.86	4.58	4.48	4.20	3.65	5.76	4.03	4.38	2.29
23	8.51	6.22	6.88	5.87	6.11	6.04	5.14	3.65	5.80	5.94	5.28	2.22
24	8.68	7.22	7.01	6.56	7.22	7.01	5.94	5.00	7.67	7.67	5.63	2.88

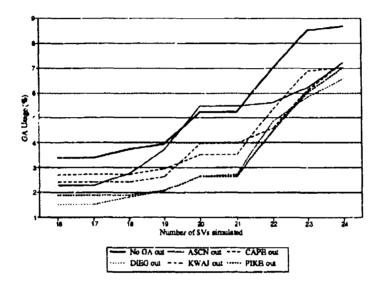


Fig. e 4.19. Percentage of Time Three GAs in Simultaneous Use.

4.4.2.4 Ground Antenna Utilization Results. As with the activity scheduling results, the Ground Antenna utilization rates under the varying experimental conditions confirm the proper operation of the MCS model. Mean utilization in the simulation is within 13% of the sample value for the real MCS GA utilization calculated from MCS logs. The simulated effects of single and dual GA outages on the utilization of the remaining Ground Antennas are consistently explained by the geographical arrangement of the GAs and their coverage areas.

The variations in the percentage of GAs in use for each scenario are consistent and logical. The results, along with crewmember loading studies, could be used to justify changes to crew manning, as an SSO and other crewmembers are "in use" to the same degree the GAs are in use. The results also add more weight to the conclusion that the model is functioning in a manner similar to the MCS and consistent with current MCS operations.

V. Conclusions/Recommendations.

5.1 Conclusions.

The ultimate source for the direction and implementation of this research was the problem of giving the Global Positioning System planners and managers a tool to describe how system growth could affect operational mission performance. Has this tool been created, and what is its efficacy in predicting MCS mission performance?

In the final analysis, the MCS operations simulator created during this research provides results comparable to the MCS in terms of schedule validity and operational effectiveness. It produced an valid satellite contact schedule for the full 24-satellite constellation, with any four of the five Ground Antennas available for use. With three GAs available, the scheduler failed to schedule at most four of 200 activities. The resulting schedule differed from the MCS-generated schedule due to its use of a subset of all the possible MCS activities and the rudimentary scheduling algorithm used in the simulation. While this difference precluded direct comparison of model results to MCS output, the schedule created was sufficiently close to a "real" schedule to cause the simulation GA utilization results to match the known response of the MCS to GA outages. With additional and more detailed rules for defining the scheduling process, the simulation could schedule even closer to reality. Of course, if such a sophisticated, rule-based scheduler was created (one that that produced better schedules than the MCS schedulers), the human element could be taken "out-of-the-loop" at the MCS and the operational and simulated schedules would be identical. An operational simulation under these conditions could predict MCS performance very accurately.

The MCS simulator created here has other useful aspects. The behavior of the scheduler was used to rate the relative significance of a number of MCS failure scenarios. This was done by assuming the human scheduler would have increased difficulty creating a schedule under the same conditions as the simulator. Using this gauge, the various combinations of GA outages were rated by impact to operations, with the combination of a KWAJ/DIEG outage being the most

significant. Another indication was that a DIEG outage was more difficult to schedule than some combinations to two simultaneous GA outages. GA utilization rates produced by the simulation were explained by the geographical distribution of Ground Antennas and the effects of the outages on the world-wide coverage.

With few changes, this model could be adarted to simulate any satellite operational environment. The factors that make the model uniquely MCS-based are all contained in data files, not coded into the simulator program. In fact, with slight modification, the simulation could be used to model any operational environment that is driven by scheduled events with fixed time windows.

5.2 Recommendations.

The simulation as it stands is a major first step in creating a usable MCS operations simulation. It also provides clues to guide the creation of an artificial intelligence-based scheduler or scheduler's assistant. The following list describe recommendations for improvement of the MCS model and directions of further research in this subject.

- 5.2.1 Add Activities. Only a limited number of MCS activities were used in this model, under the assumption that the number used were sufficient to test model performance and perform initial experiments. There are other less-frequently scheduled activities, related to satellite eclipse operation, for instance, that need to be added to improve the fidelity of the model.
- 5.2.2 Extend Environment. The MCS can operate with as many as 30 GPS satellites simultaneously. In addition, if 30 satellites are required someday and the present ground station network proves to be incapable of handling the load, another GA might need to be added. Experiments could be run to test this hypothesis, and to aid in determining the optimum location for the additional GA.

- 5.2.3 GA Selection. The simulation currently steps through the list of GAs in fixed order when selecting the location for scheduling an activity. The next order of embellishment to the scheduling algorithm should include a smarter method for selecting GAs, either a random selection of equally valid choices or more rules to perform the selection smartly.
- 5.2.4 Operations. There was insufficient time and cause to experiment the stochastic operations performance model written into the simulation. This was due to the conservative activity duration times associated with GPS operations. If this model is used to simulate other operations, or research into the effects of activity durations modeled as random variables, the operations simulation segment is there for that use.

Appendix A. MCS Simulation SIMSCRIPT II.5 Program Listing

```
A.1 PREAMBLE
1 ,,
2 ,,
       MCS.SIM
  ,,
  " Global Positioning System (GPS)
  , ,
       Master Control Station (MCS) Simulation
  ,,
  '' Capt David M. Koster
  '' 19 Oct 1992
8
   ,,
9
10
       PURPOSE: This SIMSCRIPT II.5 program creates a valid
11
         MCS (managed by the 2 satellite contact schedule, then
12
         executes that schedule in a manner simulating the
   ,,
13
         performance by the MCS operations crew.
14
15
16 preamble
17 normally, mode is real
18 define minutes to mean units
19
20
   '' This is the list of system queues.
21
22
23
24 the system owns
25
    a TO.BE.SCHEDULED.
26
     a SCHEDULED,
27
     a VIS. TABLE,
28
    'a MASTER,
29
     a PERFORMED.
30
    a UNSCHEDULED
31
32
    '' The only resource entity is used in the operations simulation
33
    '' processes that immediately follow.
34
35
36
37 resources include
38
     CONTACT
39
40 processes include
     PERFORM. ACT.
41
42
     START . OPS
43
44
     every PERFORM. ACT has a THIS. ACT
45
46
     define
        THIS.ACT
47
```

```
48
         as a pointer variable
 49
 50 permanent entities include
<sub>-</sub> δ1
     MASTER, ACT
 52
 53
 54
      '' The MASTER.ACT entities act as a permanent, unchanged record
 55
      '' of the activities required by each SV. An iteration through
 56
      '' MASTER. ACT is used when the various queues of temporary
 57
      " activities must be searched for one of each type of activity
 58
      " for each SV.
 59
 60
 61
      every MASTER. ACT has
                        '' Support name
 62
         A MNAME.
                               '' Start time (abs) of last performance
 63
         a MSTART.TIME,
                              '' Time until act due again (minutes)
 64
         a MLEAD.TIME,
                              '' Duration of activity (minutes)
 65
         a MDURATION,
 66
         a MINTERVAL,
                             '' Max interval between performance
 67
                            '' Status of this activity
         a MSTATUS,
                       '' SV name (launch number format)
 68
         a MSV,
                       '' GA name
 69
         a MGA,
         a MGA.INDEX,
 70
                             '' GA number (internal)
                          '' SV Block number (I or II)
 71
         a MBLOCK,
 72
                            '' Activitiy priority
         a MPRIORITY,
                              '' Duration variance
 73
         a MVARIANCE.
 74
                                '' Importance to mission
         a MCRITICALITY
 75
         and may belong to
 76 the MASTER
 77
  78
      define
  79
         MSTART.TIME,
  80
         MLEAD. TIME.
  81
         MDURATION,
         MINTERVAL,
  82
  83
         MGA.INDEX,
  84
         MBLOCK.
  85
         MPRIORITY.
  86
          MVARIANCE.
  87
          MCRITICALITY
  88
          as integer variables
  89
  90
       define
  91
          HNAME.
  92
          MSTATUS,
  93
          MSV,
  94
          MGA
  95
          as text variables
  96
  97 temporary entities
      include ACT and VIS.EVNT
```

```
99
100
101
      '' The ACT entity is the building block for the scheduling routines
102
      " and the unit of performance during operations simulation. Each
103
      '' entity is self-contained, as its attributes define the state
      " of the activity completely. The attribute definitions are
104
      '' identical to those of the MASTER.ACT, except for the dynamic
105
      '' attributes which the MASTER. ACT entities do not use.
106
107
108
109
      every ACT has
110
         a MAME,
111
         a START. TIME,
                                '' Tentative start of new activity
112
         a WEXT. START. TIME,
113
         a LEAD. TIME,
114
         a DURATION,
115
         a INTERVAL,
                                '' Additional interval time
116
         a INTERVAL OFFSET,
117
         a STATUS.
         a SV,
118
119
         a GA,
120
         a GA. INDEX,
121
         a BLOCK,
122
         a PRIORITY,
123
         a VARIANCE,
124
         a CRITICALITY
125
         and may belong to
126 the TO.BE.SCHEDULED,
127 the SCHEDULED,
128 the PERFORMED,
129 the UESCHEDULED
130
131
         define
132 START. TIME,
133 WEXT.START.TIME,
134 DURATION,
135 INTERVAL,
136 INTERVAL OFFSET,
137 PRIORITY,
138 LEAD TIME,
139 VARIANCE,
140 BLOCK,
141 GA. INDEX,
142 CRITICALITY
143 as integer variables
144
 145
          define
 146 MANE,
 147 STATUS,
 148 SV,
```

149 GA

```
150 as text variables
151
152
153
      " The VIS. BVWT entities maintain the calculated rise and set
154
      '' times of SV/GA pairs. The source of the data is external to
155
      '' this program.
156
157
158
      every VIS.EVET has
159
        a SV,
160
         a GA.
161
         a GA. INDEX,
                         " Rise time of SV at GA
162
        a RISE.TIME,
                       '' Set time of SV at GA
163
         a SET.TIME
164
         and may belong to
165 the VIS.TABLE
166
167
         define
168 RISE.TIME,
169 SET.TIME
170 as integer variables
171
172 ''
173 '' The following commands specify the order in which entities are
174 " maintained in the various system queues. The rationale for
175 '' indicated rankings are described in the routines where the queues
176 '' are used.
177 "
178
179 define
180
     TO . BE . SCHEDULED
181
     as a set ranked by
182
        high PRIORITY, then by
183
         low INTERVAL
184
185
186 define
187 SCHEDULED
188
     as a set ranked by
189
         low START.TIME, then by
190
         low SV
191
192 define
193
     PERFORMED
194
     as a set ranked by
195
        high START.TIME, then by
         low SV
196
197
198 define
199
      CHANGE . FLAG.
                         '' Indicates an exogenous parameter has changed
200
                         '' Minutes since 1 Jan 1992
      CURRENT. TIME,
```

```
201
     EL.LINIT,
                   '' Elevation threshold
                  '' Desired simulation end time
202 END.DAY.
203 EMD. HR.
                  ,,
204
     END. MIN.
                   '' Number of Ground Antennas used
205
     MUM. OF. GA,
                 " Number of Satellite Vehicles used
206
     NUM.OF.SV.
     MUM.OF.OUTAGES, '' Number of original (scheduled) GA outages
207
208 HUM.SIMUL.CONTACTS, '' Max number of simultaneous SV/GA contacts open
    SIM.RMD.TIME, '' Simulation completion time, jminutes
209
                      '' Simulation start time, jminutes
210 SIM.START.TIME,
211 START.DAY, '' Desired simulation start time
                   ,,
212 START.HR,
                     ,,
213 START. MIN.
                      '' Variables for ACT interval offset statistics
214 MMBR.OFFSET.
215 MAX.OFFSET,
                     ,,
216 MIN.OFFSET,
                          **
217
     OPPSET,
                   '' Variables for ACT priority statistics
218 MMBR.PRI,
     MAX.PRI,
                  ,,
219
                           -
    MIM.PRI,
                   , ,
220
               ,,
221
     PRI.
222 MMBR.UTIL,
                    '' Variables for GA utilization statistics
                   ,,
223
     MAX.UTIL,
                    ,,
224 MIN.UTIL,
                          **
                 ,,
225 UTIL,
     MNBR.START.OFFSET, '' Variables for validation offset statistics
226
227
     MAX.START.OFFSET,
                       , ,
                               - 11
228 MIN.START.OFFSET,
                        ,,
229 START.OFFSET,
230 RESV, '' Variables for GA reservation statistics
     MAX.PRIORITY, '' Maximum ACT priority
231
232
     INTERVAL.OFFSET.STEP, '' Value of interval offset increase
233 FAILED, '' Number of failed supports
                     '' Variables for individual GA statistics
234
     ASCH.UTIL.
                     ,,
235 CAPE.UTIL,
236 DIEG.UTIL,
                     ,,
                     ,,
237 KWAJ.UTIL,
                      ,,
238
    PIKE.UTIL,
                      ,,
239 ASCH.RESV,
                      ,,
240 CAPE.RESV,
                     ,,
 241 DIEG.RESV.
 242 KWAJ.RESV,
                     ,,
 243 PIKE.RESV
 244
     as integer variables
 245
 246 define
 247 REMARK1,
                   '' Text description of experiment, parameters, etc.
                    '' Text description of experiment, parameters, etc.
     REMARK2.
 248
                    '' Text description of experiment, parameters, etc.
 249 REMARKS,
                   " Pointer to file containing visibility data
 250
     VIS.FILE.
                    '' Pointer to file containing sim parameter data
 251
     TIME.FILE,
```

```
252
     OUTAGE.FILE.
                         " Pointer to file containing GA outage data
253
     ACT.FILE,
                      " Pointer to file containing initial activity data
254
     OUT.FILE.
                      '' Pointer to file to contain sim results
                    '' Pointer to file to contain validation results
255
     VAL. OUT
256
     as text variables
257
258 ''
     " The GA.USAGE arrays maintains the GA utilization data for each
259
260 '' GA for each minute of the simulation.
261 ''
262
263 define
ን64
     GA. USAGE
26
     as a 2-dim text array
28'
257 "
268
    '' The following TALLY statements invoke autonomous system monitoring
269 '' of the variables named in the last line. Each time the variables
270 '' change, the new value is automatically noted and the listed
271 '' statistics recalculated.
272 ''
273
274 tally
275
     MM.OFFSET as the mean.
276
     STD.OFFSET as the std.dev,
277
      WMBR.OFFSET as the number.
278
     MAX.OFFSET as the maximum.
279
      MIN.OFFSET as the minimum.
280
      HIST.OFFSET(0 to 240 by 10) as the histogram
281
      of OFFSET
282
283 tally
     WMBR.UTIL as the number,
284
285
      ME.UTIL as the mean,
286
     STD.UTIL as the std.dev,
287
      MAX.UTIL as the maximum,
288
      MIN.UTIL as the minimum,
289
      HIST.UTIL(O to 5 by 1) as the histogram
290
      of UTIL
291
292 tally
     NMBR.RESV as the number,
293
294 MW.RESV as the mean,
295
      STD. RESV as the std.dev,
296
      MAX.RESV as the maximum,
297
      MIN. RESV as the minimum,
298
      HIST.RESV(0 to 5 by 1) as the histogram
299
      of RESV
300
301 tally
302 MMBR.PRI as the number,
```

```
303
     MM.PRI as the mean,
304
     STD. PRI as the std.dev,
305
     MAX.PRI as the maximum,
306
     MIN. PRI as the minimum,
307
     HIST.PRI(0 to 11 by 1) as the histogram
308
     of PRI
309
310 tally
311
     NMBR.START.OFFSET as the number,
312
     MW.START.OFFSET as the mean,
313
      STD.START.OFFSET as the std.dev,
314
     MAX.START.OFFSET as the maximum,
     MIW.START.OFFSET as the minimum,
315
316
      HIST.SO(0 to 200 by 10) as the histogram
3:7
      of START. OFFSET
318
319 tally
320
     MN. AUTIL as the mean,
321
      STD. AUTIL as the std.dev
322
     of ASCN.UTIL
323
324 tally
325
      MW.CUTIL as the mean,
326
     STD.CUTIL as the std.dev
327
      of CAPE.UTIL
328
329 tally
     MM.DUTIL as the mean,
330
331
      STD. DUTIL as the std.dev
332
     of DIEG.UTIL
333
334 tally
335
     MW. KUTIL as the mean,
336
      STD.KUTIL as the std.dev
      of KWAJ.UTIL
337
338
339 tally
     MW. PUTIL as the mean.
341
     STD.PUTIL as the std.dev
342
     C PIKE.UTIL
343
344 tally
345
     MM. ARESV as the mean,
346
      STD.ARESV as the std.dev
     c ASCH. RESV
347
348
349 tally
     MW. CRESV as the mean,
350
351
     STD.CRESV as the std.dev
      of CAPE. RESV
352
```

```
354 tally
355 NW.DRESV as the mean,
     STD.DRESV as the std.dev
356
357
     of DIEG.RESV
358
359 tally
360
    ME.KRESV as the mean,
     STD.KRESV as the std.dev
361
     of KWAJ.RESV
362
363
364 tally
365 MW.PRESV as the mean,
366
    STD.PRESV as the std.dev
    of PIKE.RESV
367
368
369 end ''the preamble
```

```
A.2 MAIN
 370
 371
      ****************
 372
     373
     ·*/********************************
 374
 375 main
 376
 377
 378 '' The MAIN program segment control the flow of the program by
 379
     '' defining the execu'ion sequence of the program routines and
 380 '' processes.
 381
- - 382
      '' The first four routines initialize the simulation and are only
      " executed once at the start of the simulation
 384
 385
 286 now RZAD DATA
   7 now READ. VIS
  8 now INIT.GA.USE
 389 now INIT.ACTS
 390
 391
 392 '' The PRESCHEDULE routine kicks off the scheduling sequence. It
 393 '' will be called anytime a new schedule is required.
 394 ''
 395
 396 now PRESCHEDULE
 397
 398
 399 '' The START.OPS process is the first step in the operations
 400 '' simulation process, which starts after the initial schedule has
  401 '' been calculated.
 402
  403
  404 schedule a START. OPS now
  405 start simulation
  406
  407 "
  408 "The remainder of the routines called by MAIN describe the state
  409 '' of the simulation after the simulation has completed, and
  410 '' statistics pertaining to critical performance measures.
  411
  412
  413 now REPORT.USE
  414 now REPORT . QUEUES
  415 now ANALYSIS
  416 new VALIDATE
  417
      stop
  418
  419 end' main
```

```
A.S READ. DATA Routine
          420
1<del>2 4</del>2, 1.4. -421
               *****************
          422
               425
          424
              routine READ.DATA
          425
          426
          427
          428 "This routine "bootstraps" the remainder of the initialization
          429 '' sequence. It reads data from the system default data file, which
          430 '' sets the basic simulation parameters and points to the subsequent
              " input data files. This "indirect" input method allows all
              " simulation parameters to be changed by modifying the default data
          432
              " file only.
          433
          434
          435
          436
              '' Where are the other data files?
          437
          438 ''
          439
          440 read
          441
               VIS.FILE,
               TIME.FILE,
          442
               OUTAGE.FILE.
          443
          444
               ACT.FILE,
          445
               OUT.FILE,
               VAL.OUT
          446
          447
          448
               "These variables allow the user to describe the simulation run.
           450 "They are later included with the output files to "stamp" the
              '' simulation operating conditions on the output.
           451
               ,,
           452
           453
           454 read
           455
               REMARK1.
               REMARK2.
           456
           457
                REMARK3
           458
                as 3 T *
           459
           460 open unit 2 for in; %, name is TIME FILE
           461 use unit 2 for input
           462
           464 '' head the simulation time period desired.
               ,,
           465
           468
           467 read
           468
               START.DAY,
               START.HR,
           469
```

```
470 START. HIE.
471 RND.DAY,
     RND. HR.
472
      END. MIN
473
474
475
476 '' Convert the desired time limits to "jminutes".
 477
478
 479 let SIM.START.TIME = START.DAY+1440 +
       START.HR+60 +
 480
 491
        START.MIN
482 let SIM.END.TIME = END.DAY+1440 +
483 BND. HR+60 +
 484
     RWD . MIN
 485 let CURRENT.TIME = SIM.START.TIME
 486
 487
 488 '' Read the number of simultaneous contacts allowed, the maximum
 489 '' activity priority, and the initial interval offset incrementation
 490 '' value.
     . .
 491
 492
 493 read
 494 WUM.SIMUL.CONTACTS,
 495 MAX.PRIORITY,
 496 INTERVAL. OFFSET. STEP
 497
 498 close unit 2
 499
 500 "
 501 '' The operations simulations process uses resources to keep track
 502 '' of GA utilization. These resources are created now.
 503 ''
 504
 505 create every CONTACT(1)
 506 let u.CONTACT(1) = NUM.SIMUL.CONTACTS
 507
 508 end''READ.DATA
```

```
A.4 READ. VIS Routine
509
     510
511
     <sup>,</sup>,*******************************
    ****************************
512
513
514 routine READ. VIS
Б15
516
517
    '' This routine is used to read the externally prepared visibility record.
518
    '' The data must be synchronized with the simulation time period and the
    '' preliminary activity data. The visibility data is then used to
519
520
    '' initialize VIS. EVMT entities and the VIS. TABLE set.
Б21
522
523 define
                 '' Counter
524
     .i,
525
                    '' Counter
526
      .NUM.OF.EVHTS, '' The number of visibility periods to be read
Б27
                    '' The day of the month the SV rises at GA
      . RDAY.
                   '' The hour of the month the SV rises at GA
528
      .RHOUR.
                   '' The minute of the month the SV rises at GA
529
      .RMIN.
530
                   '' The day of the month the SV sets at GA
      .SDAY.
                  '' The hour of the month the SV sets at GA
531
      . SHOUR.
                    '' The minute of the month the SV sets at GA
532
      .SM1W,
                  '' The numerical "tag" for this GA
533
      .GA.INDEX,
534
      . JDAY . OFFSET
                  " The number of days from 1 Jan to the 1st of
               ,,
535
                   the month of the simulation
536
      as integer variables
537
538
     define
539
      .SV.
               '' SV text name
              " The GA text name
540
      . GA
541
      as text variables
542
543
     open unit 2 for input, name is VIS.FILE
544
     use unit 2 for input
545
546
     " First, the number of Ground Antennas used in the alkulation is read,
547
548
     '' then the days from 1 Jan to the start of the n and the simulation
     '' begins. Note that for simplicity, the simul
                                                          nnot "urap"
     '' around a month boundry.
550
551
552
553
554
     '' Here is a sample of the data format for the visibility data file,
     " for reference.
555
     ,,
556
          5 90
557
                  "NUM. OF. GA, . JDAY. OFFSET"
     '' ASCN 1
558
                  ".GA,.GA.INDEX"
```

```
559 '' BII-26 9 18 39 1 11
                                   ".SV..RDAY,.RHOUR,.RMIN,.SHOUR,.SMIN
560 '' ...
561 '' BII-06 14 23 4 5 42
562 ''
        END 0 0 0 0 0
563 ,,
         CAPE 2
564 ''
        BII-04 9 18 17 1
565 '' ...
566 '' BI-010 14 23 56 4 50
567 " END 0 0 0 0
568 ''
569
570 read
571 NUM.OF.GA.
572 . JDAY. OFFSET
673
574 for .i = 1 to NUM.OF.GA do
575
576
     read
577
        .GA,
578
        .GA.IMDEX
579
680 let .SV = ""
681 until .SV = "BMD" do
582
583
         read
584 .SV.
585 RDAY,
586 . RHOUR,
587 .RMIN,
588 .SHOUR,
589 .SMIN
590
591
         if .SV ne "END"
592
593 create a VIS.EVNT
594
595 ''
596 '' Make a correction to the set day if the event overlaps
 597 '' 0000hrs.
 598 ''
 599
 600 if .RHOUR <= .SHOUR
       let .SDAY = .RDAY
 601
 602 else
      let .SDAY = .RDAY + 1
 603
 804 endif
 605
 606 "
 607 '' Calculate the number of minutes from 0000 1 Jan (called
 608 '' jminutes).
 609 ''
```

```
610
      611 let RISE TIME(VIS.EVNT) = 1440+(.RDAY + .JDAY.OFFSET) +
      612 60*.RHOUR +
613 .RMIN
      614 let SET.TIME(VIS.EVET) = 1440+(.SDAY + .JDAY.OFFSET) +
      615 60+.SHOUR +
      616 .SMIN
      617 let GA(VIS.EVET) = .GA
      618 let GA.INDEX(VIS.EVNT) = .GA.INDEX
      619 let SV(VIS.EVNT) = .SV
      620
      621 "
      622 " Once the entity has its attributes set, it is filed in
      $23 '' the VIS.TABLE queue by low START.TIME, which amounts to
      624 '' chronological order.
      625 "
      626
      627 file this VIS.EVWT in VIS.TABLE
      628
      629
               endif
      630
           loop
      631 loop
      632
       633 close unit 2
       634
       635 end ''routine READ.VIS
```

general and a service services of a service se

```
A.5 INIT.GA. USE Routine
636
637
    ******************************
638
    ***********************
639
    *******************************
640
641 routine INIT.GA.USE
642
    ,,
643
644 '' This routine reads in an initial ground antenna use schedule. The
645 '' use could be from prescheduled contacts, scheduled GA maintenance,
646 '' MCS scheduled maintenance or testing, or, in the case of validation,
647 '' unscheduled outages that occured during the validation period. An
    '' array is used to reduce storage requirements and improved processing
    '' time.
649
    ,,
650
651
652 define
653
     .GA.
    .GA.INDEX,
654
655
    .START.TIME.
656
      .DURATION,
657
      .i.
668
      .OFFSET '' Used to translate jminutes to array index
660
     as integer variables
661
662 define
663
     .CAUSE '' The reason for the outage.
664
      as a text variable
665
666
667 reserve
668
    GA.USAGE(+,+)
669
     as 5 by (SIM.END.TIME - SIM.START.TIME + 1)
670
671 open unit 2 for input, name is OUTAGE.FILE
672 use unit 2 for input
673
674
675
    " A sample outage file is shown below:
676
    ,,
677
         "ASCH out for entire simulation."
                                           "REMARK1"
678 ''
         "DIEG out for entire simulation."
                                           "REMARK2"
679
         2 "WUN.OF.OUTAGES"
680
    ,,
         1 144000 2880 DUTAGE
                               ".GA.INDEX,.START.TIME,.DURATION, .CAUSE"
681
         3 144000 2880 OUTAGE
632 ''
683
684 read
685 REMARK1,
```

로 프로그램 (1951년 1962년 전 1962년 1962 - 1962년 1962년

```
686
       REMARK2
       as 2 T +
 687
 688
 689 read NUM.OF.GUTAGES
 690
 691 if MUM.OF.OUTAGES > 0
 692
 693
      for .GA = 1 to NUM.OF.OUTAGES do
 694
 695
          read
 696 .GA.INDEX,
 697 .START.TIME,
  698 . DURATION,
 699 .CAUSE
 700
          let .OFFSET = .START.TIME - SIM.START.TIME
  701
          for .i = .OFFSET to .OFFSET + .DURATION do
  702 let GA.USAGE(.GA.INDEX, .i+1) = .CAUSE
  703
          loop
  704
      loop
  705
  706 endif
  707
  708 close unit 2
  709
  710 ''
  711 " The following routine replaces the default vaules for the
  712 '' GA.USAGE array (a null string) with 10 spaces. This improves
  713 '' the appearance of the array when output.
  714 ''
  715
....716 for .i = 1 to 5 do
  717 for .j = 0 to SIM.END.TIME - SIM.START.TIME do
           if GA.USAGE(.i, .j+1) = ""
  718
  719 let GA.USAGE(.i, .j+1) = "
  720
           endif
  721
       loop
  722 loop
  723
  724 end ''routine INIT.GA.USE
```

```
A.6 INIT.ACTS Routine
725
726
    ******************************
727
    *****************************
728 *******************************
729
730 routine INIT.ACTS
731
732
733 '' This routine initializes the ACT entities needed to "hotstart" the
734 " simulation. They provides the necessary contact history (prior to
735 '' the start of the simulation) so the priority of the contacts can be
736 '' calculated.
737 "
738
739 define
740 .SV,
741 .BLOCK.
742 .START.TIMB,
    .DURATION,
743
744 .INTERVAL,
745 .PRIORITY,
746 . VARIANCE.
747 .CRITICALITY,
748
     . WUM . OF . ACTS
749 as integer variables
750
751 define
752
     .SV. MAME,
753
     .ACT. HAMB
764 as text variables
755
756 open unit 2 for input, name is ACT.FILE
757 use unit 2 for input
758
759 **
760 '' Here is a sample ACT.FILE:
761 **
762 '' 16
               "MUM.OF.SV"
763 '' 61
               ".NUM.OF.ACTS"
764 '' BI-008 1 ADDKEYS 143815 10 1 1 4320
765 '' BI-008 1 MAV
                       143232 5 1 1 1440
766 '' BI-008 1 SOE
                        143805 10 1 1 480
767 ''
768 ''
        ".SV.WAME, .BLOCK, .ACT.WAME, .START.TIME, .DURATION,
769 '' . VARIANCE, . CRITICALITY, .INTERVAL"
770 ''
771 read
772 WUM.OF.SV,
 773 . NUM. OF. ACTS
```

international designation of the con-

```
775 create every MASTER.ACT(.MUM.OF.ACTS)
775
777
    for every NASTER. ACT do
778
779
     read
780
         .SV. WAME,
781
         .BLOCK.
782
        . ACT . WAME,
783
        .START.TIME.
784
         .DURATION,
785
        .VARIANCE,
786
        .CRITICALITY.
787
         . INTERVAL
788
789
      let MSV(MASTER.ACT) = .SV. MANE
790
      let MBLOCK(MASTRR.ACT) = .BLOCK
791
      let MSTATUS(MASTER.ACT) = "UMSCHD"
792
      let MWAME(MASTER.ACT) = .ACT.WAME
793
      let MSTART.TIME(MASTER.ACT) = .START.TIME
      let MDURATION(MASTER.ACT) = .DURATION
794
795
      let MINTERVAL(MASTER.ACT) = .INTERVAL
796
      let MCRITICALITY(MASTER.ACT) = .CRITICALITY
797
      let MVARIANCE(MASTER.ACT) = .VARIANCE
798
799
      '' The ordering of the MASTER.ACT entities in MASTER is not
800
      '' important.
801
802
803
804
      file this MASTER. ACT in MASTER
805
806
     create an ACT
807
808
     let SV(ACT) = MSV(MASTER.ACT)
809
      let BLOCK(ACT) = MBLOCK(MASTER.ACT)
810 let STATUS(ACT) = MSTATUS(MASTER.ACT)
811
      let MANE(ACT) = NHAME(MASTER.ACT)
812
     let LEAD.TIME(ACT) = MLEAD.TIME(MASTER.ACT)
813 let START.TIME(ACT) = MSTART.TIME(MASTER.ACT)
814 let DURATION(ACT) = MDURATION(MASTER.ACT)
815
      let INTERVAL(ACT) = MINTERVAL(MASTER.ACT)
     let CRITICALITY(ACT) = MCRITICALITY(MASTER.ACT)
817
      let VARIANCE(ACT) = MVARIANCE(HASTER.ACT)
818
819
820
      '' The ordering of the ACT entities in PERFORMED is by high
821
       '' START.TIME, so that when the queue is searched the latest
       " example of each type of entity is found first.
822
       ,,
823
824
825
      file this ACT in PERFORMED
```

```
826
827 loop
828
829 close unit 2
830
831 end'' routine INIT.ACTs
```

```
A.7 MAKE.NEW.ACT Routine
832
    833
834
    836
837 routine MAKE. MEW. ACT given . ACT yielding . MEW. ACT
838
839 ,,
840 '' This routine simply takes the ACT entity whose pointer it is
841 '' passed and duplicates it, creating a new entity with the same
842 '' attribute values. The pointer of the new entity is then
843 '' passed back to the calling routine.
844
845
846 define
847
    . MEW. ACT,
848
    . ACT
849
    as pointer variables
850
851 create an ACT called .MEW.ACT
852
853 let MANE(.MEW.ACT) = MANE(.ACT)
854 let SV(.MEW.ACT) = SV(.ACT)
855 let BLOCK(.MEW.ACT) = BLOCK(.ACT)
856 let START.TIME(.WEW.ACT) = START.TIME(.ACT)
857 let WEXT.START.TIME(.WEW.ACT) = WEXT.START.TIME(.ACT)
858 let DURATION(.MEW.ACT) = DURATION(.ACT)
859 let IMTERVAL(.MEW.ACT) = IMTERVAL(.ACT)
860 let INTERVAL.OFFSET(.MEW.ACT) = INTERVAL.OFFSET(.ACT)
861 let VARIANCE(.MEW.ACT) = VARIANCE(.ACT)
862 let CRITICALITY(.WEW.ACT) = CRITICALITY(.ACT)
863 let STATUS(.BEW.ACT) = STATUS(.ACT)
864 let PRIORITY(.MEW.ACT) = PRIORITY(.ACT)
865 let GA(.WEW.ACT) = GA(.ACT)
866 let GA.IMDEX(.MEW.ACT) = GA.IMDEX(.ACT)
867
 868 end'' routine MAKE. HEW. ACT
```

```
A.8 PRESCHEDULE Routine
869
870
    ************************
    **************
871
    872
873
874 routine PRESCHEDULE
875
876
877 " FUNCTION:
          - Create one ACT for each SV activity that needs to be performed
878 ''
879 ''
            at least once more prior to the end of the simulation, and place
880 ,,
            this ACT in TO.BE.SCHEDULED if there isn't one corresponding
    ,,
881
            to that activity in TO.BE SCHEDULED already.
    ,,
882
883
    " IMPUT CONDITIONS:
884 ''
         - One MASTER.ACT entity filed in MASTER for each routinely
885 ''
            required SV activity for each SV.
886 ''
887 ''
          - One ACT entity filed in PERFORMED as above, with STARY.TIME(ACT)
888 ''
            indicating the last time this activity was performed prior
    ,,
889
            to the simulation start time.
890
891
892 define
     . NEW . ACT
893
894
      as pointer variables
895
896 print 1 line thus
PRESCHEDULE
898
899
900 '' Iterate through the list of activities, taking each ACT
    '' one at a time.
901
    ,,
902
203
904 for every MASTER. ACT in MASTER do
905
906
907
      '' Find the latest occurrance of the specific ACT in the
      " PERFORMED queue.
809
      ,,
909
910
      for every ACT in PERFORMED
911
912 with SV(ACT) = MSV(MASTER.ACT) and
         WAME(ACT) = MWAME(MASTER.ACT)
913
914 find the first case
      if found
915
916
917
         " This ACT need onlt be considered for further performance if
918
```

```
919
         " it can be completed prior to the end of the simulation
920
921
922
         if START.TIME(ACT) +
       INTERVAL(ACT) +
923
924
       INTERVAL.OFFSET(ACT) +
925
      DURATION(ACT) < SIM.END.TIME
926
927 **
928 '' This ACT meets all the criteria for being scheduled again.
929 '' Make a copy, change the status to "UNSCHD", and file it in
930 '' TO.BE.SCHEDULED. In addition, trigger the collection of
931 '' the performance statistics on this entity.
932 **
933
934 now MAKE. HEW. ACT giving ACT yielding . NEW. ACT
935 let STATUS(.MEW.ACT) = "UMSCHD"
936 let LEAD.TIME(.MEW.ACT) = START.TIME(.MEW.ACT) +
937 INTERVAL(.NEW.ACT) -
938
      CURRENT. TIME
939 let OFFSET = INTERVAL.OFFSET(ACT)
940 let PRI = PRIORITY(ACT)
941 let PRIORITY(.MEW.ACT) = 0
942 let INTERVAL.OFFSET(.NEW.ACT) = 0
943 file this .NEW.ACT in TO.BE.SCHEDULED
944
945
         endif
946
      endif
947 loop
948
949 ''
950 '' Do not call the scheduler if there are no activities to be scheduled.
951 '' This routine falls through to the start of the simulation if the
952 '' schedule is completed. The tentatively scheduled activities are
953 '' promoted to the SCHEDULED queue and the others are destroyed.
954 ''
965
956 if TO.BE.SCHEDULED is empty
957
958
      for every ACT in PERFORMED do
959
          if STATUS(ACT) = "TENT"
960
 961 let STATUS(ACT) = "SCHD"
 962 remove this ACT from PERFORMED
 963 file this ACT in SCHEDULED
 964
 965
          else
 966
 967 remove this ACT from PERFORMED
 968 destroy this ACT
```

970 endif 971 loop 972 973 else 974 975 now SCHEDULE 976 977 endif 978 979 end ''PRESCHEDULE

```
A.9 SCHEDULE Routine
980
981
    983
    984
985
986 routine SCHEDULE
987
988
    '' This routine takes all the activity entities in the TO.BE.SCHEDULED
989
990
    '' queue and iteratively tries to find suitable time periods in the
991
    '' schedule for them to be performed. If it succeeds, it calls
    '' PRESCHEDULE to determine if there are further activities required.
    '' If it cannot schedule one or more activities due to rescurce
994 '' it "forgets" what it has done to that point and restarts itself,
    '' except that the troublesome activities have been modified to improve
996
    '' the probability they will be scheduled. If it exhausts all
997
    '' possibility of scheduling an activity, the routine marks the
998
   '' activity as hopeless and henceforth ignores it.
999
1000 define
      .CMCT.START, '' Tentative start time currently under test
1001
      .CONFLICT, '' Flag indicating a scheduling conflict occurred
1002
1003
      .T, '' Counter (in minutes)
      .i. '' General purpose counter
1004
1005
      .M.CONTACTS, '' Mumber of contacts currently scheduled
      .OFFSET, '' Maps from sim time to GA.USAGE index
1006
1007
      .START.TIME, '' Temporary storage for START.TIME calculations
1008
1009
      .CUT.OFF, '' Earliest search time
      .RESCHD.FLAG, '' Set if reschedule is required
1010
1011
      .PRIORITY.STEP '' Current priority step value
1012
     as integer variables
1013
1014 define
      .USE '' temp storage for current GA.USAGE value
1015
1016
     as a text variable
1017
1018 define
1019
      . NEW . ACT
1020 as a pointer variable
1021
1022 let .RESCHD.FLAG = 0
1023 let .PRIORITY.STEP = 1
1024
1025 for every ACT in TO.BE.SCHEDULED
         with STATUS(ACT) = "UNSCHD"
1026
1027
         while STATUS(ACT) = "UNSCHD" do
1028
 1029
      remove this ACT from TO.BE.SCHEDULED
```

```
1030
      let .CUT.OFF = max.f(START.TIME(ACT), CURRENT.TIME)
1031
1032
      for .CECT.START back from START.TIME(ACT) +
1033
        INTERVAL(ACT) +
1034
         INTERVAL.OFFSET(ACT) to .CUT.OFF
1035 with .CMCT.START <= SIM.EMD.TIME
1036 while STATUS(ACT) = "UNSCHO" do
1037
1038
1039
          '' Step thru the visibility periods of the SV that's being
1040
          '' scheduled at any (and all) GAs, provided the period of
          " visibility is longer than the length of the desired contact
1041
1042
1043
1044
          for every VIS.EVMT in VIS.TABLE
1045
        with SV(VIS.EVNT) = SV(ACT) and
1046 RISE.TIME(VIS.EVNT) <= .CNCT.START and
1047 SET.TIME(VIS.EVNT) >= .CMCT.START + DURATION(ACT)
1048
        while STATUS(ACT) = "UNSCHD" do
1049
1050 ''
1051 '' Assume this prospective contact will not conflict
1052 '' with another GA use
1053 ''
1054
1055 let .COMFLICT = 0
1056
1057 ''
1058 '' Step through the minutes of the prospective contact,
1059 '' checking for correspondence in the GA use set.
1060 '' If there is, set the conflict flag.
1051 "
1062
1063 for .t = 0 to DURATION(ACT)
1064
           while .CONFLICT ne 1 do
1085
1066
        let .OFFSET = (.CMCT.START + .t + 1) - SIM.START.TIME
1067
       if GA.USAGE(GA.INDEX(VIS.EVHT), .OFFSET) ne " "
1068
1069
           let .CONFLICT = 1
1070
1071
        endif
1072
1073
       let .M.CONTACTS = 0
1074
        for .GA = 1 to 6
1075 with .GA ne GA.INDEX(VIS.EVNT)
1076 while .COWFLICT ne 1 do
1077
1078
            if substr.f(GA.USAGE(.GA,.OFFSET),1,6) = SV(ACT)
1079 let .CONFLICT = 1
1080
           endif
```

```
1081
1082
           let .USE = GA.USAGE(.GA..OFFSET)
1083
           if substr.f(.USE,1,1) = "B"
1084 add 1 to .W.CONTACTS
           endif
1085
1086
       loop'' next .GA
1087
1088
1089
        if .H.CONTACTS >= NUM.SIMUL.CONTACTS
1090
           let .CONFLICT = 1
1091
        endif
1092
1093 loop'' next .t
1094
1095 ''
1096 '' If this spot is reached with no conflict, the current ACT
1097 '' is tentatively scheduled starting at .CMCT.START.
1099
1100 if .CONFLICT = 0
1101
1102
        let NEXT.START.TIME(ACT) = .CNCT.START
1103
        let GA(ACT) = GA(VIS.EVNT)
1104
        let GA.INDEX(ACT) = GA.INDEX(VIS.EVNT)
        let STATUS(ACT) = "TENT"
1105
1106
1107
        2 )
        '' create a GA.IM.USE event to prevent further
1108
        '' use of this time slot
1109
        ,,
1110
1111
        let .OFFSET = .CNCT.START - SIN.START.TIME
1112
1113
1114
       for i = .OFFSET to .OFFSET + DURATION(ACT) do
           let GA.USAGE(GA.INDEX(VIS.EVNT), .i+1) =
1115
1116
       concat.f(SV(ACT),"/",WAME(ACT))
1117
        loop
1118
1119 endif '' if .COMPLICT = 0
1120
1121
          loop '' next visibility VIS.EVNT
1122
1123
       loop '' next .CMCT.START.TIME
1124
1125
1126
        '' Any activity involved in a conflict will arrive here still UNSCHD.
1127
       '' Either its PRIORITY or INTERVAL.OFFSET will be increased.
1128
       ,,
1129
1130
       if STATUS(ACT) = "UNSCHD"
1131
```

```
1132
          if PRIORITY(ACT) < MAX.PRIORITY
1133
1134 add .PRIORITY.STEP to PRIORITY(ACT)
1135 add 1 to .PRIORITY.STEP
1136 let STATUS(ACT) = "RESCHD"
1137 let .RESCHD.FLAG = 1
1138
1139
          else
1140
1141 let STATUS(ACT) = "MISSED"
1142 add INTERVAL.OFFSET.STEP to INTERVAL.OFFSET(ACT)
1143 let .RESCHD.FLAG = 1
1144
1145
          endif '' PRICRITY(ACT) < MAX.PRIORITY
1146
1147
       endif '' STATUS is UNSCHO
1148
1149
1150
      '' If a "conflicting" entity has just had its INTERVAL.OFFSET raised
      "' to one-half its default interval, the routine will mark the place
1151
1152
       '' in the schedule where the activity would have 'naturally" fallen
1153
       '' (keeping its high priority and offset for the data colection
1154
      '' process). It then increments the FAILED counter, and makes a
1186
       '' new activity with default priority and offset. This activity
1156
      '' is placed in TO.BE.SCHEDULED, representing the next time this
1157
       '' activity is needed after the FAILED one.
1158
1159
      if INTERVAL.OFFSET(ACT) <= 0.5 * INTERVAL(ACT)
1160
1161
1162
          file this ACT in TO.BE.SCHEDULED
1163
1164 else
1165
          let STATUS(ACT) = "PAILED"
1166
1167
          let .i = START.TIME(ACT)
1168
          let START.TIME(ACT) = .i + INTERVAL(ACT)
1169
          add 1 to FAILED
1170
          now MAKE. MEW. ACT giving ACT yielding . MEW. ACT
1171
          subtract 1 from PRIORITY(.MEW.ACT)
1172
          file this . WEW. ACT in PERFORMED
1173
          1et PRIORITY(ACT) = 0
1174
          let INTERVAL.OFFSET(ACT) = 0
1175
          file this ACT in TO.BE.SCHEDULED
1176
1177
       endif
1178
 1179 loop ''for every ACT in TO.BE.SCHEDULED
 1180
 1181
 1182 '' If every activity was scheduled, the new activities are updated
```

```
1183 '' and moved to PERFORMED to form the basis for the next activity
1184 '' selection process. If not, this iteration of SCHEDULE is undone,
1185 '' the activities made UNSCHD, then SCHEDULE called again.
1186 ''
1187
1188 if .RESCHD.FLAG = 0
1189
1190
     for every ACT in TO.BE.SCHEDULED do
1191
1192
         let STATUS(ACT) = "TEMT"
1193
        let START.TIME(ACT) = NEXT.START.TIME(ACT)
1194
        remove this ACT from TO.BE.SCHEDULED
1195
         file this ACT in PERFORMED
1196
1197
     loop
1198
     now PRESCHEDULE
1199
1200
1201 else
1202
1203 for every ACT in TO.BE.SCHEDULED
1204 with STATUS(ACT) = "TENT" do
1205
1206
         let .OFFSET = WEXT.START.TIME(ACT) - SIM.START.TIME
1207
         for .i = .OFFSET to .OFFSET + DURATION(ACT) do
1208
         let GA.USAGE(GA.INDEX(ACT),.i+1) = " "
1209
          loop
1210
1211
      loop
1212
1213 for every ACT in TO.BE.SCHEDULED,
         1et STATUS(ACT) = "UNSCHD"
1214
1215
1216 now SCHEDULE
1217
1218 endif
1219
1220 end ''routine SCHEDULE
```

이는 이 현존되면 그림으로 가득했습니다. 이 시

```
A. 10 START. OPS Process
1222
1223
     ***************************
1224 *****************************
1226
1227 process START.OPS
1228
1229 ''
1230 '' This process calls itself every simulation minute. When called,
1231 '' it updates the CURRENT.TIME, checks for available CONTACT
1232 '' resources and, if any are available, draws the next ACT from
1233 " the SCHEDULED queue andassigns it to a PERFORM.ACT process.
1234 ''
1235
1236 define
1237
      . i
1238
     as an integer variable
1239
1240 let CURRENT.TIME = SIM.START.TIME + time.v
1241
1242 if SCHEDULED is not empty
1243
     if u.COMTACT(1) > 0
1244
1245
1246
         for .i = 1 to u.CONTACT(1) do
1247
1248 for every ACT in SCHEDULED,
1249
       find the first case
1250
1251 remove this ACT from SCHEDULED
1252 activate a PERFORM.ACT giving ACT now
1253
1254
         loop
      endif
1255
1256 endif
1257
1258 If CURRENT.TIME < SIM.END.TIME
1259
1260 schedule a START.OPS in 1 minutes
1261
1262 endif
1263
1264 end''START.OPS
```

```
A.11 PERFORM.ACT Process
1265
1266
     ***************
1267
     ******************************
1268
    ****************************
1269
1270 process PERFORM. ACT given THIS. ACT
1271
1272 '' This process takes the activity entity passed from
1273 '' START.OPS and:
1274 "
           - grabs one of the CONTACT resources,
1275
    ,,
           - adds variation to the activity, if required,
1276
          - tests to ensure the activity can be performed
1277 ''
            as scheduled,
1278 ''
           - simulates the performance of the valid activities, and
1279 ''
           - calls for a new schedule if an activity cannot be
1280 ''
             performed.
1281
1282 define
1283
      .i,
      .OFFSET, '' Minutes into simulation
1284
1285
      .DURATION .
1286
      .OFF.SCHEDULE '' Set to one if activity cannot be performed
1287
     as integer variables
1288
1289 define
1290
      .USE
1291
      as a text variable
1292
1293 define
1294
      THIS.ACT
1295 as a pointer variable
1296
1297 request 1 CONTACT(1)
1298
1299
1300 '' These statements are modified if activity duration variance
1301 '' is desired.
1302 ''
1303
1304 let .DURATION = DURATION(THIS.ACT)
1305 let DURATION(THIS.ACT) = .DURATION
1306
1307 let .OFF.SCHEDULE = 0
1308
1309 ''
1310 '' It is good if the activity start time equals the current sim time.
1311 '' Test further and take corrective action if they are not equal.
1312 ''
1313
1314 if START.TIME(THIS.ACT) ne CURRENT.TIME
```

```
1315
1316
      '' If the start time if later than the current time, the process
1317
      '' halts until time catches up. If earlier, check the visibility
1318
1319
       '' table and GA use for the possibility the activity cannot be
1320
       " performed now. If it cannot, set the .OFF.SCHEDULE flag.
1321
1322
1323
       if START.TIME(THIS.ACT) > CURRENT.TIME
1324
1325
         wait START.TIME(THIS.ACT) - CURRENT.TIME + 1 minutes
1326
1327
       else
1328
1329
         1et .OFFSET = CURRENT.TIME - SIM.START.TIME
1330
1331
         for .i = .OFFSET to .OFFSET + DURATION(THIS.ACT)
1332
        while .OFF.SCHEDULE = 0 do
1333
1334 for every VIS.EVNT in VIS.TABLE
1335
           with GA(VIS.EVNT) = GA(THIS.ACT) and
1336
        SV(VIS.EVMT) = SV(THIS.ACT) and
        RISE.TIME(VIS.EVET) >= CURRENT.TIME + (.i - .OFFSET) and
1337
1338
        SET.TIME(VIS.EVMT) <= CURRENT.TIME + (.i - .OFFSET)
1339 find the first case
1340 if none
1341
1342
        let .OFF.SCHEDULE = 1
1343
1344 endif
1345
1346 let .USE = GA.USAGE(GA.IWDEX(THIS.ACT), .i+1)
1347 if .USE ne "
                       " and .USE ne SV(THIS.ACT)
1348
1349
        let .OFF.SCHEDULE = 1
1350
1351 endif
1352
          1000
1353
       endif
1354 endif
1355
1366 if .OFF.SCHEDULE = 0
1357
1358
       " The activity can be performed now...hold on to the resource
1359
       " for the duration of the activity, then file the activity in
1360
1361
       " PERFORMED.
1362
1363
1364
      let START.TIME(THIS.ACT) = CURRENT.TIME
1365
       let STATUS(THIS.ACT) = "PERF"
```

```
1366
1367 wait DURATION(THIS.ACT) minutes
1368 relinquish 1 CONTACT(1)
1369
      file THIS.ACT in PERFORMED
1370
1371 else
1372
1373
1374 '' Remove the the scheduled activities from SCHEDULED, clear
       '' their GA reservations from GA.USAGE, give up the CONTACT
1375
1376 '' token, and call PRESCHEDULE to create a new schedule.
1377 ''
1378
1379 for every ACT in SCHEDULED
1380 with START.TIME(ACT) >= CURRENT.TIME do
1381
1382
         remove this ACT from SCHEDULED
1383
1384
         let .OFFSET = START.TIME(ACT) - SIM.START.TIME
1385
1386
          for .i = .OFFSET to .OFFSET + DURATION(ACT) do
1387 let GA.USAGE(GA.INDEX(ACT),.i+1) = "
1388
          loop
1389
1390
          destroy this ACT
1391
1392
       loop
1393
1394 relinquish 1 CONTACT(1)
1395 now PRESCHEDULE
1396
1397 endif
1398
1399
1400 end''PERFORM.ACT
```

```
A.12 REPORT. QUEUES Routine
1401
     ************************
1402
1403
     ****************************
1404
1405
1406 routine REPORT.QUEURS
1407
1408
     '' This routine prints the contents of the systems queues.
1409
1410
1411
1412 define
1413
      . i,
1414
      ٠j
1415
      as integer variables
1416
1417 print 1 line with CURRENT.TIME thus
REPORT QUEUES: Current time = *****
1419
1420 if SCHEDULED is not empty
1421 print 2 lines thus
SCHEDULED:
START
          (MEXT) SV/ACT DUR/INT
                                   PRI/OFFSET
1424 for each ACT in SCHEDULED do
1425 print 1 line with
1426
         trunc.f(START.TIME(ACT)/1440),
1427
          trunc.f(frac.f(START.TIME(ACT)/1440)+24),
         mod.f(mod.f(START.TIME(ACT),1440),60),
1428
         trunc.f(WEXT.START.TIME(ACT)/1440),
1429
1430
         trunc.f(frac.f(MEXT.START.TIME(ACT)/1440)+24).
1431
         mod.f(mod.f(WEXT.START.TIME(ACT),1440),60),
1432
          SV(ACT),
1433
          MANE(ACT)
1434
          DURATION (ACT),
1435
          INTERVAL(ACT),
1436
          PRIORITY(ACT),
1437
          INTERVAL.OFFSET(ACT),
          GA(ACT) thus
1438
***/**:** (***/**:**) ************* **/**** **/***
1440 loop
1441 endif
1442
1443 if TO.BE.SCHEDULED is not empty
1444 print 2 lines thus
TO.BE.SCHEDULED:
START
          (NEXT) SV/ACT DUR/INT
                                    PRI/OFFSET GA
1447
1448 for each ACT in TO BE SCHEDULED do
1449 print 1 line with
1450
         trunc.f(START.TIME(ACT)/1440),
```

```
1451
         trunc.f(frac.f(START.TIME(ACT)/1440)*24),
1452
         mod.f(mod.f(START.TIME(ACT),1440),60),
1453
         trunc.f(MRXT.START.TIMB(ACT)/1440).
1454
         trunc.f(frac.f(WEXT.START.TIME(ACT)/1440)+24),
1455
         mod.f(mod.f(MEXT.START.TIME(ACT),1440),60),
1456
         SV(ACT).
1457
         MAME(ACT).
1458
         DURATION(ACT).
1459
         INTERVAL(ACT).
1460
         PRIORITY(ACT).
1461
         INTERVAL OFFSET (ACT).
1462
         GA(ACT) thus
1464 loop
1465 endif
1466
1467 if PERFORMED is not empty
1468 print 2 lines thus
PERFORMED:
START
         (MEXT) SV/ACT
                        DUR/INT
                                   PRI/OFFSET
1471 for each ACT in PERFORMED do
1472 print 1 line with
1473
         trunc.f(START.TIME(ACT)/1440),
1474
         trunc.1(frac.f(START.TIMB(ACT)/1440)*24),
1475
         mod.f(mod.f(START.TIME(ACT), 1440), 60),
1476
         trunc.f(MEXT.START.TIME(ACT)/1440),
1477
         trunc.f(frac.f(MEXT.START.TIME(ACT)/1440) *24),
1478
         mod.f(mod.f(MEXT.START.TIME(ACT),1440),60),
1479
         SV(ACT),
1480
         NAME(ACT),
1481
         DURATION(ACT),
1482
         INTERVAL(ACT),
1483
         PRIORITY(ACT).
1484
         IMTERVAL.OFFSET(ACT),
1485
          GA(ACT) thus
***/**:** (***/**:**) ************** **/**** **/***
1487 loop
1488 endif
1489
1490 if UNSCHEDULED is not empty
1491 print 2 lines thus
UNSCHEDULED:
START
          (NEXT) SV/ACT DUR/INT PRI/INT.OFFSET
1494 for each ACT in UNSCHEDULED do
1495
     print 1 line with
1496
          trunc.f(START.TIME(ACT)/1440),
1497
          trunc.f(frac.f(START.TIME(ACT)/1440)*24),
1498
          mod.f(mod.f(START.TIME(ACT),1440),60),
1499
          trunc.f(MEXT.START.TIME(ACT)/1440),
1500
          trunc.f(frac.f(WEXT.START.TIME(ACT)/1440)+24),
1601
          mod.f(mod.f(MEXT.START.TIME(ACT),1440),60),
```

5

```
A.13 REPORT. USE Routine
1513
1514
    1515
     ****************
     1516
1517
'518 routine REPORT.USE
1519
1520 ''
1521
    '' This routine prints the GA use array to the standard
    '' output file. Minutes if no GA in use are not printed.
1522
1523
1524
1825 define
1526
    .i,
1527
1528
     as integer variables
1529
1530 print 1 line with CURRENT.TIME thus
REPORT USE: Current time = *****
1532
1533 for .i = 1 to SIM.END.TIME - SIM.START.TIME do
1534
     let .j = .i + 144000
1535
     if GA.USAGE(1,.i) ne "
1536
      GA.USAGE(2,.i) ne " or
1537
1538
       GA.USAGE(3, i) ne "
                           " or
       GA.USAGE(4,.i) ne "
1539
                          " or
       GA.USAGE(5,.i) ne "
1540
1541
1542
        print 1 line with
1543 trunc.f(.j/1440),
1544 trunc.f(frac.f(.j/1440)+24),
1545 mod.f(mod.f(.j,1440),60),
1546 GA. USAGE(1,.i),
1547 GA. USAGE(2,.i),
1548 GA. USAGE(3,.i),
1549 GA. USAGE(4,.i),
1550 GA. USAGE(5,.i) thus
***/**:**
           ******** ******* ******* ******* *****
     endif
1552
1553 loop
1554
1555 end'', REPORT. USE
```

```
A. 14 REPORT. VIS Routine
1556
1557
     ***
1558
1559
     ·*******************************
1560
1561 routine REPORT.VIS
1562
1563 ''
1564 '' This routine prints the visibility table to the standard
1565 '' output file.
1566 ''
1567
1568 print 1 line with CURRENT.TIME thus
REPORT VIS: Current time = *****
1570
1571 for each VIS.EVNT in VIS.TABLE do
1572 print 1 line with
1573
         trunc.f(RISE.TIME(VIS.EVNT)/1440),
1574
         trunc.f(frac.f(RISE.TIME(VIS.EVNT)/1440)+24),
1575
        trunc 1/2 rac.f(frac.f(RISE.TIME(VIS.EVNT)/1440) +24) +60),
1576
       trope Attack.TIME(VIS.EVNT)/1440),
1577
        trunc.t(: ac.f(SET.TIME(VIS.EVNT)/1440)*24),
1578
        trunc.:(frac.f(frac.f(SET.TIME(VIS.EVNT)/1440)+24)+60),
1579
         RISE.TIME(VIS.EVNT),
         SET.TIME(VIS.EVNT),
1580
        GA(VIS.EVMT),
1581
1582
       GA.INDEX(VIS.EVNT).
1583
       SV(VIS.EVNT) thus
***/**:** ***/**:** (******-*****) ****(*) ******
1585 loop
1586
1587 end '' REPORT.VIS
```

```
A.15 ANALYSIS Routine
1588
1589
     *****************
     ***************
1590
     * **************
1591
1592
1593 routine ANALYSIS
1594
1595
1596 '' This routine print the simulation conditions, parameters, and
1597 '' summarizes the results in terms of the predefined
     '' performance measures. The output is saved to the filename
1598
     " defined by OUT.FILE.
1599
1600
1601
1602 define
1603
     ιi,
     .GA.INDEX.
1604
1605
     .START.TIME.
1606
     .DURATION,
1607
      ٠j.
      .RESV.TEMP,
1608
1609
      .UTIL.TEMP
     as integer variables
161C
1611
1612 define
1613
     . CAUSE
1614
     as a text variable
161E
1616 open unit 3 for output,
1617
     name is OUT.FILE
1618 use unit 3 for output
1619
1620 ''
1621
     '' This crude but effective routine scans the GA.USE array and
1622 " results in utilization statistics for individual and total
1623 '' GA use.
1624
1625
1626 for .i = 1 to SIM.END.TIME - SIM.START.TIME do
1627
1628
     let .UTIL.TEMP = 0
1629
     let .RESV.TEMP = 0
     if substr.f(GA.USACE(1,.i),1,1) = 'B"
1630
1631
         let ASCN.UTIL = 1
1632
         add 1 to .UTIL.TEMP
1633
     else
1634
         let ASCN.UTIL = 0
1835
      endif
      if substr.f(GA.USAGE(1,.i),1,1) ne " "
1636
1637
         let ASCN.RESV = 1
```

```
1638
          add 1 to .RESV.TEMP
1639
       else
1640
          let ASCM.RESV = 0
1641
       endif
1642
       if substr.f(GA.USAGE(2,.i),1,1) = "B"
1643
          let CAPE.UTIL = 1
1644
          add 1 to .UTIL.TEMP
1645
       else
          let CAPE.UTIL = 0
1646
1647
       endif
       if substr.f(GA.USAGE(2,.i),1,1) ne " "
1648
          let CAPE.RESV = 1
1649
          add 1 to .RESV.TEMP
1650
1651
       else
1652
          let CAPE.RESV = 0
1653
       if substr.f(GA.USAGE(3,.i),1,1) = "B"
1654
1655
          let DIEG.UTIL = !
1656
          add 1 to .UTIL.TEMP
1657
       e].se
          let DIEG.UTIL = 0
1658
1659
       endif
1660
       if substr.f(GA.USAGE(3,.i),1,1) ne " "
1661
          let DIEG.RESV = 1
1662
          add 1 to .RESV.TEMP
1663
       else
1664
          let DIEG.RESV = 0
1665
       endif
1666
       if substr.f(GA.USAGE(4,.i),1,1) = "B"
          let KWAJ.UTIL = 1
1667
1668
          add 1 to .UTIL.TEMP
1669
       else
1670
          let KWAJ.UTIL = 0
1671
       endif
1672
       if substr.f(GA.USAGE(4,.i),1,1) ne " "
          let KWAJ.RESV = 1
1673
1674
          add 1 to .RESV.TEMP
       else
1675
1876
          let KWAJ.RESV = 0
       endif
1677
1678
       if substr.f(GA.USAGE(5,.i),1,1) = "B"
1679
          let PIKE.UTIL = 1
          add 1 to .UTIL.TEMP
1680
1681
       else
1682
          1ot PIKE.UTIL = 0
1683
       endif
       if substr.f(GA.USAGE(5,.i),1,1) ne " "
1684
          let PIKE.RESV = 1
1685
          add 1 to .RESV.TEMP
1686
1687
       else
1688
          let PIKE.RESV = 0
```

```
1689
      endif
1690
      let UTIL = .UTIL.TEMP
      let RESV = .RESV.TEMP
1691
1692
      print 1 line with
1693
        ٠i,
1694
         .UTIL.TEMP thus
***** *
1696
1697
     loop
1698
1699 print 10 lines with
     START. DAY,
1700
1701
     START. HR,
1702
     START.MIN,
1703 END. DAY,
1704
      END. HR,
1705
      END. MIN,
      SIM.START.TIME,
1706
1707
      SIM. END. TIME,
1708 WUH.OF.SV,
1709
    NUM.OF.GA,
1710 NUM.SIMUL.CONTACTS,
1711
      REMARK1,
1712
      REMARK2,
      REMARK3 thus
1713
SCHEDULING INPUTS
TIME: ***/**: ** to ***/**: ** (***** to ******)
** SVs
** GAS
* Max contacts
*****************
1724 open unit 2 for input,
1725 name is OUTAGE.FILE
1726
1727 use unit 2 for input
1728
1729 read
1730
     REMARK1,
1731
      REMARK2
1732
     as 2 T +
1733
1734 read .i
1735
1736 print 1 line with NUM.OF.OUTAGES thus
** Prescheduled outages:
1739 if NUM.OF.OUTAGES > 0
```

```
1740
1741 print 2 lines thus
        Start Duration
                         Cause
Index
1744 for .i = 1 to NUM.OF.QUTAGES do
1745
1746
         read
1747 .GA. INDEX,
1748 .START.TIME,
1749 .DURATION.
1750 .CAUSE
1751
         print 1 line with
1752
1753 .GA. INDEX,
1754 .START.TIME,
1755 .DURATION,
1756 .CAUSE thus
. .....
                       **********
1758
1759
     loop
1760 endif
1761
1762 close unit 2
1763
1764 print 4 lines thus
RESULTS:
                                    Std Dev
   Number Max
                      Min
                            Mean
 1769 print 6 lines with
 1770 NMBR. OFFSET,
 1771 MAX.OFFSET,
 1772 MIN.OFFSET,
 1773 MW.OFFSET,
 1774 STD.OFFSET,
 1775 WMBR.PRI,
 1776
       MAX.PRI.
       MIN.PRI,
 1777
 1778
       MN.PRI.
       STD.PRI,
 1779
 1780
      WMBR.UTIL,
       MAX.UTIL,
 1781
 1782
       MIN.UTIL,
 1783
       MM.UTIL,
 1784
        STD.UTIL,
 1785
        MMBR.RESV,
 1786
        MAX.RESV,
        MIN.RESV,
 1787
 1788
        MB.RESV,
        STD.RESV,
 1789
 1790 FAILED thus
```

```
PRIORITY:
           ****
                    ****
                             **** **.***
                             **** **.***
                                            **,****
GA UTIL:
           *****
                    ++++
           ****
                             **** **,****
                                            **,****
GA RESV:
                    ****
FAILED SUPPORTS: **
1797 print 4 lines thus
OFFSET:
Value Humber Percent
1802 for .i = 1 to 25 do
1803 print 1 line with
          (.i-1)+10,
1804
1805
         HIST.OFFSET(.i),
        100+HIST.OFFSET(.i)/MMBR.OFFSET thus
1806
1808 loop
1809
1810 print 4 lines thus
PRIORITY:
Value Number Percent
1815 for .i = 1 to 12 do
1816 print 1 line with
1817
          .i-1.
1818
         HIST.PRI(.i).
1819
          100+HIST.PRI(.i)/WMBR.PRI thus
     ****
1821 loop
1822
1823 print 10 lines with
 1824 MW.AUTIL+100,
 1825 MM. ARESV+100,
 1826 STD.AUTIL+100,
      STD.ARESV+100,
 1827
 1828 MM.CUTIL+100,
 1829 MW.CRESV+100,
 1830 STD.CUTIL+100,
 1831 STD.CRES! +100,
 1832 MN.DU. 1 *100,
 1833 MM.DRE3V+100,
 1834 STD.DUT'IL*100,
 1835 STD.DRESV+100,
 1836 MW.KUTIL+100,
 1837 MN. KRESV + 100,
 1838 STD.KUTIL+100,
 1839 STD.KRESV+100,
 1840 MM.PUTIL*100,
 1841 MM.PRESV+100,
```

**** **,****

OFFSET:

```
1842 STD.PUTIL+100,
1843 STD.PRESV+100 thus
INDIVIDUAL GA UTILIZATION(SV)/RESERVATION(ALL):
GA % used/resv std used/resv
ASCN **.**/**.** **.**/**.**
CAPE **. **/**. ** **. **/**. **
DIEG **.**/**.** **.**/**.**
KWAJ **.**/**.** **.**/**.**
PIKE **.**/**.** **.**/**.**
1854 print 4 lines thus
GA UTILIZATION (SV)/RESERVATION (ALL):
Value Number Percent
1859 for .i = 1 to 6 do
1860 print 1 line with
1861
         .i-1,
         HIST.UTIL(.i),
1862
        HIST.RESV(.i),
1863
1864
        100+HIST.UTIL(.i)/WMBR.UTIL,
        100+HIST.RESV(.i)/WMBR.RESV thus
1865
+ ++++/++++
               ***,**/***,**
1867 loop
1868
1869 close unit 3
1870
1871 end''AWALYSIS
```

```
A.16 VALIDATE Routine
1872
1873
      *****************
1874
1875
1876
1877
1878 routine VALIDATE
1879
1880
1881
      " This routine is used only during the simulation validation step,
      '' to provide quantative measure of the ability of the simulation to
1882
1883
      " duplicate the MCS scheduling results. The file "valact.dat"
1884
      '' contains a set of "real" MCS-scheduled activities for the
      '' validation period; the validation results are written to the
1885
      " file pointed to by VAL.OUT.
1886
1887
1888
1889 define
1890
      .i,
1891
       .START.TIME,
1892
       .DURATION,
1893
       .BLOCK.
1894
       .INTERVAL.
189E
       .PRIORITY.
1896
       .VARIABCE,
1897
       .CRITICALITY,
1898
       .GA.FLAG,
1899
       . WUM . OF . VALACTS
1900
       as integer variables
1901
1902 define
1903
       .CAUSE,
1904
       .GA,
       .SV . NAME .
1905
       . ACT . WAME
1906
1907
       as text variables
1908
1909 open unit 2 for input,
      name is "valact.dat"
1910
1911 use unit 2 for input
1912
 1913 open unit 3 for output,
 1914
      name is VAL.OUT
 1915 use unit 3 for output
 1916
 1917
 1918 '' Here is a excerpt from a "valact.dat" file:
 1919 ''
            57
                     ".NUM.OF. VALACTS"
 1920 ''
            BI-008 1 NAV
                             144660 5 1 1
                                              1440 CAPE
 1921 ''
            BI-008 1 SOH
                             144200 10 1 1
                                                480 DIEG
```

```
1922
            BI-009 1 WAV
                              144145 5 1 1
                                              1440 CAPE
1923 ''
              .SV. WANE, .BLOCK, .ACT. WANE, .START.TIME, .DURATION,
1924 ''
              .VARIANCE, .CRITICALITY, .INTERVAL, .GA
1925 ''
1926
1927 read . MUH. OF . VALACTS
1928 for .i = 1 to .WUM.OF.VALACTS do
1929
1930 read
1931
          .SV. TAME,
1932
          . BLOCK,
1933
          . ACT. NAME.
1934
          .START.TIME,
1935
          .DURATION.
1936
          . VARIANCE,
1937
          . CRITICALITY,
1938
          . INTERVAL,
1939
          . GA
1940
1941
1942
       '' The first activity entity in the SCHEDULED queue that
       " matches the current validation activity is referenced.
1943
1944
       " the difference between the two start times is found and
       " a counter is incremented if the system scheduled the same
1945
1946
       '' GA that was used in reality. A TALLY statement is tracking
1947
       " value of the START.OFFSET.
       ,,
1948
1949
       for every ACT in SCHEDULED
1950
1951 with SV(ACT) = .SV. WAME and
1952
          MAME(ACT) = .ACT.WAME and
1953
          START.TIME(ACT) >= SIM.START.TIME
1954
       find the first case
1955
       if found
1956
1957
          let START.OFFSET = abs.f(.START.TIME - START.TIME(ACT))
1958
          if .GA = GA(ACT)
1959 add 1 to .GA.FLAG
1960
          endif
1961
       endif
1962 loop
1963
1964 print 10 lines with
1965
      START.DAY,
1966
       START.HR.
       START.MIN,
1967
1968
       END . DAY,
1969
       END. HR,
1970
       END . MIN ,
1971
       SIM. START. TIME,
1972 SIM. END. TIME,
```

```
1973 WUM.OF.SV,
  1974 WUM.OF.GA.
1975 WUN.SINUL.CONTACTS.
   1976
        REMARK1.
  1977 REMARK2,
  1978 REMARKS thus
   SCHEDULING INPUTS
   TIME: ***/**:** to ***/**:** (****** to ******)
   ** SVs
   ** GAS
   * Max contacts
   **********************************
   ********
   1989 print 4 lines thus
   VALIDATION RESULTS:
   Number Max Min
                          Mean Std Dev
   1994 print 4 lines with
   1995 MMBR.START.OFFSET,
   1996 MAX.START.OFFSET,
   1997 MIN.START.OFFSET,
   1998 MW.START.OFFSET,
   1999
        STD.START.OFFSET,
   2000 .GA.FLAG thus
   START OFFSET: ***** ****
                               **,****
                                          **.***
   GA MATCH:
               ****
   2005 print 4 lines thus
    ACTIVITY START OFFSET:
    Value Number Percent
    2010 for .i = 1 to 21 do
    2011 print 1 line with
    2012
          (.i-1)+10,
    2013
            HIST.SO(.i),
          100+HIST.SO(.i)/WMBR.START.OFFSET thus
    2014
    * ****
            ***.**
   2016 loop
    2017
    2018 close unit 2
    2019 close unit 3
    2020
    2021 end''VALIDATE
```

Appendix B. MCS Simulation Example Input Data.

B.1 Overview.

This appendix describes the input data used during a typical experimental run of the simulation program. As the total experimental suite was 108 runs, with each set of input data different for each run, it is not practical to provide the complete input data array here. However, the input data modification process required to duplicate the experimental results is described in Chapter 4. The purpose and use of the various data is described in Chapter 3 and the comments contained in the MCS.SIM program listing, Appendix A. General notes are provided with the listing of each data set.

B.2 Simulation VAX Command File.

This is not input data per se, but one of the command files used to direct the execution of the simulation runs. The MCS.SIM program was executed on the Air Force Institute of Technology Digital Equipment Corporation VAX 6420 (9:8.1). The default method for submitting SIMSCRIPT programs on this system does not easily allow batch submissions, so the following simple command file was created to speed the experiment process. Each input file was prepared in advance to specify the appropriate input and output files.

- \$ define/user sys\$input s16.dat
- \$ define/user sys\$output s16.out
- \$ run mcs
- define/user sys\$input s17.dat
- \$ define/user sys\$output s17.out
- \$ run mcs
- \$ define/user sys\$input s18.dat
- \$ define/user sys\$output s18.out
- \$ run mcs
- \$ define/user sys\$input s19.dat
- \$ define/user sys\$output s19.out
- \$ run mcs
- \$ define/user sys\$input s20.dat

```
$ define/user sys$output s20.out
$ run mcs
$ define/user sys$input s21.dat
$ define/user sys$output s21.out
$ run mcs
$ define/user sys$input s22.dat
$ define/user sys$output s22.out
$ run mcs
$ define/user sys$output s23.dat
$ define/user sys$output s23.out
$ run ucs
$ define/user sys$output s24.dat
$ define/user sys$output s24.dat
$ define/user sys$output s24.out
$ run mcs
```

B.3 Simulation Parameter Data

The following data is maintained in the SIMSCRIPT system default input datafile. For the VAX implementation of SIMSCRIPT, this is the file specified by the user in response to the "Enter name of Input File (.DAT assumed) (ie. DATA1)" request of the RUNSIM start program. When using the command file listed in Section B.2, the SIMSCRIPT executive will read the data file defined as the default system input (e.g., "s24.dat").

```
visall.dat
time.dat
outage.dat
act24.dat
r24.out
val.out
"Mo GA maint - no j100 CPU outage"
"Pike full availability"
"24 SVs"
```

B.4 Visibility Data

This data describes the visibility periods between satellite vehicle and ground autenna pairs.

The data used for the validation and testing of this MCS Simulator was created by the program

Pass Scheduler (6). A description of Pass Scheduler and the process used to create the data below is described in Appendix D.

5 90					
ASCN 1					
BII-26	9	18	39	1	11
BII-09	9	19	8	1	9
BII-27	9	20	32	3	11
BII-01	9	20	43	3	16
BII-13	9	21	3	2	32
BII-04	9	22	14	7	2.
BI-00B	9	22	38	J	54
BII-06	9	23	24	6	3
BII-02	10	0	39	9	52
BII-29	10	0	58	7	30
BII-09	10	1	9	5	50
BI-009	10	3	26	12	31
BII-25	10	5	6	11	52
EII-14	10	5	6	9	7
BII-11	10	5	31	13	14
BII-03	10	6	47	13	21
BI-010	10	7	31	14	19
BII-14	10	9	7	15	46
BII-02	10	9	16	10	10
BII-07	10	10	4	16	49
BII-05	10	10	16	19	33
BII-30 BII-10	10	10	34	17	13
BI-011	10	11	19	20	34
BI-011	10 10	11 11	38 57	17	58
BI-009 BI-010	10	13		12 18	31 23
BII-08	10	13	6 14	22	23 28
BII-13	10	15 15	41	20	49
BII-12	10	16	3	22	40
BII-07	10	16	49	20	40
BI-008	10	16	52	22	34
BII-28	10	16	52	23	21
BII-26	10	18	35	1	6
BII-09	10	19	1	1	5
BII-10	10	19	58	20	43
BII-27	10	20	28	3	7
BII-01	10	20	38	3	12
BII-13	10	20	58	2	27
BII-04	10	22	10	7	21
BI-008	10	22	34	3	50
BII-06	10	23	20	5	59
BII-02	11	0	34	9	12
BII-29	11	0	δ1	7	26

BII-09	11	1	5	5	46
BI-009	11	3	22	11	53
BII-25	11	5	2	11	48
BII-14	11	8	2	9	2
BII-11	11	8	27	13	9
BII-03	11	6	43	13	17
BI-010	11	7	27	14	15
BII-14	11	9	2	15	42
BII-02	11	9	12	10	6
BII-07	11	10	0	16	45
BII-05	11	10	12	19	29
BII-30	11	10	30	17	9
BII-10	11	11	15	19	54
BI-011	11	11	34	17	54
BI-009	11	11	53	12	27
BI-010	11	13	2	18	19
BII-08	11	13	10	22	24
BII-13	11	15	36	20	45
BII-12	11	15	59	22	36
BII-07	11	16	45	20	36
BI-008	11	16	48	22	30
BII-28	11	16	48	23	17
BII-26	11	18	31	1	2
BII-09	11	18	57	1	0
BII-10	11	19	54	20	39
BII-27	11	20	24	3	3
BII-O1	11	20	34	3	8
BII-13	11	20	54	2	23
BII-04	11	22	6	7	17
BI~008	11	22	30	3	46
BII-06	11	23	16	5	55
BII-O2	12	0	30	9	7
BII-29	12	0	47	7	22
BII-09	12	1	0	Б	41
BI-009	12	3	18	11	49
BII-25	12	4	58	11	44
BII-14	12	4	58	8	58
BII-11	12	5	23	13	5
BII-03	12	6	40	13	13
BI-010	12	7	23	14	11
BII-14	12	8	58	15	37
BII-02	12	9	7	10	2
BII-07	12	9	56	16	41
BII-05	12	10	8	19	25
BII-30	12	10	25	17	5
BII-10	12	11	11	19	50
BI-011	12	11	30	17	50
BI-009	12	11	49	12	23
BI-010	12	12	58	18	15
BII-08	12	13	6	22	20
BII-13	12	15	32	20	40

BIT-12	12	15	55	22	32
BII-07	12	16	41	20	32
BI-008	12	16	44	22	26
BII-28	12	16	44	23	13
BII-26	12	18	27	0	58
BII-09	12	18	53	0	56
BII-10	12	19	50	20	35
BII-27	12	20	20	2	59
BII-01	12	20	30	3	4
BII-13	12	20	50	2	19
BII-04	12	22	2	7	13
BI-CO8	12	22	26	3	42
BII-06	12	23	12	ь	51
BII-02 BII-29	13	0	26 43	9	3
BII-09	13 13	0	43 56	7 8	18 37
BI- 009	13	3	14	11	45
BII-25	13	4	53	11	40
BII-14	13	4	54	8	54
BII-11	13	5	19	13	1
BII-03	13	6	36	13	9
BI-010	13	7	19	14	6
BII-14	13	8	54	15	33
B11-02	13	Ð	3	9	58
BII-07	13	9	52	16	37
BII-05	13	10	4	19	21
BII-30	13	10	21	17	1
BII-10	13	11	7	19	46
BI-011	13	11	26	17	46
BI-009	13	11	45	12	19
BI-010	13	12	53	18	11
BII-08	13	13	2	22	16
BII-13	13	15	28	20	36
BII-12	13	15	51	22	28
BII-07	13	16	37	20	27
BI-008	13	16	40	22	22
B11-28	• •	16	40	23	9
BII-26	.,	18	22	0	54
PII-08	13	18	49	0	52
BII-10	13	19	46	20	31
BII-27	13	20	16	2	55
BII-01	13	20	26	3	C
BII-13	13	20	46	2	15
B11-04	13	21	58 22	7	9
BI008	13	22	22		38
BII-06	13	23	8	ن 0	47
BII-02	14	0	22	8	59
BII-29	14	0	39 50	7 5	13
BII-09	14	0 3	52 10	δ 11	33
BI-009	14		10	11	40
BII-25	14	4	49	11	36

BII-14	14	4	50	8	50
BII-11	14	Б	14	12	56
BII-03	14	6	32	13	5
BI-010	14	7	15	14	2
BII-14	14	8	50	15	29
BII-02	14	8	59	9	53
BII-07	14	9	48	16	33
BII-05	14	9	59	19	17
BII-30	14	10	17	16	57
BII-10	14	11	3	19	42
BI-011	14	11	22	17	42
BI-008	14	11	40	12	15
BI-010	14	12	49	18	7
BII-08	14	12	57	22	12
BII-13	14	15	24	2 C	32
BII-12	14	15	46	22	23
BII-07	14	16	33	20	23
BI-008	14	16	36	22	18
BII-28	14	16	36	23	8
BII-26	14	18	18	0	50
BII-09	14	18	45	0	48
BII-10	14	19	42	20	27
BII-27	14	20	11	2	50
BII-01	14	20	22	2	56
BII-13	14	20	42	2	11
BII-04	14	21	53	7	5
BI-008	14	22	18	3	34
BII-06	14	?3	4	5	42
END 0 0 0	0 0				
CAPE 2	_				
BII-04	8	18	17	1	0
BII-09	9	19	58	1	19
BII-02	9	20	30	3	50
BI-009	8	22	41	6	49
BII-06	9	22	57	5	43
BII-01	•	23	0	3	1
BII-25	9	23	3	3	42
BII-11	10	0	49	8	52
BI-010	10	1	6	4	49
BII-03	10	2	24	9	18
BII-29	10	2	56	7	10
BII-07	10	3	56	7	54
BII-14	10	5	7	11	22
BI-011	10	5 e	39 14	11	3
BII-30	10	6	14	13	8
BII-05	10	6	26 40	13	6
ມີມີ:-10 BT1-96	10 10	6 6	49 57	14	45
BII-25	10 10	8		11	24
BI-010 BII-08	10	8 8	49 59	13	38
BII-08	10	່າ	42	16	5 13
D11-12	10	Ü	42	16	13

BI-008	10	10	38	17	44
BII-28	10	10	43	15	45
BI-011	10	11	3	18	13
BII-07	10	11	19	16	18
BII-12	10	11	23	18	59
BII-09	10	12	56	16	10
BII-26	10	14	11	21	1
BII-01	10	14	35	19	33
BII-27	10	15	42	23	32
BII-13	10	16	13	21	58
BII-06	10	17	22	22	53
BI-008	10	17	44	23	14
BII-04	10	18	13	0	56
BII-29	10	18	43	23	33
BII-28	10	19	8	23	6
BII-09	10	19	53	1	15
BII-02	10	20	26	3	46
BI-009	10	22	37	6	45
BII-06	10	22	53	5	39
BII-01	10	22	56	2	57
BII-25	10	22	59	3	38
BII-11	11	0	44	8	48
BI-010	11	1	1	4	45
BII-03	11	2	20	9	14
BII-29	11	2	52	7	6
BII-07	11	3	52	7	50
BII-14	11	5	3	11	18
BI011	11	δ	35	10	59
BII-30	11	6	10	13	4
BII-05	11	6	22	13	2
BII-10	11	6	45	14	41
BII-25	11	6	53	11	20
BI-010	11	8	45	13	33
BII-08	11	8	55	16	1
BII-13	11	9	38	16	9
BI-008	11	10	34	13	4
BII-28	11	10	39	15	41
BI-011	11	10	59	18	9
BII-07	11	11	15	16	14
BII-12	11	11	19	18	58
BII09	11	12	52	16	6
BII-26	11	14	7	20	57
BII-01	11	14	31	19	28
BII-27	11	15	38	23	28
BII-13	11	16	9	21	54
BII-06	11	17	18	22	49
BI-008	11	17	40	23	10
BII-04	11	18	9	0	52
BII-29	11	18	39	23	29
BII-28	11	19	4	23	2
BII-09	11	19	49	1	11
					_

BII-02	11	20	22	3	41
BI-009	11	22	32	6	41
BII-06	11	22	49	5	35
BIT-01	11	22	52	2	53
LT 25	11	22	6 5	3	34
Ll1-11	12	0	40	8	44
BI-01C	12	0	57	4	41
BII-03	12	2	16	9	10
BII~29	12	2	48	7	2
BII-07	12	3	48	7	46
BII-14	12	4	59	11	14
BI-011	12	5	31	10	55
BII-30	12	6	6	13	0
BII-05	12	6	18	12	88
BII-10	12	6	40	14	37
BII-25	12	6	49	11	15
BI-010	12	8	41	13	29
BII-08	12	8	50	15	57
BII-13	12	9	34	16	5
BI-008	12	10	30	13	0
BII-28	12	10	38	15	37
BI-011	12	10	δŝ	18	5
BII-07	12	11	11	16	9
BII-12	12	11	15	18	51
BII~09	1.2	12	48	16	2
BII-26	12	14	3	20	53
BII-01	12	14	27	19	24
BII-27	12	15	33	23	24
BII-13	12	16	5	21	49
BII-06	12	17	14	22	45
BI-008	12	17	36	23	6
BII-04	12	18	5	0	48
BII-29	12	18	35	23	25
BII-28	12	19	0	22	58
BII-09	12	19	45	1	7
BII-02	12	20	17	3	37
BI-009	12	22	28	6	37
BII-06	12	22	45	Б	31
BII-01	12	22	48	2	49
BII-25	12	22	51	3	30
BII-11	13	0	36	8	40
BI-010	13	0	53	4	37
BII-03	13	2	12	9	6
BII-29	13	2	44	6	58
BII-07	13	3	44	7	42
BII-14	13	4	55	11	10
BI-011	13	5	27	10	51
BII-30	13	6	2	12	55
BII-05	13	6	14	12	54
BII-10	13	6	36	14	33
BII-25	13	6	45	11	11

BI-010	13	8	37	13	25
BII-08	13	8	46	15	δ3
BII-13	13	9	30	16	1
BI-008	13	10	26	12	56
BII-28	13	10	31	15	32
BI-011	13	10	51	18	1
BII-07	13	11	7	16	5
BII-12	13	11	10	18	47
BII-09	13	12	44	15	58
BIT-26	13	13	58	20	49
BII-01	13	14	23	19	20
BII-27	13	15	29	23	20
BII-13	13	16	1	21	45
BII-06	13	17	10	22	41
BI-008	13	17	31	23	2
BII-04	13	18	1	0	43
BII-29	13	18	30	23	21
BII-28	13	18	88	22	54
BII09	13	19	41	1	3
BII-02	13	20	13	3	33
BI-009	13	22	24	6	32
BII-06	13	22	41	5	27
BII-01	13	22	44	2	45
BII-25	13	22	46	3	26
BII-11	14	0	32	8	36
BI-010	14	0	49	4	33
BII-03	14	2	8	9	3
BII-29	14	2	40	6	54
BII-07	14	3	40	7	38
BII-14	14	4	Б0	11	5
BI-011	14	5	23	10	47
BII-30	14	Б	58	12	51
BII-O5	14	6	10	12	50
BII-10	14	6	32	14	29
BII-25	14	6	41	11	7
BI-010	14	8	33	13	21
BII-08	14	8	42	15	49
BII-13	14	9	26	15	57
BI-008	14	10	22	12	52
BII-28	14	10	27	15	28
BI-011	14	10	47	17	57
BII-07	14	11	3	16	1
BII-12	14	11	6	18	42
BII-08	14	12	39	15	53
BII-26	14	13	54	20	44
BII-01	14	14	19	19	16
BII-27	14	15	25	23	15
BII-13	14	15	57	21	41
BII-06	14	17	6	22	37
BI-008	14	17	27	22	58
BII-04	14	17	56	0	39

BII-29	14	18	26	23	17
BII-28	14	18	52	22	50
BII-09	14	19	37	0	59
BII-02	14	20	9	3	29
BI-009	14	22	20	6	28
BII-06	14	22	37	δ	23
BII-01	14	22	40	2	41
BII-25	14	22	42	3	21
BMD 0 0 0	0 0				
DIEG 3					
BII-07	9	16	38	1	50
BI-011	9	18	54	0	33
BII-05	9	19	36	2	7
BII-12	9	20	58	3	49
BII-26	9	21	58	6	26
BII-13	9	22	17	7	23
BII-08	9	23	17	4	65
BII-10	9	23	18	3	14
BII-28	9	23	29	6	32
BI-008	9	23	34	5	0
BII-07	10	1	22	1	50
BII-27	10	1	29	8	13
BII-09	10	1	34	10	47
BII-01	10	3	21	10	23
BII-04	10	3	22	7	16
BI-008	10	4	24	9	16
BII-02	10	5	48	6	34
BII-06	10	6	7	12	50
BII-13	10	6	48	7	54
BII-04	10	7	14	13	56 48
BII-29	10	7	28 44	14 14	29
BI-009	10	8	11	11	7
BII-09	10	10 10	13	18	37
BII-03 BII-11	10 10	10	43	17	16
BII-1)	10	11	4	16	32
BII-02 BII-25	10	11	42	19	37
BII-25 BII-14	10	11	47	21	4
BI-010	10	13	55	23	1
BII-30	10	14	4	22	26
BI-009	10	14	29	19	11
BII-05	10	15	26	19	32
BII-07	10	16	34	1	18
BII-10	10	16	34	23	14
BII-08	10	18	9	23	12
BI-011	10	18	50	0	29
BII-05	10	19	32	2	2
BII-12	10	20	64	3	45
BII-26	10	21	54	6	22
BII-13	10	22	12	6	43
BI-010	10	22	25	23	23

BII-08	10	23	12	4	51
BII-10	10	23	14	3	10
BII-28	10	23	25	6	28
BI-008	10	23	30	4	38
BII-07	11	1	18	1	46
BII-27	11	1	25	8	9
BII-09	11	1	30	10	7
BII-01	11	3	17	10	19
BII-04	11	3	18	7	12
BI-008	11	4	20	8	12
BII-02	11	5	44	10	54
BII-06	11	6	3	12	46
BII-13	11	6	43	7	50
BII-04	11	7	10	13	52
BII-29	11	7	24	14	44
BI-009	11	8	40	14	25
BII-09	11	10	7	11	3
BII-03	11	10	9	18	33
BII-11	11	10	39	17	12
BII-02	11	11	0	16	28
BII-25	11	11	37	19	32
BII-14	11	11	43	21	0
BI-010	11	13	51	22	21
B11-30	11	14	0	22	22
BI-009	11	14	25	19	6
BII-05	11	15	22	19	28
BII-07	11	16	30	1	14
BII-10	11	16	30	23	10
BII-08	11	18	δ	23	8
BI-011	11	18	46	0	24
BII-05	11	19	28	1	58
BII-12	11	20	50	3	41
BII-26	11	21	49	6	17
BII-13	11	22	8	6	39
BI-010	11	22	21	23	19
BII-08	11	23	8	4	47
BII-10	11	23	10	3	6
BII-28	11	23	21	6	24
BI-008	11	23	26	4	34
BII-07	12	1	14	1	42
BII-27	12	1	21	8	5
BII-09	12	1	26	10	3
BII-01	12	3	12	10	15
BII-04	12	3	13	7	8
BI-008	12	4	16	9	8
BII-02	12	δ	40	10	50
BII-06	12	Б	59	12	42
BII-13	12	6	39	7	46
BII-04	12	7	6	13	47
BII-29	12	7	20	14	40
BI-009	12	8	35	14	21
	- -				

			_		
BII-09	12	10	3	10	58
BII-03	12	10	6	18	29
BII-11	12	10	35	17	7
BII-02	12	10	56	16	24
BII-25	12	11	33	19	28
BII-14	12	11	39	20	55
BI-010	12	13	47	22	17
BII-30	12	13	56	22	18
BI-009	12	14	21	19	2
BII-OS	12	15	18	19	24
BII-07	12	16	25	1	9
BII-10	12	16	26	23	6
BII-08	12	18	1	23	4
BI-011	12	18	42	0	20
BII-05	12	19	24	1	54
BII-12	12	20	45	3	37
BII-26	12	21	45	6	13
BII-13	12	22	4	6	35
BI-010	12	22	17	23	15
BII-08	12	23	4	4	43
BII-10	12	23	6	,	2
BII-28	12	23	17	6	20
BI-008	12	23	22	4	30
BII-07	13	1	9	1	38
BII-27	13	1	17	8	1
BII-09	13	1	22	9	59
BII-01	13	3	8	10	11
BII-04	13	3	9	7	4
BI-008	13	4	12	9	4
BII-02	13	5	35	10	46
BII-06	13	5	55	12	38
BII-13	13	6	35	7	41
BII-04	13	7	2	13	43
BII-29	13	7	16	14	35
BI-009	13	8	31	14	17
BII-09	13	9	59	10	54
BII-03	13	10	2	18	25
BII-11	13	10	31	17	3
BII-02	13	10	52	16	19
B11-25	13	11	29	19	24
BII-14	13	11	35	20	51
BI-010	13	13	43	22	13
BII-30	13	13	٤2	22	14
BI-009	13	14	17	18	58
BII-05	13	15	14	19	20
BII-07	13	16	21	1	Б
BII-10	13	16	22	23	2
BII-08	13	17	57	23	0
BI-011	13	18	38	0	16
BII-06	13	19	20	1	50
BII-12	13	20	41	3	32

			_	_	_
BII-26	13	21	41	6	9
BII-13	13	22	0	6	31
BI-010	13	22	13	23	11
BII-08	13	23	0	4	39
BII-10	13	23	2	2	58
BII-28	13	23	13	6	16
BI-008	13	23	18	4	25
BII-07	14	1	δ	1	33
BII-27	14	1	13	7	57
BII-09	14	1	18	9	55
BII-01	14	3	4	10	7
BII-04	14	3	5	7	0
BI-008	14	4	8	9	0
BII-02	14	8	31	10	42
BII-06	14	δ	51	12	34
BII-13	14	6	31	7	37
BII-04	14	6	57	13	39
BII-29	14	7	12	14	31
BI-009	14	8	27	14	12
BII-09	14	9	55	10	50
BII-03	14	9	58	18	21
BII-11	14	10	26	16	59
BII-02	14	10	48	16	15
BII-25	14	11	25	19	20
BII-14	14	11	31	20	47
BI-010	14	13	3،	22	9
BII-30	14	13	48	22	10
BI-009	14	14	12	18	54
BII-05	14	15	10	19	16
BII-07	14	16	17	1	1
BII-10	14	16	18	22	58
BII-08	14	17	53	22	56
BI-011	14	18	34	0	12
BII-O5	14	19	16	1	46
BII-12	14	20	37	3	28
B11-26	14	21	36	6	5
BII-13	14	21	56	6	27
BI-010	14	22	9	23	7
BII-08	14	22	56	4	35
BII-10	14	22	58	2	54
BII-28	14	23	9	6	12
BI-008	14	23	14	4	21
END 0 0 0		•		•	
KWAJ 4					
BII-03	9	17	13	0	42
BII-25	9	17	53	2	29
BIT-14	9	17	59	3	13
BI-010	9	20	13	5	8
BI-010	9	20	25	1	40
BII-30	9	21	7	4	34
BII-30	9	22	46	4	55
PII-IO	5	22	70	7	55

UII-07	9	22	49	8	5
1.II-08	10	0	23	4	57
BI-011	10	0	56	6	47
BII-05	10	1	21	8	16
BII-12	10	3	14	9	48
BII-13	10	4	27	13	44
BI-010	10	4	32	8	46
BII-10	10	4	55	9	33
BII-08	10	4	67	11	11
BII-26	10	Б	2	12	27
BI-008	10	5	37	10	59
BII-28	10	5	41	13	30
BII-07	10	7	31	8	5
BII-27	10	7	43	14	8
BII-09	10	7	46	17	1
BII-01	10	9	28	17	16
BII-04	10	9	36	13	0
BI-008	10	9	56	15	17
BII-02	10	12	2	16	49
BII-06	10	12	20	19	8
BII-04	10	13	0	20	6
BII-29	10	13	38	22	3
BI-009	10	14	47	22	45
BII-09	10	16	25	17	12
BII-02	10	16	49	22	50
BII-11	10	16	49	23	17
BII-03	10	17	10	0	38
BII-25	10	17	49	2	25
BII-14	10	17	55	3	9
BI-010	10	20	9	4	28
BI009	10	20	20	1	36
BII-30	10	21	3	4	30
BII-05	10	21	35	1	17
BII-10	10	22	42	4	51
BTI-07	10	22	44	7	27
BII-08	11	0	19	4	53
BI-011	11	0	52	8	43
BII-05	11	1	17	8	12
BII-12	11	3	10	8	44
BII-13	11	4	23	13	4
BI-010	11	4	28	5	42
BII-10	11	4	51	9	29
BII-08	11	4	53	11	7
BII-26	11	4	58	12	23
BI-008	11	5	33	9	52
BII-28	11	5 •	37	13	26
BII-07	11	7	27	8	1
BII-27	11	7	39	14	4
BII-09	11	7	42	16	21
BII-01	11	9	24	17	12
BII-04	11	9	31	12	56

BI-008	11	9	52	15	13
BII-02	11	11	57	16	45
BII-06	11	12	16	19	4
BII-04	11	12	56	20	2
BII-29	11	13	33	21	59
BI-009	11	14	43	22	41
BII-09	11	16	21	17	8
BII-02	11	16	45	22	46
BII-11	11	16	45	23	13
BII-03	11	17	6	0	34
BII-25	11	17	45	2	21
BII-14	11	17	81	3	5
BI-010	11	20	5	4	24
BI-009	11	20	16	1	31
BII-30	11	20	69	4	26
BII-05	11	21	31	1	13
BII-10	11	22	38	4	47
BII-07	11	22	40	7	23
BII-08	12	0	14	4	49
BI-011	12	0	48	6	39
BII-05 BII-12	12	1 3	13 6	8	8
BII-12 BII-13	12	3 4		9	40
BI-13 BI-010	12 12	4	18 24	13 5	0 38
BII-10	12	4	24 47	9	36 25
BII-08	12	4	49	11	3
BII-26	12	4	54	12	19
BI-008	12	5	29	9	48
BII-28	12	5	33	13	22
BII-07	12	7	23	7	56
BII-27	12	7	35	14	0
BII-09	12	7	38	16	17
BII-01	12	9	20	17	9
BII-04	12	9	27	12	51
BI-008	12	9	48	15	9
BII-02	12	11	53	16	40
BII-06	12	12	12	19	0
BII-04	12	12	Б1	19	58
BI1-29	12	13	29	21	55
BI-009	12	14	39	22	37
BII-09	12	16	17	17	4
BII-02	12	16	40	22	42
BII-11	12	16	41	23	8
BII-03	12	17	2	0	30
BII-25	12	17	41	2	17
BII-14	12	17	47	3	1
BI-010	12	20	1	4	20
BI-009	12	20	12	1	27
BII-30	12	20	55	4	21
BII-05	12	21	27	1	9
BII-10	12	22	34	4	43

BII-07	12	22	36	7	19
BII-08	13	0	10	4	45
BI-011	13	0	44	6	35
BII-05	13	1	9	8	4
BII-12	13	3	1	9	36
BII-13	13	4	14	12	56
BI-010	13	4	20	5	34
BII-10	13	4	43	9	21
BII-08	13	4	45	10	59
BI1-26	13	4	49	12	15
BI-008	13	Б	25	9	44
BII-28	13	5	28	13	18
BII-07	13	7	19	7	52
BII-27	13	7	31	13	56
BII-09	13	7	34	16	13
BII-01	13	9	16	17	5
BII-04	13	9	23	12	47
BI-008	13	9	44	15	5
BII-02	13	11	49	16	36
BII-06	13	12	8	18	56
BII-04	13	12	47	19	53
BII-29	13	13	25	21	51
BI-009	13	14	35	22	33
BII-09	13	16	13	17	0
BII-02	13	16	36	22	38
BII-11	13	16	37	23	4
BII-03	13	16	58	0	26
BII-25	13	17	36	2	13
BII-14	13	17	43	2	56
BI-010	13	19	57	4	16
BI-009	13	20	8	1	23
BII-30	13	20	50	4	17
BII-05	13	21	23	1	Б
BII-10	13	22	30	4	39
BII-07	13	22	32	7	15
BII-08	14	0	6	4	41
BI-011	14	0	40	6	31
BII-05	14	1	Б	8	0
BII-12	14	2	57	9	32
BII-13	14	4	10	12	52
BI-010	14	4	16	5	30
BII-10	14	4	39	9	17
BII-08	14	4	41	10	55
BII-26	14	4	45	12	11
BI-008	14	5	21	9	40
BII-28	14	5	24	13	14
BII-07	14	7	15	7	48
BII-27	14	7	27	13	52
BII-09	14	7	30	16	9
BII-01	14	9	12	17	1
BII-04	14	9	19	12	43
-		-			

BI-008	14	9	40	15	1
BII-02	14	11	45	16	32
BII-06	14	12	4	18	52 52
BII-04	14	12	43	19	49
BII-29	14	13	21	21	46
BI-009	14	14	30	22	29
BII-09	14	16	9	16	55
BII-02	14	16	32	22	34
BII-11	14	16	33	23	0
BII-03	14	16	54	0	22
BII-26	14	17	32	2	9
BII-14	14	17	38	2	52
BI-010	14	19	53	4	12
BI-009	14	20	4	1	19
BII-30	14	20	46	4	13
BII-O5	14	21	19	1	1
BII-10	14	22	26	4	35
BII-07	14	22	28	7	10
	0 0				
PIKE 5					
BII-04	9	17	4	0	16
BII-02	9	19	37	3	1
BII-09	9	21	8	1	7
BII-14	9	21	43	1	17
BII-25	9	22	4	3	34
BI-009	9	22	35	5	39
BII-30	9	23	10	1	25
BI-010	10	0	16	5	11
BII-01	10	0	41	2	50
BII-11	10	0	47	7	38
BII-03	10	2	49	8	30
BII-07	10	2	59	8	6
BI-011	10	4	18	10	39
BII-29	10	4	33	В	59
BII-05	10	5	12	12	22
BII-14	10	5	48	10	55
BII-10	10	6	25	13	46
BII-30	10	6	38	12	18
BII-26	10	7	3	9	25
BII-08	10	7	59	15	16
BII-25	10	8	33	11	13
BII-13	10	8	35	12	51
BII-28	10	8	41	15	30
BI-008	10	9	46	13	57
BI-010	10	9	52	13	35
BII-12	10	11	34	17	48
BII-09	10	11	56	16	40
BII-07	10	12	41	16	8
BII-01	10	13	34	19	17
BII-26	10	14	36	20	14
BII-27	10	15	50	22	19

To the second

BII-06	10	16	15	22	25
BII-04	10	17	0	0	12
BII-13	10	17	12	21	42
BI-011	10	17	15	18	10
BII-29	10	17	42	23	23
BI-008	10	18	33	23	6
BII-03	10	19	15	21	35
BII-02	10	19	33	2	56
BII-28	10	20	48	22	55
BII-09	10	21	4	1	3
BII-14	10	21	39	1	13
BII-25	10	22	0	3	29
BI-009	10	22	31	Б	35
BII-30	10	23	6	1	21
BI-010	11	0	12	Б	7
BII-01	11	0	36	2	46
BII-11	11	0	43	7	34
BII-03	11	2	45	8	26
BII-07	11	2	55	8	2
BI-011	11	4	14	10	35
BII-29	11	4	29	6	55
BII-05	11	5	8	12	18
BII-14	11	5	43	10	51
BII-10	11	6	21	13	41
BII-30	11	હ	34	12	14
BII-26	11	6	59	9	21
BII-08	11	7	55	15	12
BII-25	11	8	29	11	9
BII-13	11	8	31	12	47
BII-28	11	9	37	15	26
BI-008	11	9	42	13	53
BI-010	11	9	48	13	31
BII-12	11	11	30	17	44
BII-09	11	11	52	16	36
BII-07	11	12	37	16	3
BII-01	11	13	30	19	13
BII-26	11	14	32	20	9
BII-27	11	15	46	22	14
BII-06	11	16	11	22	21
BII-04	11	16	53	0	8
BII-13	11	17	8	21	38
BI-011	11	17	11	18	15
BII-38	11	17	37	23	19
BI-008	11	18	29	23	2
BII-03	11	19	11	21	31
BII-02	11	19	28	2	52
BII-28	11	20	44	22	51
BII-09	11	21	0	0	59
BII-14	11	21	35	1	9
BII-25	11	21	55	3	25
BI-009	11	22	27	5	31

I-30	11	23	2	1	17
BI-010	12	0	8	Б	2
BII-01	12	0	32	2	42
BII-11	12	0	38	7	30
BII-03	12	2	41	8	22
BII-07	12	2	51	7	57
BI-011	12	4	10	10	31
BII-29	12	4	25	6	51
BII-05	12	5	4	12	14
BII-14	12	5	39	10	46
BII-10	12	6	17	13	37
BII-30	12	6	30	12	10
BII-26	12	6	55	9	17
BII-08	12	7	51	15	8
BII-25	12	8	25	11	5
BII-13	12	8	26	12	43
BII-28	12	9	33	15	22
BI-008	12	9	37	13	49
BI-010	12	9	44	13	27
BII-12	12	11	26	17	40
BII-09	12	11	48	16	32
BII-07	12	12	33	15	59
BII-01	12	13	26	19	9
BII-26	12	14	27	20	6
BII-27	12	15	41	22	10
BII-06	12	16	7	22	17
BII-04	12	16	51	0	4
BII-13	12	17	4	21	34
BI-011	12	17	7	18	11
BII-29	12	17	33	23	15
BI-008	12	18	25	22	58
BI1-03	12	19	7	21	27
BII-02	12	19	24	2	48
BII-28	12	20	40	22	46
BII-09	12	20	56	0	55
BII-14	12	21	31	1	6
BII-25	12	21	51 23	3	21
BI-009	12	22 22	23	5	27
BII.30	12		58	1	13
BI-010	13	0	4	4	58 34
BII-01 BII-11	13 13	0	28 34	2 7	38
BII-11 BII-03	13	0 2	34 37	8	26
	13	2	47	7	18
BII-07 BI-011	13	4	4.7 6	10	53 27
BII-29	13	4	21	6	47
BI1-29	1.5	5	0	12	10
BII-05 BII-14	13	5 5	35	10	42
BII-14 BII-10	13	6	36 12	13	33
BII-30	13	8	12 26	13	33 6
BII-30 BII-26	13	6	26 50		13
D11-50	13	O	٥٠	9	13

BII-08	13	7	47	16	4
BII-25	13	8	21	11	1
BII-13	13	8	22	12	39
BII-28	13	9	29	15	18
BI-008	13	9	33	13	45
BI-010	13	9	40	13	23
BII-12	13	11	22	17	36
BII-09	13	11	44	16	28
BII-07	13	12	29	15	55
BII-O1	13	13	22	19	Б
BII-26	13	14	23	20	1
BII-27	13	15	37	22	6
BII-06	13	16	3	22	13
BII-04	13	16	47	0	0
BII-13	13	17	0	21	29
BI-011	13	17	3	18	7
BII-29	13	17	29	23	11
BI-008	13	18	21	22	54
BII-03	13	19	3	21	23
BII-02	13	19	20	2	44
BII-28	13	20	36	22	42
BII-09	13	20	52	0	51
BII-14 BII-25	13	21	27	1	1
BI-009	13	21	47	3	17
BII-30	13 13	22 22	19 54	5	23 9
BI-010	14	0	0	1 4	5 4
BII-01	14	0	24	2	34
BII-11	14	Ö	30	7	21
BII-03	14	2	33	8	14
BII-07	14	2	42	7	49
BI-011	14	4	2	10	23
BII-29	14	4	17	6	43
BII-06	14	4	56	12	6
BII-14	14	5	31	10	38
BII-10	14	6	8	13	29
BII-30	14	в	21	12	2
B11-26	14	6	46	9	9
B71-08	14	7	43	15	0
BII-25	14	8	17	10	56
BII-13	14	8	18	12	35
BII-28	14	9	25	15	14
BI008	14	9	29	13	41
BI010	14	9	36	13	19
BI -:2	14	11	18	17	31
B_ 9	14	11	40	16	24
BII-07	14	12	24	15	51
BII-01	14	13	18	19	1
BII-26	14	14	19	19	67
BII-27	14	15	33	22	2
BJI-06	14	15	59	22	9

BII-04	14	16	43	23	56
BII-13	14	16	56	21	25
BI-011	14	16	59	18	2
BII-29	14	17	25	23	7
BI-008	14	18	17	22	50
BII-03	14	18	59	21	19
BII-02	14	19	16	2	40
BII-28	14	20	32	22	38
BII-09	14	20	48	0	47
BII-14	14	21	23	0	57
BII-25	14	21	43	3	13
BI-009	14	22	15	5	19
BII-30	14	22	50	1	5
BI-010	14	23	56	4	0
END 0 0 0 0	0				

B.5 Simulation Time and Parameters.

This short data file is pointed to by the value in the TIME.DAT variable and contains the simulation start and end times and three simulation parameters.

B.6 Ground Antenna Outage Data.

This data file is pointed to by the value in the OUTAGE.DAT variable and contains textual labels, then a list of GA index numbers and outage time data. The simulation will not have the GAs listed available for use during the periods described. A text description of the reason for the outage is also provided

```
"ASCW out for entire simulation."
"DIEG out for entire simulation."
2
1 144000 2880 OUTAGE
3 144000 2880 OUTAGE
```

B.7 Preliminary Activity Data.

To schedule activities immediately at the start of the simulation, the MCS Simulation program requires information on the activities that occured just prior to the start time. The file pointed to by the variable ACT.DAT contains this data. The file below contains the activities for all 24 test SVs.

24							
93							
BI-008	1	ADDKEYS	143815	10	1	1	4320
BI-008	1	VAW	143232	5	1	1	1440
BI-008	1	SOH	143805	10	1	1	480
B1-009	1	ADDKEYS	142703	10	1	1	4320
BI-009	1	MAV	142713	5	1	1	1440
BI-009	1	SOH	143758	10	1	1	480
BI-010	1	ADDKEYS	143783	10	1	1	4320
BI-010	1	MAV	142800	5	1	1	1440
BI-010	1	SOF	143773	10	1	1	480
BI-011	1	ADDKEYS	143408	1C	1	1	4320
BI-011	1	MAV	142975	5	1	1	1440
BI-011	1	SOH	143922	10	1	1	480
BII-01	2	GBDDUMP	143508	10	1	1	720
B1I-01	2	WAV	143518	15	1	1	1440
BII-01	2	WAVBUFF	142808	5	1	1	1440
BII-01	2	SOH	143498	10	1	1	720
BII-02	2	GBDDUMP	143863	10	1	1	720
BII-02	2	MAV	143873	15	1	1	1440
BII-02	2	WAVBUFF	143302	5	1	1	1440
BII-02	2	SOI	143853	16	1	1	720
BJ.I-03	2	GBDDUMP	143841	10	1	1	720
BII- 3	2	WAV	143851	15	1	1	1440
BII-03	2	MAVBUFF	143090	5	1	1	1440
BII-03	2	SOH	143831	10	1	1	720
BII-04	2	GBDDUMP	143770	10	1	1	720
BII-04	2	MAV	143780	15	1	1	1440
BII-04	2	MAVBJFF	142925	5	1	1	1440
BII-04	2	SOH	143760	10	1	1	720
BII05	2	GBDDUMP	143900	10	1	1	720
BII-05	2	WAV	143075	15	1	1	1440
BII-05	2	NAVBUFF	143910	5	1	1	1440
BII-05	2	SOH	143890	10	1	1	720
BII-06	2	GBDDUMP	143637	10	1	1	720
BII-08	2	VAV	143647	15	1	1	1440
BII06	2	NAVBUFF	142891	5	1	1	1440
BII-06	2	SOF	143627	10	1	1	720
BII-07	2	GBDDUMP	143576	10	1	•	720
	_				_	•	

BII-07 2 MAV	142918	15 1	1	1440
BII-07 2 NAVBUFF	143586	5 1	1	1440
BII-07 2 SOH	143566	10 1	1	720
BII-08 2 GBDDUMP	143837	10 1	1	720
BII-08 2 WAV	143243	15 1	1	1440
BII-08 2 MAVBUFF	143847	5 1	1	1440
BII-08 2 SOE	143827	10 1	1	720
BII-09 2 GBDDUMP	143477	10 1	1	720
BII-09 2 WAV	143487	15 1	1	1440
BII-09 2 MAVBUFF	142835	5 1	1	1440
BII-09 2 SON	143467	10 1	1	720
BII-10 3 GBDDUMF	143898	10 1	1	720
BII-10 3 WAV	143125	20 1	1	1440
BII-10 3 WAVBUFF	143908	5 1	1	1440
BII-10 3 SOLAR	143878	15 1	1	360
BII-10 3 SOR	143888	10 1	1	720
BII-11 3 GBDDUMF	143455	10 1	1	720
BII-11 3 WAV	142831	20 1	1	1440
BII-11 3 MAVBUFF	143465	5 1	1	1440
BII-11 3 SOH	143445	10 1	1	720
BII-12 3 MDUTLM	143299	10 1	1	720
BII-12 3 MAV	143309	20 1	1	1440
BII-12 3 WAVBUFF	142742	5 1	1	1440
BII-12 3 SOH	143289	10 1	1	720
BII-13 3 GBDDUMI	143980	10 1	1	720
BII-13 3 NAV	143260	20 1	1	1440
BII-13 3 WAVBUF	F 143990	5 1	1	1440
BII-13 3 SOH	143970	10 1	1	720
BII-14 3 GBDDUM	P 143400	10 1	1	720
BII-14 3 MAV	142700	20 1	1	1440
BII-14 3 WAVBUF	F 143410	5 1	1	1440
BII-14 3 SOH	143390	10 1	1	720
BII-25 3 GBDDUM		10 1	1	720
BII-25 3 WAV	143660	20 1	1	1440
BII-25 3 WAVBUF		5 1	1	1440
BII-25 3 SOH	143640	10 1	1	720
BII-26 3 GBDDUM		10 1	1	720
BII-26 3 NAV	143150	20 1	1	1440
BII-26 3 WAVBUF		5 1	_	1440
BII-26 3 SOH	143910	10 1		720
BII-27 3 GBDDUM		10 1		720
BII-27 3 WAV	142640	20 1		1440
BII-27 3 WAVBUP		5 1	_	1440
#11-27 3 SOH	1.43340	10 1	_	720
BII-28 3 GBDDUM		10 1		720
BII-28 3 WAV	143920	20 1	_	1440
BII-28 3 WAVBUF		5 1		1440
BII-28 3 SOH	143900	10 1		720
BII-29 3 GBDDUM		10 1	_	720
BII-29 3 NAV	143825	20 1		1440
BII-29 3 MAVBUF	F 143120	5 1	1	1440

BII-29	3	SOH	143805	10	1	1	720
BII-JO	3	GBDDUMP	143440	10	1	1	720
BII-30	3	WAV	142778	20	1	1	1440
BII-30	3	MAVBUFF	143450	5	1	1	1440
RTT-30	3	SOR	143430	10	1	1	720

Appendix C. MCS Simulation Validation Data/Results

C.1 Validation Activity Data.

Following is the data used in the validation of the MCS Simulation. These data describe 57 actual activities that were scheduled at the MCS on J100 (10 Apr) 1992. The data were used to compare the scheduling performance of the simulation to the schedule created at the MCS. The format is identical to the format of the data in ACT.FILE, except that the GA at which the activity was historically scheduled is appended.

57			
BI-008 1 MAV	144660	5 1 1	1440 CAPE
BI-008 1 SOH	144200	10 1 1	480 DIEG
BI-009 1 MAV	144145	5 1 1	1440 CAPE
BI-009 1 SOH	144135	10 1 1	480 CAPE
BI-010 1 WAV	144235	5 1 1	1 44 0 KWAJ
BI-010 1 SOH	144225	10 1 1	480 KWAJ
BI-011 1 MAV	144385	5 1 1	1440 PIKE
BI-011 1 SOE	144375	10 1 1	480 PIKE
BII-01 2 GBDI	DUMP 144220	10 1 1	720 DIEG
BII-01 2 WAV	144950	15 1 1	1440 KWAJ
BII-01 2 MAV	BUFF 144230	5 1 1	1440 DIEG
BII-01 2 SOE	144210	10 1 1	720 DIEG
BII-02 2 GBD	DUMP 144730	10 1 1	720 DIEG
BII-02 2 WAV	145310	15 1 1	1440 KWAJ
BII-02 2 WAV	BUFF 144740	5 1 1	1440 DIEG
BII-02 2 SOH	144720	10 1 1	720 DIEG
BII-03 2 GBD	DUMP 144520	10 1 1	720 CAPE
BII-03 2 WAV	145280	15 1 1	1440 PIKE
BII-03 2 MAV	BUFF 144530	5 1 1	1440 CAPE
BII-03 2 SOH	144510	10 1 1	720 CAPE
BII-04 2 GBD	DUMP 144355	10 1 1	720 DIEG
BII-04 2 WAV	145115	15 1 1	1440 KWAJ
BII-04 2 WAV	BUFF 144365	5 1 1	1440 DIEG
BII-04 2 SOH	144345	10 1 1	720 DIEG
BII-05 2 GBD	DUMP 144505	10 1 1	720 PIKE
BII-05 2 MAV	144515	15 1 1	1440 PIKE
BII-05 2 MAV	BUFF 145340	5 1 1	1440 KWAJ
BI1-06 2 SOR	144495	10 1 1	720 PIKE
BII-06 2 GBD	DUMP 144310	10 1 1	720 ASCN
BII-06 2 MAV	145070	15 1 1	1440 KWAJ
BII-08 2 WAV	BUFF 144320	5 1 1	1440 ASCN
BII-06 2 SOH	1/ 1300	10 1 1	720 ASCN

2	GBDDUMP	144350	10	1	1	720 PIKE
2	VAY	144360	15	1	i	1440 PIKE
2	WAVBUFF	145025	5	1	1	1440 DIEG
2	SOH	144340	10	1	1	720 PIKE
2	GBDDUMP	144655	10	1	1	720 PIKE
2	MAV	144665	15	1	1	1440 PIKE
2	NAVBUFF	145280	5	1	1	1440 DIEG
2	SOH	144645	10	1	1	720 PIKE
2	GBDDUKP	144260	10	1	1	720 ASCN
2	WAV	144905	15	1	1	1440 KWAJ
2	WAVBUFF	144270	5	1	1	1440 ASCN
2	SOH	144250	10	1	1	720 ASCN
3	GBDDUMP	144550	10	1	1	720 PIKE
3	WAV	144560	20	1	1	1440 PIKE
3	MAVBUFF	145310	5	1	1	1440 DIEG
3	SOLAR	144240	15	1	1	360 KWAJ
3	SOH	144540	10	1	1	720 PIKE
3	GBDDUMP	144265	10	1	1	720 CAPE
3	WAV	144275	20	1	1	1440 CAPE
3	MAVBUFF	144905	5	1	1	1440 DIEG
3	SOE	144255	10	1	1	720 CAPE
3	MDUTLM	144175	10	1	1	720 DIEG
3	RAV	144740	20	1	1	1440 CAPE
3	MAVBUFF	144185	5	1	1	1440 DIEG
3	SOH	144165	10	1	1	720 DIEG
	222222222223333333333333	2 WAVBUFF 2 SOH 2 GBDDUMP 2 WAV 2 WAVBUFF 2 SOH 2 GBDDUMP 2 WAV 2 WAVBUFF 2 SOH 3 GBDDUMP 3 WAV 3 WAVBUFF 3 SOLAR 3 SOH 3 GBDDUMP 3 WAV 3 WAVBUFF 3 SOH 3 MAV 3 MAVBUFF 3 SOH 3 MAV 3 MAVBUFF 3 SOH 3 MAV	2 MAV 144360 2 MAVBUFF 145025 2 SOH 144340 2 GBDDUMP 144655 2 MAV 144665 2 MAVBUFF 145280 2 SOH 144645 2 GBDDUMP 144260 2 MAV 144905 2 MAVBUFF 144270 2 SOH 144250 3 GBDDUMP 144560 3 MAV 144560 3 MAVBUFF 145310 3 SOLAR 144240 3 SOH 144240 3 GBDDUMP 144265 3 MAV 144560 3 MAVBUFF 145310 3 SOH 144250 3 SOH 144255 3 MAV 144275 3 MAVBUFF 144905 3 SOH 144255 3 MAVBUFF 144905 3 SOH 144255 3 MDUTLM 144175 3 MAV 144740 3 MAVBUFF 144185	2 MAV 144360 15 2 MAVBUFF 145025 5 2 SOH 144340 10 2 GBDDUMP 144655 10 2 MAV 144665 15 2 MAVBUFF 145280 5 2 SOH 144645 10 2 GBDDUMP 144260 10 2 MAV 144905 15 2 MAVBUFF 144270 5 2 SOH 144250 10 3 GBDDUMP 144250 10 3 GBDDUMP 144560 10 3 MAV 144560 20 3 MAVBUFF 145310 6 3 SOLAR 144240 15 3 SOH 144240 16 3 GBDDUMP 144265 10 3 GBDDUMP 144265 10 3 GBDDUMP 144265 10 3 GBDDUMP 144265 10 3 MAV 144275 20 3 MAVBUFF 144905 5 3 SOH 144256 10 3 MAV 144275 20 3 MAVBUFF 144905 5 3 SOH 144256 10 3 MAV 144276 20 3 MAVBUFF 144185 5	2 MAV 144360 15 1 2 MAVBUFF 145025 5 1 2 SOH 144340 10 1 2 GBDDUMP 144655 10 1 2 MAV 144665 15 1 2 MAVBUFF 145280 5 1 2 SOH 144645 10 1 2 GBDDUMP 144260 10 1 2 GBDDUMP 144260 10 1 2 MAV 144905 15 1 2 MAVBUFF 144270 5 1 3 GBDDUMP 144250 10 1 3 MAV 144560 20 1 3 MAVBUFF 145310 6 1 3 SOLAR 144240 15 1 3 SOLAR 144240 15 1 3 GBDDUMP 144265 10 1 3 MAV 144275 20 1 3 MAVBUFF 144905 5 1 3 SOH 144256 10 1 3 MAVBUFF 144905 5 5 1	2 MAV 144360 15 1 1 2 MAVBUFF 145025 5 1 1 1 2 SOH 144340 10 1 1 1 2 GBDDUMP 144685 10 1 1 1 2 MAV 14665 15 1 1 1 2 SOH 144645 10 1 1 1 2 GBDDUMP 144645 10 1 1 1 2 GBDDUMP 144260 10 1 1 1 2 MAV 144905 15 1 1 1 2 MAVBUFF 144270 5 1 1 1 2 SOH 144250 10 1 1 1 3 GBDDUMP 144560 20 1 1 1 3 SOLAR 144240 15 1 1 3 SOLAR 144240 15 1 1 3 GBDDUMP 144265 10 1 1 1 3 GBDDUMP 144265 10 1 1 1 3 GBDDUMP 144265 10 1 1 1 3 MAV 144275 20 1 1 1 3 MAVBUFF 144905 5 1 1 1 3 MAV 144275 20 1 1 1 3 MAVBUFF 144905 5 1 1

C.2 Validation Results.

There were 14 validation runs made. The object was to "tune" the simulation parameters affecting scheduling to maximize the correlation between a set of MCS-scheduled activities and that same set of activities scheduled by the simulation. The raw results of these runs are below.

SCHEDULING IMPUTS

TIME: 100/ 0: 0 to 101/ 0: 0 (144000 to 145440)

16 SVs

5 GAs

3 Max contacts

Actual GA J100 maint, w/ j100 CPU outage

Pike full availability V1 - TBS rank: hP/hL/hI

9 Prescheduled outages:

Index	Start	Duration	Cause
1	144000	90	CPU_OUT
2	144000	90	CPU_OUT
3	144000	90	CPU_GUT
4	144000	90	CPU_OUT
5	1 44 000	90	CPU_OUT
1	144120	120	PKI
3	144420	120	PMI
4	144600	120	PMI
2	145080	359	LAUMCH

RESULTS:

	Mumber	Max	Min	Hean	Std Dev
OFFSET:	92	90	0	1.9565	13.1247
PRIORITY:	92	3	0	0.0652	0.4375
GA UTIL:	1440	3	0	0.7132	0.7864
GA RESV:	1440	Б	0	1.5313	1.2396

PRIORITY:

Value	Mumber	Percent
0	90	97.83
1	0	0.
2	0	0.
3	2	2.17
4	0	0.
5	0	٥.
6	0	0.

7	0	٥.
8	0	0.
9	0	٥.
10	0	0.
11	0	٥.

GA UTILIZATION/KESERVATION:

0	678/	227	47.08/	15.76
1	530/	632	36.81/	43.89
2	199/	368	13.82/	25.56
3	33/	106	2.29/	7.36

Value Number Percent

OFFSET:

Value	Number	Percent
0	90	97.83
10	0	0.
20	0	Ο.
30	0	0.
40	0	Ο.
50	Q	O .
60	0	Ο.
70	0	0.
80	0	٥.
90	2	2.17
100	0	0.
110	0	٥.
120	0	0.
130	0	Ο.
140	0	0.
150	0	Ο.
160	0	Ο.
170	0	Ο.
180	0	Ο.
190	0	Ο.
200	0	0.
210	0	0.
220	0	٥.
230	0	0.
240	0	C.

SCHEDULING IMPUTS

TIME: 100/ 0: 0 to 101/ 0: 0 (144000 to 145440)

18 SVs

5 GAs

3 Max contacts

Actual GA J100 maint, w/ j100 CPU outage

Pike full availability

V1 - TBS rank: hP/hL/hI

VALIDATION RESULTS:

Number Max Min Mean Std Dev

START OFFSET: 57 148 0 41.1404 40.9769

GA NATCH: 23

ACTIVITY START OFFSET:

TIME: 100/ 0: 0 to 101/ 0: 0 (144000 to 145440)

16 SVs

5 GAs

3 Max contacts

Actual GA J100 maint, w/ j100 CPU outage

Pike full availability
V2 - TBS rank: hP/hL/lSV

VALIDATION RESULTS:

	Number	Max	Min	Mean	Std Dev
START OFFSET: GA MATCH:	57 23	148	0	41.1404	40.9769

ACTIVITY START OFFSET:

0	36	63.16
50	15	26.32
100	6	10.53
150	0	0.
200	0	0.
250	0	0.
300	0	0.
350	0	٥.
400	0	0.
450	Ω	٥

Value Number Percent

TIME: 100/ 0: 0 to 101/ 0: 0 (144000 to 145440)

16 SVs

5 GAs

3 Max contacts

Actual GA J100 maint, w/ j100 CPU outage

Pike full availability V3 - TBS rank: hP/lL/hI

VALIDATION RESULTS:

	Number	Nax	Min	Mean	Std Dev
START OFFSET:	57	163	o	41 . 9825	43.0053
GA MATCH:	25				

Value	Number	Percent
0	37	64.91
50	13	22.81
100	6	10.53
150	1	1.75
200	0	0.
250	0	0.
300	0	٥.
350	0	0.
400	0	٥.
450	0	0.

TIME: 100/ 0: 0 to 101/ 0: 0 (144000 to 145440)

16 SVs

5 GAs

3 Max contacts

Actual GA J100 maint, w/ j100 CPU outage

Pike full availability
V4 - TBS rank: hP/lL/lSV

VALIDATION RESULTS:

	Number	Max	Min	Mean	Std Dev
START OFFSET: GA MATCH:	57 25	168	0	41.9825	43.0053

Value	Number	Percent
-------	--------	---------

0	37	64.91
50	13	22.81
100	6	10.53
150	1	1.75
200	0	٥.
250	0	٥.
300	0	0.
350	0	٥.
400	0	٥.
450	0	٥.

TIME: 100/ 0: 0 to 101/ 0: 0 (144000 to 145440)

16 SVs

5 GAS

3 Max contacts

Actual GA J100 maint, w/ j100 CPU outage

Pike full availability
V5 - TBS rank: hP/1SV/1L

VALIDATION RESULTS:

	Number	Max	Min	Mean	Std Dev
START OFFSET:	57	148	0	40.8947	41.0565
GA MATCH:	25				

ACTIVITY STATE OFFSET:

0	37	64.91
50	14	24.56
100	6	10.53
150	0	٥.
200	0	0
260	0	٥.
300	0	0.
360	0	0.
400	0	٥.
450	0	0.

Value Lumber Percent

TIME: 100/ 0: 0 to 101/ 0: 0 (144000 to 145440)

16 SVs

5 GAB

3 Max contacts

Actual GA J100 maint, w/ j100 CPU outage

Pike full availability V6 - TBS rank: hP/lSV/hL

VALIDATION RESULTS:

	Eumber	Max	Min	Mean	Std Dev
START OFFSET: GA NATCH:	57 25	148	0	40.8947	41.0565

Value	Number	Percent
0	37	64.91
50	14	24.56
100	6	10.53
150	0	٥.
200	0	٦.
250	0	0.
300	0	٥.
350	0	0.
400	0	0.
450	0	0.

TIME: 100/ 0: 0 to 101/ 0: 0 (144000 to 145440)

16 SVs

5 GAS

3 Max contacts

Actual GA J100 maint, w/ j100 CPU outage

Pike full availability V7 - TBS rank: hP/hI

VALIDATION RESULTS:

	Number	Max	Min	Mean	Std Dev
START OFFSET: GA MATCH:	57 25	148	0	40.8246	41.0648

ACTIVITY START OFFSET:

Value Number Percent

0	20	35.09
10	7	12.28
20	3	5.26
30	3	5.26
40	4	7.02
50	2	3.51
60	4	7.02
70	1	1.75
80	3	5. 26
90	4	7.02
100	2	3.51
110	2	3.51
120	0	0.
130	0	0.
140	2	3.51
150	0	0.
160	0	Ο.
170	0	Ο.
180	0	0.
190	0	0.
200	0	0.

---SCHRDULING INPUTS

TIME: 100/ 0: 0 to 101/ 0: 0 (144000 to 145440)

16 SVs

5 GAB

3 Max contacts

Actual GA J100 maint, w/ j100 CPU outage

Pike full availability

V8 - TBS rank: hP/1I

VALIDATION RESULTS:

	Mumber	Max	Min	Mean	Std Dev
START OFFSET: GA NATCH:	57 26	148	0	40.5965	40.8355

Value	Number	Percent
0	20	35.09
10	7	12.28
20	3	5.26
30	3	5.26
40	4	7.02
50	2	3.51
60	5	8.77
70	0	Ο.
80	3	5.26
90	4	7.02
100	2	3.51
110	2	3.51
120	0	٥.
130	0	٥.
140	2	3.51
150	0	Ο.
160	0	٥.
170	0	٥.
180	0	٥.
190	0	0.
200	0	0

TIME: 100/ 0: 0 to 101/ 0: 0 (144000 to 145440)

16 SVs

Tett (1 to 10 to 1

- 1

5 GAs

3 Max contacts

Actual GA J100 maint, w/ j100 CPU outage

Pike full availability V9 - TBS rank: hP/hD

VALIDATION RESULTS:

	Number	Max	Min	Mean	Std Dev
START OFFSET:	57	148	0	40.5965	40.8355
GA MATCH:	26				

ACTIVITY START OFFSET:

Value Number Percent

0	20	35.09
- 10	7	12.28
20	3	5.26
30	3	8.26
40	4	7.02
50	2	3.51
60	8	8.77
70	0	0.
80	3	5.26
90	4	7.02
100	2	3.51
110	2	3.51
120	0	٥.
130	0	Ο.
140	2	3.51
150	0	0.
160	0	0.
170	0	٥.
180	0	0.
190	0	٥.
200	0	0.

TIME: 100/ 0: 0 to 101/ 0: 0 (144000 to 145440)

16 SVs

- X

5 GAs

3 Max contacts

Actual GA J100 maint, w/ j100 CPU outage

Pike full availability V10 - TBS rauk: hP/lD

VALIDATION RESULTS:

1207041705 1001	JU215.				
	Humber	Max	Min	Mean	Std Dev
START OFFSET: GA MATCH:	57 25	148	0	40.8246	41.0648

Value	Humber	Percent
0	20	35.09
10	7	12.28
20	3	5.26
30	3	5.26
40	4	7.02
50	2	3.51
60	4	7.02
70	1	1.75
80	3	5.26
90	4	7.02
100	2	3.51
110	2	3.51
120	0	0.
130	0	0.
140	2	3.51
150	0	0.
160	0	٥.
170	0	٥.
180	0	٥.
190	0	0.
200	0	٥.

TIME: 100/ 0: 0 to 101/ 0: 0 (144000 to 145440)

16 SVs

5 GAS

3 Max contacts

Actual GA J100 maint, w/ j100 CPU outage

Pike full availability
Vi1 - TBS rank: hP/lI/hSV

VALIDATION RESULTS:

	Number	Max	Min	Hean	Std Dev
START OFFSET:	57	1 4 8	0	40.6667	40.8276
GA NATCH:	26				

ACTIVITY START OFFSET:

Value Number Percent

20	35.09
7	12.28
3	5.26
3	5.26
4	7.02
2	3.51
Б	8.77
0	0.
3	5.26
4	7.02
2	3.51
2	3.51
0	0.
0	0.
2	3.51
0	0.
0	0.
0	0.
0	٥.
0	0.
0	0.
	7 3 4 2 5 0 3 4 2 2 0 0 0 0 0 0

TIME: 100/ 0: 0 to 101/ 0: 0 (144000 to 145440)

16 SVs

5 GAs

3 Max contacts

Actual GA J100 maint, w/ j100 CPU outage

Pike full availability

V12 - TBS rank: hP/lI/lSV

VALIDATION RESULTS:

	Mumber	Xax	Min	Hean	Std Dev
START OFFSET: GA MATCH:	57 26	148	0	40.5965	40.8355

ACTIVITY START OFFSET:

Value Number Percent

0	20	35.09
10	7	12.28
20	3	5.26
30	3	5.26
40	4	7.02
50	2	3.51
60	Б	8.77
70	0	Ο.
80	3	5.26
90	4	7.02
100	2	3.51
110	2	3.51
120	0	0.
130	0	٥.
140	2	3.51
150	0	٥.
160	0	0.
170	0	٥.
180	0	٥.
190	0	٥.
200	0	0.

TIME: 100/ 0: 0 to 101/ 0: 0 (144000 to 145440)

16 SVa

5 GAs

3 Max contacts

Actual GA J100 maint, w/ j100 CPU outage

Pike full availability

V13 - TBS rank: hP/1I/hST

VALIDATION RESULTS:

AWTINATION WES		W	M.s.	Mean	Std Dev
	Number	Max	Min	uaon	300 000
START OFFSET:	57	148	0	41.1404	40.9769
GA NATCH:	25				

Value	Number	Percent				
٥	20	35.09				
10	7	12.28				
20	3	5.26				
30	2	3.51				
40	4	7.02				
50	2	3.51				
60	6	10.53				
70	0	0.				
80	3	5.26				
90	4	7.02				
100	2	3.51				
110	2	3.51				
120	0	0.				
130	0	٥.				
140	2	3.51				
150	0	٥.				
160	C	0.				
170	0	Ο.				
180	0	0.				
190	0	Ο.				
200	0	0.				

TIME: 100/ 0: 0 to 101/ 0: 0 (144000 to 145440)

16 SVs

5 GAs

3 Max contacts

Actual GA J100 maint, w/ j100 CPU outage

Pike full availability
V14 - TBS rank: hP/lI/lST

VALIDATION RESULTS:

	Number Max		Min	Mean	Std Dev
START OFFSET: GA MATCH:	57 25	168	o	11.4211	42.6972

Value	Humber	Percent
0	18	31.58
10	9	15.79
20	5	8.77
30	3	5.26
40	3	5.26
50	1	1.75
60	3	5.26
70	2	3.51
80	2	3.51
90	4	7.02
100	2	3.51
110	3	5.26
120	0	0.
130	0	Ο.
140	1	1.75
150	0	Ο.
160	1	1.75
170	0	Ο.
180	0	0.
190	0	Ο.
200	0	Ο.

Appendix D. Visibility Table Generation.

This appendix describes the process used to generate the visibility table and prepare it for use with the MCS simulation. The visibility tables used in the MCS simulation were generated using an external program - Pass Scheduler. As the program author, Lt Col T. S. Kelso, describes the program in the instruction manual:

Pass Scheduler is a program which will allow the you to automatically generate schedules of satellite passes from a set of pre-selected files of observation sites and NORAD two-line orbital element sets (6). Pass Scheduler (henceforth PS) requires three type of data to generate visibilities:

1) a description of the satellite orbit; 2) a description of the observer's position; and 3) the viewing parameters. Here is the description of these data and how they were gathered.

D.O.1 Satellite Orbital Data PS requires satellite data be in the form of a North American Air Defense Command (NORAD) two-line orbital element set (commonly abbreviated TLE). NORAD uses this format to distribute satellite orbital data on the space objects tracked by NORAD's Space Surveillance Center. Among other details, a TLE describes the satellite orbit in terms of its Keplerian elements. Below is a description of the NORAD TLE format, from documentation provided with the PS manual.

Data for each satellite consists of three lines in the following format:

- ИИИИИ И И-ИИИИИ И ПИИВЕН+ ИИВИВИИИ. + МИМИВИВЕ ВЕКТИВ АЛАЧИМИ ОТВЕТТ 1
- 2 **Tuber war bark mar marr krafian** arr bara arr neru ar arraka ar baraharakan ar

Line 9 is a eleven-character name.

Lines 1 and 2 are the standard Two-Line Orbital Element Set Format identical to that used by MORAD and WASA. The format description is:

Line 1

```
Column Description
01-01 Line Number of Element Data
03-07 Satellite Number
10-11 International Designator (Last two digits of launch year)
12-14 International Designator (Launch number of the year)
15-17 International Designator (Piece of launch)
19-20 Epoch Year (Last two digits of year)
21-32 Rpoch (Julian Day and fractional portion of the day)
34-43 First Time Derivative of the Mean Motion
      or Ballistic Coefficient (Depending on ephemeris type)
45-52 Second Time Derivative of Mean Motion (decimal point assumed;
      blank if W/A)
54-61 BSTAR drag term if GP4 general perturbation theory was used. Otherwise,
      radiation pressure coefficient. (Decimal point assumed)
63-63 Ephemeris type
65-68 Element number
69-69 Check Sum (Modulo 10)
      (Letters, blanks, periods, plus signs = 0; minus signs = 1)
Line 2
Column Description
01-01 Line Number of Element Data
03-07 Satellite Number
09-16 Inclination [Degrees]
18-25 Right Ascension of the Ascending Node [Degrees]
27-33 Eccentricity (decimal point assumed)
35-42 Argument of Perigee [Degrees]
44-51 Mean Anomaly [Degrees]
53-63 Mean Motion [Revs per day]
64-68 Revolution number at epoch [Revs]
69-69 Check Sum (Nodulo 10)
All other columns are blank or fixed.
(4)
```

The TLE describes the orbit at a single instant (the epoch of the TLE), but the position of the satellite at other times can be determined by mathematically "propagating" the epoch trajectory forward or back in time.

TLE's exist for all the GPS satellites now operational. However, the simulation required TLEs for GPS satellites not yet launched. The problem was solved using the trajectories of the existing satellites and the fact that the GPS SVs are spaced at 90 degree intervals in the six orbital planes (5:88). The trajectories for the missing satellites were estimated by propagating the trajectories of the existing satellites to a common epoch. With the satellite's positions fixed at a common

time, the missing orbital "slots" in each plane becomes apparent. Assuming the newly launched satellites will fill these holes, element sets were created using the appropriate elements from the existing satellites in the plane, with the locations of the slots filling in the unknown quantities.

Once the GPS SV TLEs have been either gathered or synthesized, they are placed in a file named GPS.TLE to be called by PS at the appropriate time. The TLE file used to generate the visibility is as follows:

GP8-0010 1 152710 84 97 A 92726.07205774 -.00000014 00000-0 99999-4 0 3198 2 15274 62.7700 300.3243 0124103 339.4139 20.1675 2.00682276 56983 GPS BII-12 1 21890U 92 9 A 92226.18328797 -.00000014 00000-0 99999-4 0 1050 2 21890 64.5839 284.3872 0060248 144.1406 216.3366 2.00664032 3484 GPS BII-04 1 203020 89 85 4 92226.85446246 -.00000015 00000-0 99999-4 0 3997 2 20302 54.0381 284.6837 0016025 327.6882 32.2974 2.00568403 20687 GPS BII-07 1 20533U 90 25 A 92226.67959571 -.00000033 00000-0 99999-4 C 4013 2 20533 55.2252 344.1496 0039438 92.2444 268.1777 2.00563763 17434 GPS BII-02 1 20061U 89 44 A 92226.48398376 -.00000033 00000-0 99999-4 0 4656 2 20061 64.8872 343.8391 0109792 194.7542 164.9389 2.00561894 23309 GP8-0009 1 15039U 84 59 A 92226.62982646 -.00000004 00000-0 99999-4 0 6315 2 15039 63.5674 62.1557 0040962 218.1410 141.6396 2.00568804 59854 GPS BIT-13 1 21930U 92 19 A 92225.89620157 -.00000019 00000-0 99999-4 0 1053 GPS-0008 1 14189U 83 72 A 92226.46325186 -.00000004 00000-0 99999-4 0 2580 2 14189 63.6084 63.1610 0136525 231.4366 127.4011 2.00563083 66575 GP8-0011 1 16129U 85 93 A 92226.79394660 -.00000003 00000-3 99999-4 0 207 2 16129 64.3249 63.7938 0128101 143.6668 217.2856 2.00565079 50179 QPS BII-11 1 21652U 91 47 A 92226.22305375 .00000006 00000-0 99999-4 0 2105 2 21552 55.4011 104.5586 0042939 225.0711 134.6604 2.00573168 8113 GPS BII-09 1 20830U 90 88 A 92226.54954593 .00000007 00000-0 99999-4 0 3525 2 20830 65.1038 107.1760 0070046 109.7801 261.0560 2.005623€7 13938 1 20361U 89 97 A 92226.36914852 .00000007 00000-0 99999-4 0 3523 2 20361 55.1884 109.0680 0066568 83.3586 277.4965 2.00563324 10165 OPS BIT-01 1 19802U 89 13 A 92226.14753199 .00000016 00000-0 99999-4 0 4509 2 19502 55.0467 166.3474 0041381 172.7861 187.2444 2.00560622 25559 1 207240 90 68 4 92225.99861477 .00000016 00000-0 99999-4 0 3045 2 20724 54.6819 164.8810 0092524 131.8562 228.6309 2.00565479 14842 **GPS BII-03** 1 2018CU 89 64 A 92223.32198841 .00000016 00000-0 99999-4 0 4014 2 20185 54.8802 167.0704 0012047 198.6931 161.2345 2.00557084 21872 GPS BII-10 1 20959U 90103 A 92225.45418467 .90000016 00000-0 99999-4 0 2536

```
2 20959 54.9117 166.6623 0062140 219.0715 140.4481 2.00566067 12504 

GP8 BII-14
1 22014U 92 39 A 92225.93117523 .00000010 00000-0 99999-4 0 231 

-2 22014 55.0433 224.5891 0083797 275.3527 83.6992 2.00570262 738 

GP8 BII-06
1 20452U 90 8 A 92226.77349685 .00000009 00000-0 99999-4 0 4005 

2 20452 54.1853 223.8693 0044556 65.8465 294.6162 2.00562850 18661
```

D.0.2 Observer Position Data. The positions of the GPS ground stations, especially the Monitor Stations, have been determined very accurately. The positions provided to PS in the file GPS.OBS were taken directly from a GPS Mission Console display. The OBS file is reproduced below.

ASCE	1	07	57	04	9	014	24	43	¥	105
CAPE	1	28	29	02	ı	080	34	22	¥	-18
DIEG	1	07	16	12	8	072	22	15	E	-63
EVAJ	1	80	4.3	23	ľ	167	43	52	E	41
PIRE	1	38	48	11	ı	104	31	30	W	1913

Each line describes the location of one ground station. In the Pass Scheduler instructions, Kelso describes the format of the file thus:

Each observer file must have an extension .OBS. A sample observer file is shown below. Each line allows 20 characters for a place name, a time zone difference (not used), and the observation site's latitude (in degrees, minutes, and seconds followed by N or S), longitude (in degrees, minutes, and seconds followed by W or E), and altitude above mean sea level (in meters). (6)

D.0.3 Pass Schedule Parameters. The remaining information required by PS is to the time difference between the program users' time and Universal Coordinated Time (commonly referred to as "Greenwich" or "ZULU" time), and a number of parameters that define the users' input and output desires. This information is stored in a file called "PASSCHED.CFG", which is shown below.

```
% Time difference from UTC
% Echo satellite data
```

Y X Echo schedule to disk file
I: X Def-ult drive for support files
X Default directory for support files
GPS.OBS X Default observer database
WEWOPS.TLE X Default satellite database

During program execution, the user is prompted to enter the desired visibility time interval, the satellite elevation (from the ground station local horizon) at which the satellite is declared risen or set, and the minimum duration visibility period the user wishes to have reported.

D.O.4 Pass Schedule Output. In the raw Pass Scheduler output, the name and location of the ground stations for which the following satellite observations apply is listed first, along with the user-defined parameters. Then each satellite is listed in the order in which they rise at that station. The times (under "Rise", "ToC", and "Set") are in hour/minute/second format. "Az" and "El" are satellite azimuth and elevation, respectively. ToC is time of culmination, the instant of maximum elevation for that satellite visibility period. The last column contains codes for classifying the visibility period.

To convert this format to the one needed for the MCS Simulation program, the raw PS output was imported to a computer spreadsheet, the data formed into columns, and the extraneous data removed. The long ground station header was replaced with just the name and index number of the GA.

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attended public school in Toledo, graduating from Jesup W. Scott High School in 1971. After a brief stay at Case Western Reverve University in Cleveland, Ohio, he enlisted in the United States Air Force on 10 November 1972. As an electronic equipment repairman, he served in Misaissippi, North Dakota, California, Germany, Mississippi again, and Washington.

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13. ABSTRACT - Maximum 200 worst)

Abstract

The United States Air Force's Navstar Global Positioning System (GPS) provides high-accuracy spacebased navigation and time distribution to suitably-equipped military and civilian users. The system consists of earth-orbiting satellites and a world-wide network of ground stations. A single operational control center, the GPS Master Control Station (MCS) monitors, maintains, and commands the GPS satellite constellation. The on-going deployment of the complete satellite constellation and recent changes in the operational crew structure may invalidate previously used planning and management paradigms. There is currently no analytical method for predicting the impact of these and other environmental changes on system parameters and performance. Extensive testing cannot be performed at the MCS itself due to the criticality of the GPS mission and lack of operational redundancy. This research provides and validates a discret event simulation model of the MCS operations center task flow, focusing on the creation and testing of a sliding-widow MCS activity scheduler. The simulation was validated using MCS historical data. Experiments were conducted by varying the number of ground stations and satellite constellation size available to the simulation. The results, while not quantitatively trustworthy, were used to draw general conclusions about the GPS operational environment.

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